



## OBJECT

Strength Calculation for 4840-TK-120  
Poboya 2000tpd Expansion Project

**Client No.: 2000-4840-MES-CAL-001**



## REFERENCES

3655.15-61.08-4000-F-BOD-001	Process Design Criteria
3655.15-61.08-4000-F-CAL-001	Mass Balance

## ASSUMPTIONS


0	Issued for Information	22-May-26	Wawan	Ardiyanto	Muraasato	
A	Issued for Review	14-Apr-26	Wawan	Ardiyanto	Muraasato	
Revision	Description	Date	Prepared By	Reviewed By	Approved	Client App
CLIENT PT Citra Palu Minerals		TITLE Strength Calculation for Tailing Thickener Overflow				REV 0
PROJECT Poboya 2000tpd Expansion Project		COMO ENGINEERS		PROJECT No. E2602		
BY Wawan	DATE 22-May-26			CHECKED Ardiyanto	DATE 22-May-26	Doc No E2602-4840-CAL-101

[illegible]

	<b>POBOYA 2000 TPD EXPANSION PROJECT</b>	
(CPMWork No.) 14898	Project Doc. No.: E2602-4820-CAL-101	Revision No. : 0
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## 1. Introduction

This document consist of additional analysis for tailing overflow tank shell thickness. The dimension is 4.5 m height and 7 m for the diameter. Additional analysis needed to calculate the strength of shell tank to resist walkway platform load that located on top of the tank.

## 2. Support Document

- 3655.15-61.08-4840-M-DWG-001 – GA – Tailing Thickener Overflow Tank

## 3. Code and Reference

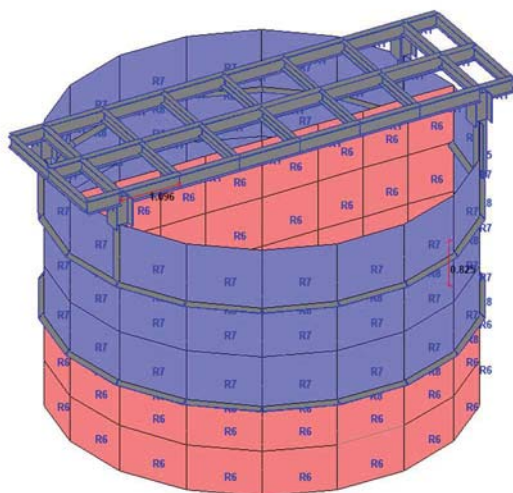
Code and reference that used for this calculation listed below :

- API STD 650 2018 – Welded Tanks for Oil Storage
- SNI 1729:2020 – Spesifikasi untuk Bangunan Gedung Baja Struktural

## 4. Design


### 4.1 Isometric Model

Analysis design for the tank use STAAD Pro software. Tank model picture showed below :



Material			
PLATE			
Shell elev. (m)	Main Thickness (mm)	Corrosion Allowance (mm)	Design Thickness (mm)
Bottom Plate	7	3	10
0 - 4.5	5	3	8
STIFFENER			
Profile elev. (m)	Type		
1.65	L.130.130.12		
3.45	L.130.130.12		
4.5	L.200.200.25		
PLATFORM SUPPORT			
Profile elev. (m)	Type		
4.5	Half Tee. 300.300.10.15		

Picture 1. Tank Model and Properties

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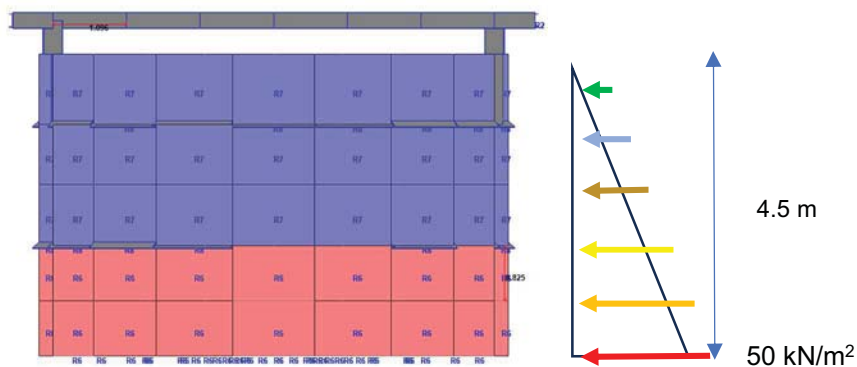
## 4.2 Load and Combination

### 4.2.1 Daad Load

Unit weight of steel used as calculation is 78.5 kN/m<sup>3</sup>.

### 4.2.2 Hidrostatic Load

Hidrostatic load is pressure load from fluida content. Specific gravity fluida is 1.11 Ton/m<sup>3</sup>



Hidrostatic load :  $1.11 \times 4.5 = 4.99 \text{ Ton/m}^2 \rightarrow 50 \text{ kN/m}^2$

### 4.2.3 Walkway Platform Load

#### GRATING

$$q = 0.45 \text{ kN/m}^2$$

$$\text{Tributary 1} = 1 \text{ m}$$

$$\text{Tributary 2} = 0.5 \text{ m}$$

$$\text{Input load 1} = 0.45 \text{ kN/m}$$

$$\text{Input load 2} = 0.225 \text{ kN/m}$$

#### LIVE LOAD


$$q = 2.5 \text{ kN/m}^2$$

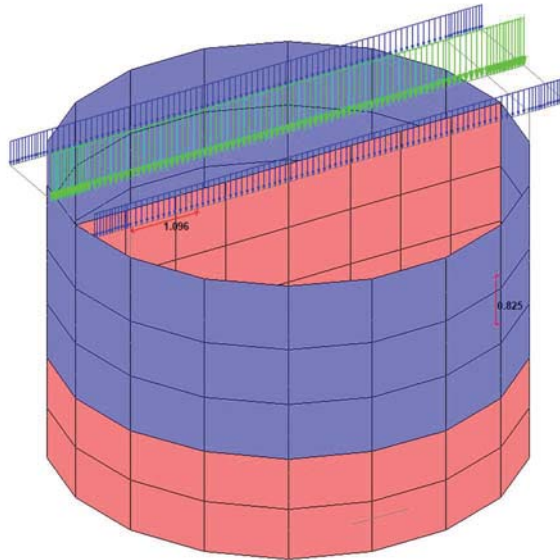
$$\text{Tributary 1} = 1 \text{ m}$$

$$\text{Tributary 2} = 0.5 \text{ m}$$

$$\text{Input load 1} = 2.5 \text{ kN/m}$$

$$\text{Input load 2} = 1.25 \text{ kN/m}$$

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Picture 2. Walkway Input Load

#### 4.2.4 Seismic Load

Seismic load calculated based on API 650.

- The seismic overturning moment at the base of the tank shell shall be the SRSS summation of the impulsive and convective components multiplied by the respective moment arms to the center of action of the forces unless otherwise specified.

Ringwall Moment,  $M_{rw}$ :

$$M_{rw} = \sqrt{[A_i(W_i X_i + W_s X_s + W_r X_r)]^2 + [A_c(W_c X_c)]^2} \quad (\text{E.6.1.5-1})$$

#### Equation Ai :



Impulsive spectral acceleration parameter,  $A_i$ :

$$A_i = S_{DS} \left( \frac{I}{R_{wi}} \right) = 2.5 Q F_a S_0 \left( \frac{I}{R_{wi}} \right) \quad (\text{E.4.6.1-1})$$

$$\text{However, } A_i \geq 0.007 \quad (\text{E.4.6.1-2})$$

and, for  $S_1 \geq 0.6$ :

$$A_i \geq 0.5 S_1 \left( \frac{I}{R_{wi}} \right) = 0.625 S_p \left( \frac{I}{R_{wi}} \right) \quad (\text{E.4.6.1-3})$$

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$$I = 1 \text{ (category II)}$$

$$S_{ds} = 0.806$$

$$R_{wi} = 3.5$$

**Table E.4—Response Modification Factors for ASD Methods**

Anchorage system	$R_{wi}$ (impulsive)	$R_{wc}$ (convective)
Self-anchored	3.5	2
Mechanically-anchored	4	2

$$A_i = 0.23$$

### **Equation Ac**

**Convective spectral acceleration parameter,  $A_c$ :**

$$\text{When, } T_C \leq T_L \quad A_c = K S_{D1} \left( \frac{1}{T_c} \right) \left( \frac{I}{R_{wc}} \right) = 2.5 K Q F_a S_0 \left( \frac{T_s}{T_c} \right) \left( \frac{I}{R_{wc}} \right) \leq A_i \quad (\text{E.4.6.1-4})$$

$$\text{When, } T_C > T_L \quad A_c = K S_{D1} \left( \frac{T_L}{T_c^2} \right) \left( \frac{I}{R_{wc}} \right) = 2.5 K Q F_a S_0 \left( \frac{T_s T_L}{T_c^2} \right) \left( \frac{I}{R_{wc}} \right) \leq A_i \quad (\text{E.4.6.1-5})$$

$$A_i = 0.23$$

$$K_s = 0.58$$



$$T_c = 2.78$$

$$R_{wc} = 2$$

$$K = 1.5$$

$$S_{D1} = 0.582$$

$$A_c = 0.16$$

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### Equation $W_i$ and $X_i$ :

#### **E.6.1.1 Effective Weight of Product**

The effective weights  $W_i$  and  $W_c$  shall be determined by multiplying the total product weight,  $W_p$ , by the ratios  $W_i/W_p$  and  $W_c/W_p$ , respectively, Equations E.6.1.1-1 through E.6.1.1-3.

When  $D/H$  is greater than or equal to 1.333, the effective impulsive weight is defined in Equation E.6.1.1-1:

$$W_i = \frac{\tanh\left(0.866\frac{D}{H}\right)}{0.866\frac{D}{H}} W_p \quad (\text{E.6.1.1-1})$$

When  $D/H$  is less than 1.333, the effective impulsive weight is defined in Equation E.6.1.1-2:

$$W_i = \left[1.0 - 0.218\frac{D}{H}\right] W_p \quad (\text{E.6.1.1-2})$$

When  $D/H$  is greater than or equal to 1.3333, the height  $X_i$  is determined by Equation E.6.1.2.1-1:



$$X_i = 0.375H \quad (\text{E.6.1.2.1-1})$$

When  $D/H$  is less than 1.3333, the height  $X_i$  is determined by Equation E.6.1.2.1-2:

$$X_i = \left[0.5 - 0.094\frac{D}{H}\right] H \quad (\text{E.6.1.2.1-2})$$

D	=	7 m
H	=	4.5 m
D/H	=	1.56
$\gamma$	=	1.11
$W_p$	=	1886.54 kN
perimeter	=	22.00 m
$3.67H/D$	=	2.36
<b><math>W_i</math></b>	=	<b>1223.09 kN</b>
<b><math>X_i</math></b>	=	<b>1.69 m</b>



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### Equation Wc and Xc :

The effective convective weight is defined in Equation E.6.1.1-3:

$$W_c = 0.230 \frac{D}{H} \tanh\left(\frac{3.67H}{D}\right) W_p \quad (\text{E.6.1.1-3})$$

The height  $X_c$  is determined by Equation E.6.1.2.1-3:

$$X_c = \left[ 1.0 - \frac{\cosh\left(\frac{3.67H}{D}\right) - 1}{\frac{3.67H}{D} \sinh\left(\frac{3.67H}{D}\right)} \right] H \quad (\text{E.6.1.2.1-3})$$

$$W_c = 663.02 \text{ kN}$$

$$X_c = 2.92 \text{ m}$$

Value of Ws and Xs :

Ws = Total weight of tank shell and appurtenances

Xs = Height from bottom of the tank shell to the shell's center of gravity

L.130.130.12

$$A = 2976 \text{ mm}^2 = 0.002976 \text{ m}^2$$

$$w = 5.14 \text{ kN}$$

$$n = 2$$

$$\text{Weight Total L.130} = 10.28 \text{ kN}$$

L.200.200.25

$$A = 9375 \text{ mm}^2 = 0.009375 \text{ m}^2$$

$$w = 16.19 \text{ kN}$$

$$n = 1$$

$$\text{Weight Total L.130} = 16.19 \text{ kN}$$



H.300.300.15.10 H.Tee

$$A = 11850 \text{ mm}^2 = 0.01185 \text{ m}^2$$

$$w = 1.53 \text{ kN}$$

$$n = 4$$

$$\text{Weight Total L.130} = 6.14 \text{ kN}$$

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Shell T 8 mm

$$A = 0.18 \text{ m}^2$$

$$w = 24.87 \text{ kN}$$

Shell T 8 mm

$$A = 0.18 \text{ m}^2$$

$$w = 37.30 \text{ kN}$$

$$\text{TOTAL WEIGHT (Ws)} = 94.78 \text{ kN}$$

$$Xs = 2.25 \text{ m}$$

From calculation above,

$$(A_i \cdot W_i) / \text{perimeter} = 12.80 \text{ kN/m, at H} = 1.69 \text{ m}$$



$$(A_i \cdot W_s) / \text{perimeter} = 0.99 \text{ kN/m, at H} = 2.25 \text{ m}$$

$$(A_c \cdot W_c) / \text{perimeter} = 4.74 \text{ kN/m, at H} = 2.92 \text{ m}$$

For simplification :

$$W = 18.53 \text{ kN/m}$$

$$H = 2.03 \text{ m}$$

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## ANCHORAGE

Bolt Pitch Circle Diameter = 7.2 m

Perimeter Bolt Pitch Circle = 22.63 m

Anchor Bolt Diameter = 27.00 mm

Root area anchor = 572.79 mm<sup>2</sup>

Bolt Material Grade 307 Class C

$f_y$  = 235 MPa

$f_u$  = 400 MPa

D = 7.00 m

Mrw = 606.41 kNm

$A_v$  (0.47 Sds) = 1.43

W2 = 71.47 kN

U = 315.82 kN

N = 12

Tb = 26.32 kN

Allowable Anchor Stress, Sd = 188 MPa (0.8 .  $F_y$ )

Tn = 80.76 kN (Sd . Ab . 0.75)

Ø Tn = 72.69 kN (0.9 Tn)

Tb < Ø Tn OK

Depth of Anchor,  $hef$  = 500 mm (embedded to concrete)



$k_c$  = 10

$\lambda$  = 1

$f_c'$  = 28.00 MPa

Concrete breakout strength, Nb = 591.61 kN ( $k_c . \lambda . \sqrt{f_c'} . hef^{1.5}$ )

Nb > Tb OK

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#### 4.2.5 Wind Load

Wind load calculated based on SNI 1727 : 2020.

#### 4.2.6 Load Combination


Load combination based on SNI 1729 : 2020

Service load combination :

1.  $D$
2.  $D + L$
3.  $D + (L_r \text{ atau } R)$
4.  $D + 0,75 L + 0,75(L_r \text{ atau } R)$
5.  $D + 0,6 W$
6.  $D + 0,75 (0,6 W) + 0,75 L + 0,75 (L_r \text{ atau } R)$
7.  $0,6D + 0,6 W$
  
8.  $1,0D + 0,7E_v + 0,7E_h$
9.  $1,0D + 0,525E_v + 0,525E_h + 0,75L$
10.  $0,6D - 0,7E_v + 0,7E_h$

Ultimate load combination :

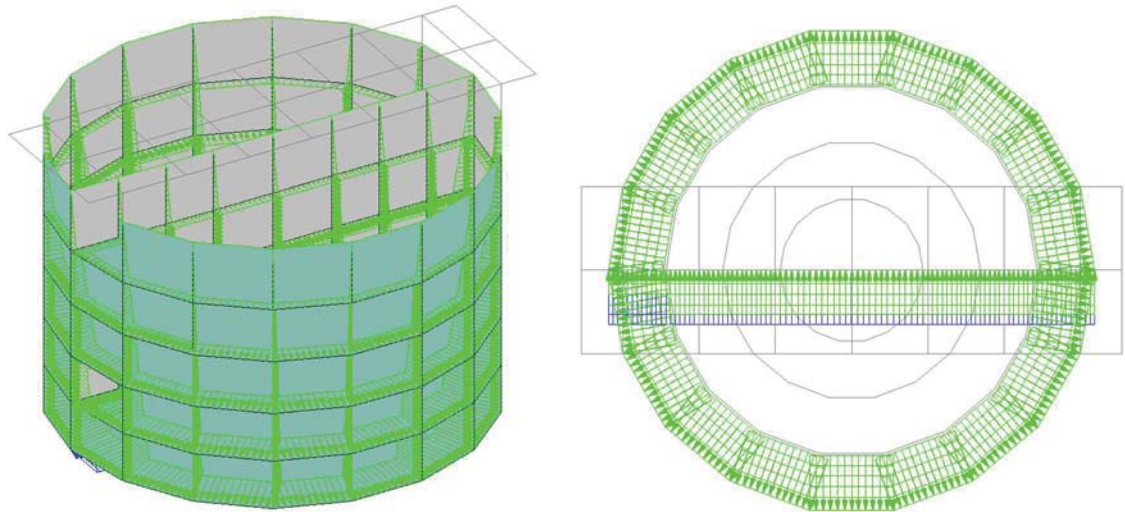
1.  $1,4D$
2.  $1,2 D + 1,6 L + 0,5 (L_r \text{ atau } R)$
3.  $1,2D + 1,6 (L_r \text{ atau } R) + (L \text{ atau } 0,5 W)$
4.  $1,2D + 1,0 W + L + 0,5(L_r \text{ atau } R)$
5.  $0,9 D + 1,0 W$
  
6.  $1,2D + E_v + E_h + L$
7.  $0,9D - E_v + E_h$

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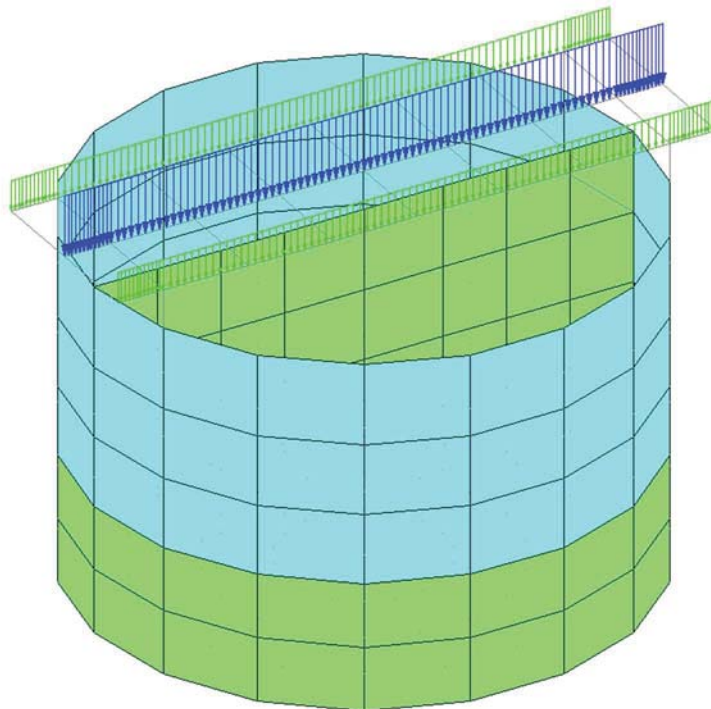
### 4.3 Analysis Design


#### 4.3.1 Input Model

##### a. Hidrostatic Load (MAX =80.34 kN/m<sup>2</sup>)

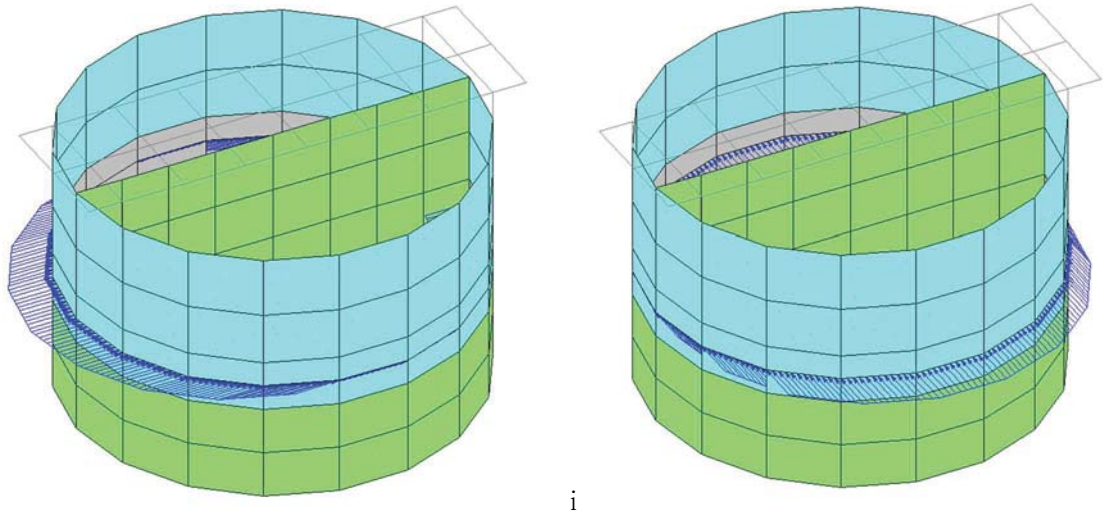


##### b. Walkway Platform Load

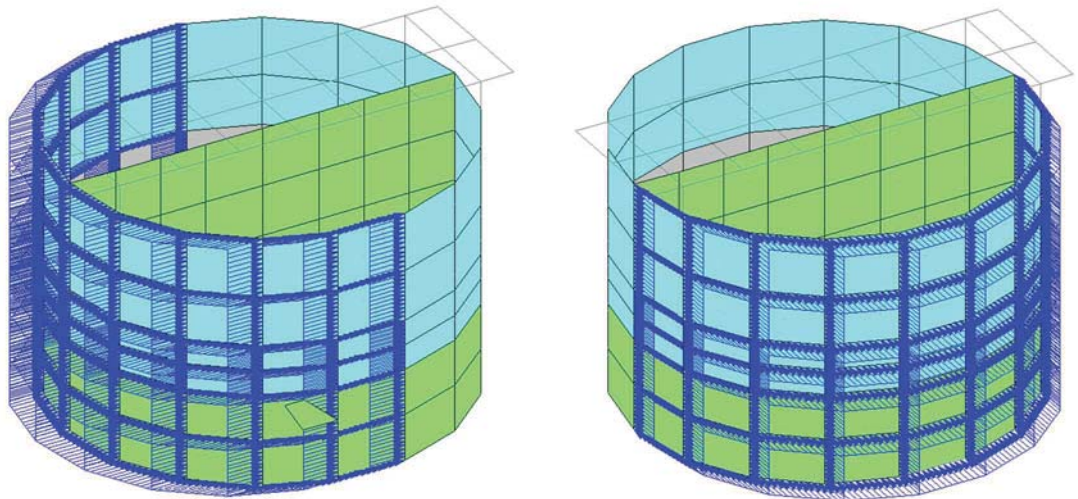


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c. Seismic Load ( $W_i = 29.91 \text{ kN/m}$ )



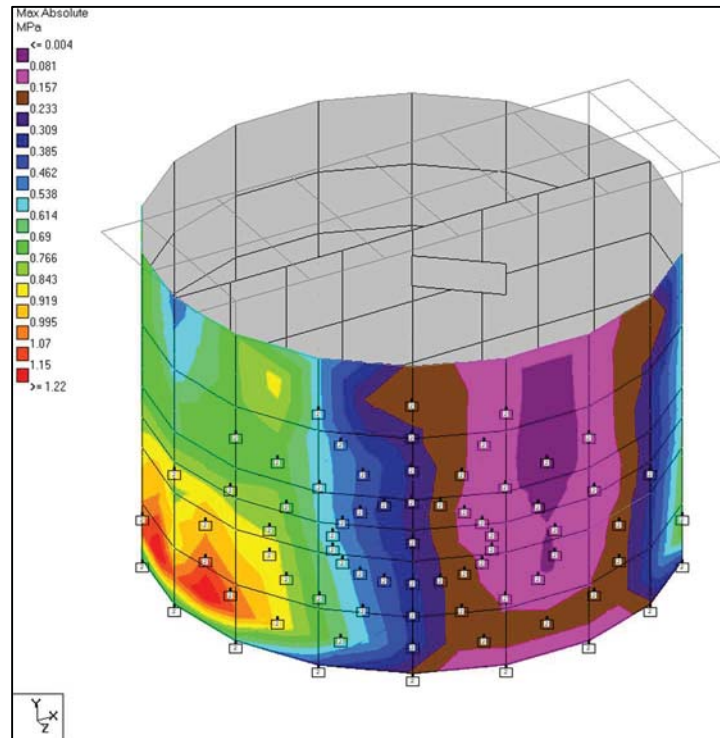
d. Wind Load (  $0.6 \text{ kN/m}^2$  )



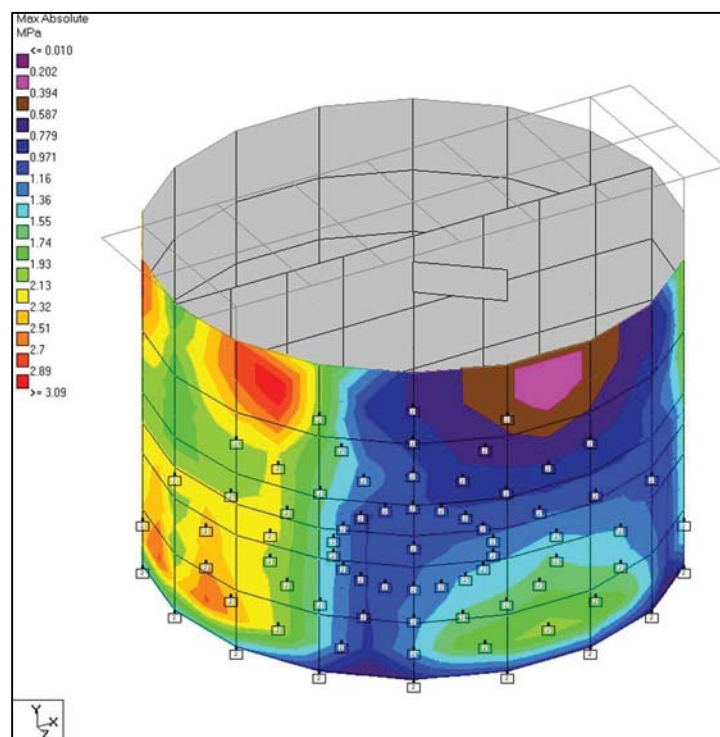


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#### 4.3.2 Output Data



**Stress Shell due Selfweight (max : 1.22 MPa)**



**Stress Shell due Platform Support (max : 3.1 MPa)**

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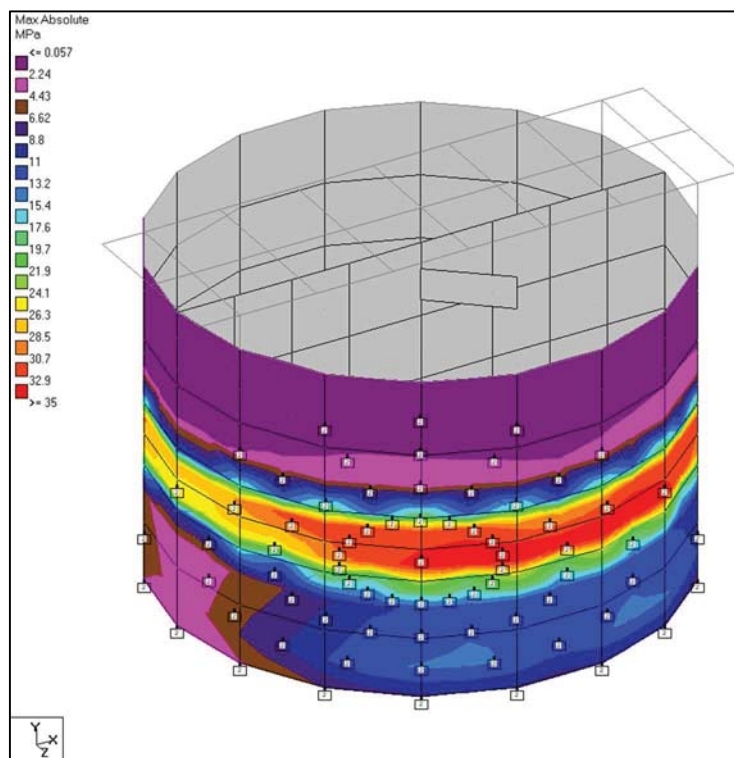
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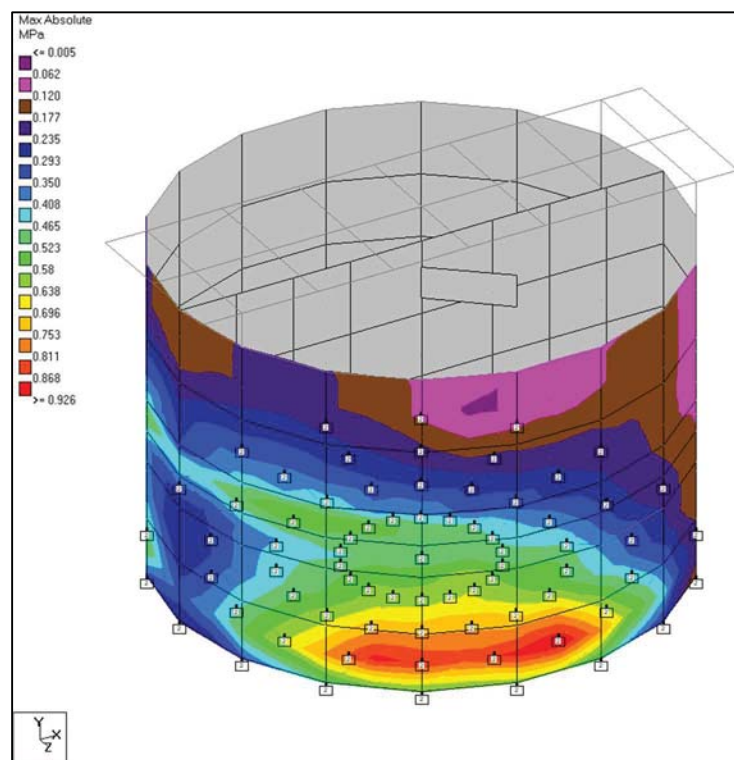
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**Stress Shell due Seismic Load EQx (max : 35 MPa)**



**Stress due Windload Wx+ (max : 1 MPa)**



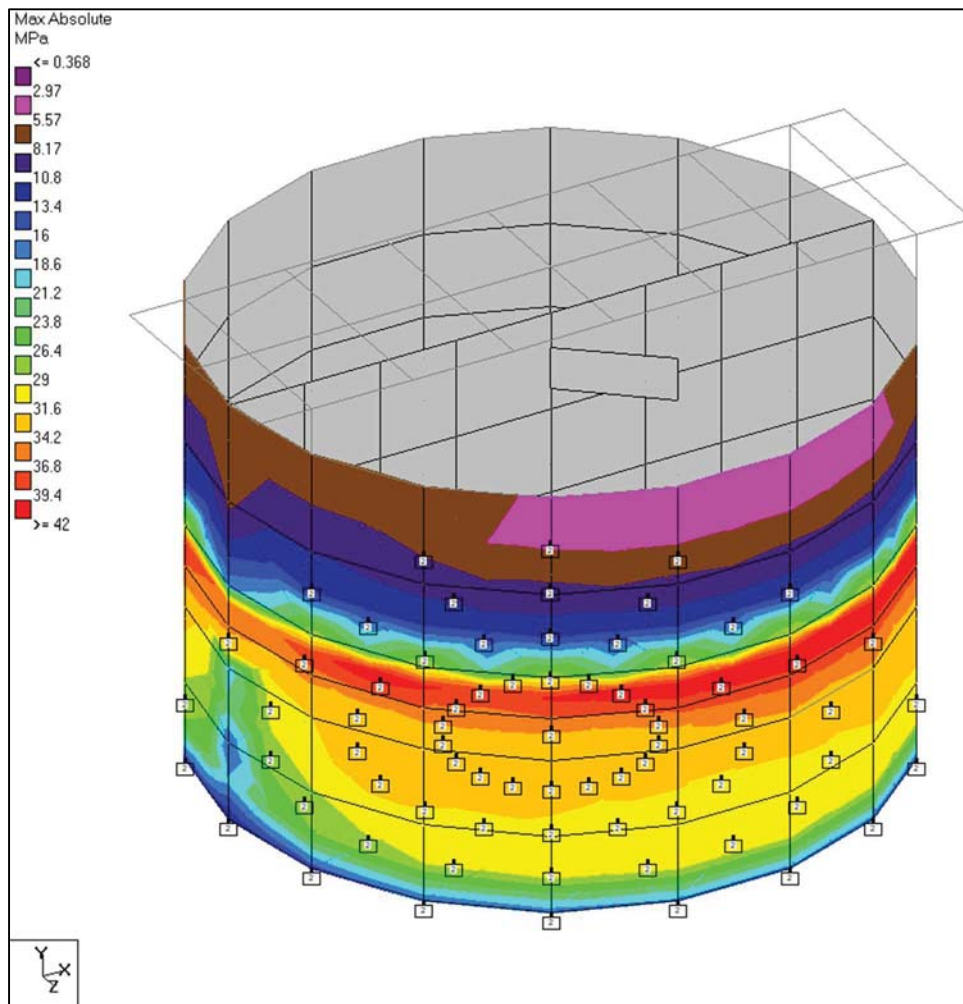

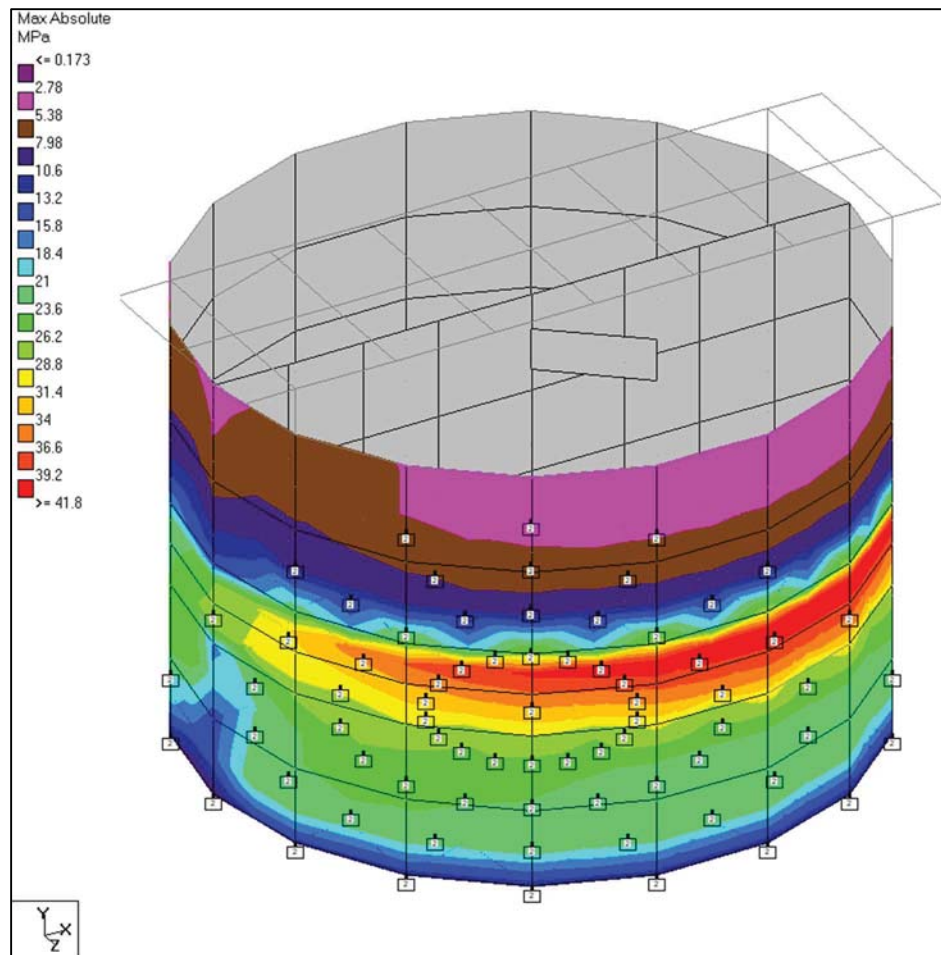

**Stress Summary Table**


	Plate	L/C	Shear (Local)		Membrane (Local)			Bending Moment (Local)		
			SQX MPa	SQY MPa	SX MPa	SY MPa	SXY MPa	MX kN-m/m	MY kN-m/m	MXY kN-m/m
Max Qx	400	19 201	0.259	0.003	-0.411	0.447	8.497	0	0	-0.07
Min Qx	382	19 201	-0.259	-0.003	0.399	-0.49	-8.771	0	0	0.07
Max Qy	403	20 202	-0.006	0.142	18.419	-0.18	0.113	0.011	0.006	-0.002
Min Qy	385	20 202	0.006	-0.142	-18.386	0.137	0.16	-0.011	-0.006	0.002
Max Sx	403	24 206	-0.004	0.101	39.427	0.454	0.538	0.01	0.01	-0.002
Min Sx	385	20 202	0.006	-0.142	-18.386	0.137	0.16	-0.011	-0.006	0.002
Max Sy	39	24 206	0	-0.008	24.936	4.583	0.846	0	0.001	0
Min Sy	268	18 103	0	0	-0.33	-7.609	3.725	0	0	0
Max Sxy	43	24 206	-0.002	-0.008	15.462	-3.049	14.214	0	0	0
Min Sxy	51	24 206	0.003	-0.009	15.006	-4.551	-14.05	0	0	0
Max Mx	276	20 202	0.004	-0.011	0.27	-0.816	0.218	0.018	0.009	-0.008
Min Mx	275	22 204	0.004	0.011	0.27	-0.816	-0.218	-0.018	-0.009	-0.008
Max My	405	23 205	0.028	0.097	38.763	-0.488	1.611	0.011	0.01	0.013
Min My	275	22 204	0.004	0.011	0.27	-0.816	-0.218	-0.018	-0.009	-0.008
Max Mxy	399	19 201	0.202	-0.004	-0.502	0.37	-0.091	0	0	0.086
Min Mxy	381	19 201	-0.202	0.004	0.535	-0.413	-0.182	0	0	-0.086

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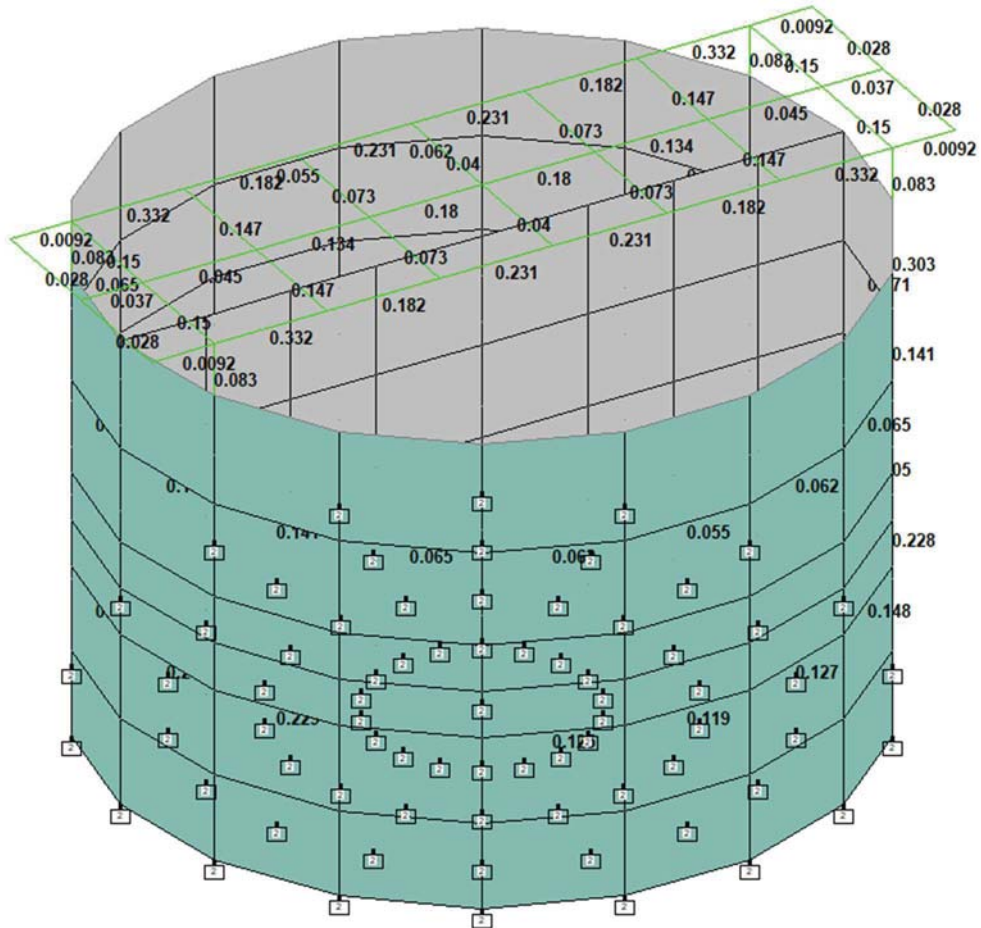
**Maximum stress Sx occurred due combination load 206 : 1.0DL + 0.525 EQ + 0.75LL + 0.75Hidrostatic  
(Load Combination service no 9, Hidrostatic included as live load)**




**Stress due Combination 103 (max : 41.8 MPa)**

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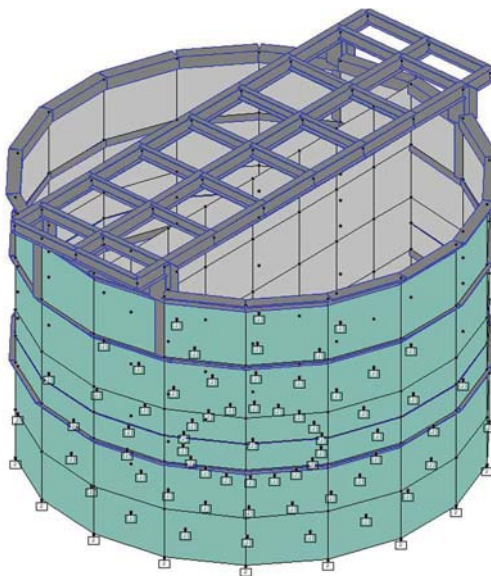
All Stress ratio stiffener frame condition are under allowance (1) :



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## 5. Conclusion



- All component, steel frame and steel shell with current configuration is capable to resist the loads.
- The configuration as follows :



Material			
PLATE			
Shell elev. (m)	Main Thickness (mm)	Corrosion Allowance (mm)	Design Thickness (mm)
Bottom Plate	7	3	10
0 - 4.5	5	3	8
STIFFENER			
Profile elev. (m)	Type		
1.65	L.130.130.12		
3.45	L.130.130.12		
4.5	L.200.200.25		
PLATFORM SUPPORT			
Profile elev. (m)	Type		
4.5	Half Tee. 300.300.10.15		

- Anchorage Data :
  - Diameter = 27 mm
  - Embedded Length = 500 mm (from top of concrete foundation)
  - n = 12



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# AMETANK REPORT

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# AMETank

**Field Erected and Shop Built Storage Tanks  
Engineering Application Software**

## DESIGN DATA

PROJECT NAME	POBOYA 2000 TPD EXPANSION PROJECT	
TAG NO.	4840-TK-120	
DESCRIPTION	TAILING THICKENER OVERFLOW TANK	
CODE	API-650 13TH EDITION	
FLUID NAME	SLURRY / PROCESS WATER	
SPECIFIC GRAVITY	1,11	gr/cm3
DESIGN PRESSURE	FULL OF CONTENT	kPa
DESIGN TEMPERATURE	32	DEG. C
M.D.M.T	0	DEG. C
OPERATING PRESSURE	ATM	kPa
OPERATING TEMP. (MIN/MAX.)	AMB	DEG. C
CORROSION ALLOWANCE	SHELL : 3	MM
	BOTTOM : 3	MM
	STRUCTURE : 3	MM
RADIOGRAPHY / JOINT EFICIENCY	SPOT / 0.85	
P.W.H.T	NO	
INSULATION THICKNESS	NO	MM
DESIGN WIND VELOCITY	40	m/s
WIND & SEISMIC CODE	API-650 - ASCE7	
MATERIAL	SHELL : ASTM A36	
	ROOF : -	
	BOTTOM : ASTM A36	
	NOZZLE (PIPE/PLATE) : ASTM A106 Gr.B / ASTM A36	
	FLANGE : ASTM A36 / ASTM A105	
	NOZZLE PAD : ASTM A36	
	TOP ANGLE : ASTM A36	
M.A.W.P	-	
HYDRO. TEST PRESSURE	FULL OF WATER	kPa
PNEUMATIC TEST PRESSURE	-	kPa

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# Project Design Data and Summary

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## Project Data

Designer : Wawan

Tag ID : 4840-TK-120

Site : Poboya

Design Basis : API-650 13th Edition Errata 1, 2021

## Design Parameters and Operating Conditions

### Design Parameters

Design Internal Pressure = 0 KPa or 0 mmh<sub>2</sub>o

Design External Pressure = -0 KPa or -0 mmh<sub>2</sub>o

D of Tank = 7 m

OD of Tank = 7.02 m

ID of Tank = 7 m

CL of Tank = 7.01 m

Shell Height = 4.5 m

S.G of Contents = 1.11

S.G of Hydrotest = 1.02

Max Design Liq. Level = 3.71 m

Max Operating Liq. Level = 3.71 m

Min Liq. Level = 0 m

Design Temperature = 32 °C

MDMT (Minimum Design Metal Temperature) = 0 °C

Tank Joint Efficiency = 0.85

Ground Snow Load = 0 KPa

Roof Live Load = 1.17 KPa

Additional Roof Dead Load = 0 KPa

Wind Load Basis: ASCE7-10

3 Second Gust Wind Speed (entered),  $V_g = 40 \text{ m/s} = 144 \text{ kph}$

Wind Importance Factor,  $I_w = 1.5$

Design Wind Speed,  $V = V_g * 0.78 = 112.32 \text{ kph}$

Seismic Method: API-650 - ASCE7 Mapped( $S_s$  &  $S_1$ )

Seismic Use Group = III

Site Class = D

$T_L$  (sec) = 4

$S_s$  (g) = 1.2

$S_1$  (g) = 0.5

$A_v$  (g) = 0.38

$Q = 0.67$

Importance Factor = 1.5

## Design Remarks



# Summary Results

## Shell

Shell #	Width (mm)	Material	CA (mm)	JE	Min Yield Strength (MPa)	Tensile Strength (MPa)	Sd (MPa)	St (MPa)	Weight (kg)
1	1828	A36	3	0.70	250	400	145	145	2,518
2	1828	A36	3	0.70	250	400	145	145	2,518
3	844	A36	3	0.70	250	400	145	145	1,162

(continued)

Shell #	Weight CA (kg)	t-min Erection (mm)	t-Des (mm)	t-Test (mm)	t-min Seismic (mm)	t-min Ext-Pe (mm)	t-min (mm)	t-Actual (mm)	Status
1	1,574	6	4.28	1.18	4.16	NA	6	8	OK
2	1,574	5	3.59	0.55	3.64	NA	5	8	OK
3	727	5	2.91	0	3.14	NA	5	8	OK

Total Weight of Shell = 6,213.87 kg

## Stiffeners

Stiffener #	Size	Elevation (m)	Z-Req'd (cm3)	Z-Actual (cm3)	Weight (kg)
1	L75X75X6	1.65	2.21	20.65	154.21
2	L75X75X6	3.45	1.29	20.65	154.21

## Roof

Type = No Roof

Plates Material = None

Structure Support Type = None

## Bottom

Type : Flat Bottom Non Annular

Bottom Material = A36

t.required = 9 mm

t.actual = 10 mm

Bottom corrosion allowance = 3 mm

Bottom Joint Efficiency = 0.7

Total Weight of Bottom = 3,118.01 kg

## Top Member

Type = Detail B

Size = L130x130x12

Material = A36

Weight = 517.97 kg

## Anchors

Quantity = 12

Size = M27 mm

Material = A307-B

Bolt Hole Circle Radius = 3.59 m

## Nameplate Information

Pressure Combination Factor	0.4
Design Standard	API-650 13th Edition Errata 1, 2021
Appendices Used	A, E
Shell (1)	A36 : 8 mm
Shell (2)	A36 : 8 mm
Shell (3)	A36 : 8 mm
Bottom	A36 : 10 mm

# Roof Design Details [Back](#)

Roof Type (Open Top) = Open Top

# Top Member Detail B Design [Back](#)

## Material Properties

Material (A36) = A36

Minimum Tensile Strength (Sut) = 400.0 MPa

Minimum Yield Strength (Sy) = 250.0 MPa

## Compression Ring Detail b Properties

A\_detail = Detail Total Area (mm<sup>2</sup>)

A\_roof = Contributing Roof Area (mm<sup>2</sup>)

A\_shell = Contributing Shell Area (mm<sup>2</sup>)

I\_shell = Contributing Shell Moment Of Inertia (mm<sup>4</sup>)

Wc = Maximum Width of Participating Shell per API-650 Figure F-2 (mm)

$$Wc = 0.6 * \text{SQRT}(((ID / 2) * (tc\text{-nominal} - CA\text{-shell})))$$

$$Wc = 0.6 * \text{SQRT}(((7,000 / 2) * (8 - 3)))$$

$$Wc = 79.37 \text{ mm}$$

## Angle Size L130X130X12 Section Properties

Description	Variable	New	Corroded	Unit
Weight	W	23.5	11.47	kg/m
Cross Sectional Area	A	3,000	1,463.71	mm <sup>2</sup>
Moment Of Inertia About X Axis	Ix	4,719,999.0	2,182,294.92	mm <sup>4</sup>
Moment Of Inertia About Y Axis	Iy	4,719,999.0	2,182,294.92	mm <sup>4</sup>
Section Modulus About X Axis	Sx	50,400	23,893.96	mm <sup>3</sup>
Section Modulus About Y Axis	Sy	50,400	23,893.96	mm <sup>3</sup>
Centroid X Coords	cx	36.4	36.4	mm
Centroid Y Coords	cy	36.4	36.4	mm
Angle Long Leg Length	L1-angle	130	124	mm
Angle Short Leg Length	L2-angle	130	124	mm
Angle Thickness	t-angle	12.0	6.0	mm
<ul style="list-style-type: none"><li>Corrosion allowance, 3 mm, is applied to all sides of the structural member.</li></ul>				

$$I_{\text{shell}} = ((Wc - h) * ((tc\text{-nominal} - CA\text{-shell})^3)) / 12$$

$$I_{\text{shell}} = ((79.3725 - 8.0) * ((8 - 3)^3)) / 12$$

$$I_{\text{shell}} = 743.46 \text{ mm}^4$$

$$A_{\text{shell}} = (Wc - h) * (tc\text{-nominal} - CA\text{-shell})$$

$$A_{\text{shell}} = (79.3725 - 8.0) * (8 - 3)$$

$$A_{\text{shell}} = 356.86 \text{ mm}^2$$

$$A_{\text{roof}} = 0$$

$$A_{\text{roof}} = 0$$

$$A_{\text{roof}} = 0 \text{ mm}^2$$

$$A_{\text{detail}} = A_{\text{shell}} + A_{\text{roof}} + A\text{-corr}$$

$$A_{\text{detail}} = 356.8627 + 0 + 1,463.7097$$

$$A_{\text{detail}} = 1,820.57 \text{ mm}^2$$

### Stiffener and Shell Combined Section Properties

Description	Variable	Equation	Value	Unit
Shell centroid	d_shell	(tc-nominal - CA-shell) / 2	2.5	mm
Stiffener centroid	d_stiff	cy + (tc-nominal - CA-shell)	41.4	mm
moment of inertia of first body	I_1	Ic + (Area * (Distance^2))	4,691,034.76	mm^4
moment of inertia of second body	I_2	Ic + (Area * (Distance^2))	2,973.86	mm^4
Total area	A_sum	A_1 + A_2	1,820.57	mm^2
Sum of moments of inertia's	I_sum	I_1 + I_2	4,694,008.61	mm^4
Combined centroid	c_combined	((Centroid-1 * Area-1) + (Centroid-2 * Area-2)) / (Area-1 + Area-2)	33.77	mm
Combined moment of inertia	I_combined	Ic - (Area * (Distance^2))	2,617,195.93	mm^4
Distance from neutral axis to edge 1 (inside)	e1	c_combined	33.77	mm
Distance from neutral axis to edge 2 (outside)	e2	((tc-nominal - CA-shell) + L1-angle) - e1	101.23	mm
Combined stiffener shell section modulus	S	I / MAX(d-1 , d-2)	25,855.22	mm^3

### Erection Requirement

As per API-650 5.9.5.2, Minimum Size of Top Corner Ring (Size-min) = L75x75x6

Minimum Section Modulus per Erection Requirement (Sx-min) = 8.41 cm^3

Sx >= Sx-min ==> PASS

### Top Wind Girder Design

Fy = Least Minimum Yield Strength of Shell and Stiffening Ring Material at Maximum Temperature per API-650 5.9.5.3 (MPa)

Pwd = Design Wind Pressure Including Inward Drag per API-650 5.9.5.3 (kPa)

Pwv = Wind Pressure where Design Wind Speed V is Used per API-650 5.9.5.3 (kPa)

Wc = Shell Contributing Length (mm)

Z\_req = Required minimum section modulus of the stiffening ring per API-650 5.9.5.3 (cm^3)

tc = Thickness Of Shell Plate (mm)

tc = 8 mm

Wc = (2 \* 16 \* tc) + (t - CA)

Wc = (2 \* 16 \* 8) + (0 - nil)

Wc = 256 mm

Pwv = 1.48 \* ((V / 190)^2)

Pwv = 1.48 \* ((112.32 / 190)^2)

Pwv = 0.52 kPa

Pwd = Pwv + 0.24

Pwd = 0.5172 + 0.24

Pwd = 0.76 kPa

$F_y = \text{MIN}(S_y, S_{y\text{-shell}})$   
 $F_y = \text{MIN}(250.0, 250.0)$   
 $F_y = 250.0$  (Set to 210 MPa since it cannot be greater than 210)

$Z_{\text{req}} = ((6 * H^2 * (D^2)) / (0.5 * F_y)) * (P_{\text{wd}} / 1.72)$   
 $Z_{\text{req}} = ((6 * 4.5 * (7.0^2)) / (0.5 * 210)) * (0.7572 / 1.72)$   
 $Z_{\text{req}} = 5.55 \text{ cm}^3$

$S \geq Z_{\text{req}} \Rightarrow \text{PASS}$

# Agitator Bridge Design [Back](#)

A-es-req = End Support Required Cross Sectional Area (mm<sup>2</sup>)  
A-platform = Platform Total Area (m<sup>2</sup>)  
As-available = End Support Compressive Stress Shell Available Area (mm<sup>2</sup>)  
As-required = Shell Required Compressive Area Around End Supports (mm<sup>2</sup>)  
CA-es = End Support Corrosion Allowance (mm)  
CA-ps = Primary Support Corrosion Allowance (mm)  
Fa-leg = Total Downward Force per Leg including Dead Load, Live Load, and Mixer Dynamic Loads, and Leg Weight (N)  
Fa-total = Total Downward Force including Dead Load, Live Load, and Mixer Dynamic Loads (N)  
L-es = End Support Length (mm)  
L-es-min = End Support Required Length (mm)  
L-es-min-overlap = End Support Shell Minimum Overlap (mm)  
L-es-shell-overlap = End Support Shell Overlap (mm)  
L-ps = Primary Support Length (m)  
L-ps-unbraced = Platform Primary Unbraced Length (mm)  
LL-platform = Platform Live Load (kPa)  
M-max-support = Primary Support Maximum Bending Moment (N.m)  
Ma-es = End Support Material  
Ma-ps = Primary Support Material  
Mx-es = End Support Bending Moment About X Axis (N.mm)  
My-es = End Support Bending Moment About Y Axis (N.mm)  
Pd = Primary Support Agitator Dynamic Point Load (N)  
Ps = Primary Support Agitator Static Point Load (N)  
Psd = Primary Support Agitator Center Load (N)  
Sb-allowable-support = Primary Support Allowable Bending Stress (MPa)  
Scs = Maximum Allowable Compressive Stress per API-620 5.5.4.2 (MPa)  
Sx-reqd-support = Primary Support Required Section Modulus (mm<sup>3</sup>)  
V-max-support = Primary Support Maximum Shear (N)  
W-add-platform = Platform Additional Weight (kg)  
W-platform = Platform Weight (kg)  
W-support = Primary Support Uniform Load (N/m)  
W-total-platform = Platform Total Weight (kg)  
d-ps = Primary Supports Distance (platform width) (mm)  
delta-d = Primary Support Static + Dynamic Loads Deflection (mm)  
delta-d-allowable = Primary Support Dynamic Load Allowable Deflection (mm)  
delta-sd = Primary Support Static + Dynamic Actual Deflection (mm)  
delta-sd-allowable = Primary Support Static + Dynamic Allowable Deflection (mm)  
fb-es = End Support Total Bending Stress (MPa)  
fb-es-req = End Support Allowable Bending Stress (MPa)  
fbx-es = End Support Bending Stress About X Axis (MPa)  
fby-es = End Support Bending Stress About Y Axis (MPa)  
fc-es = End Support Compressive Stress (MPa)  
fc-es-req = End Support Allowable Compressive Stress (MPa)  
k = End Support Effective Length Factor  
l-lateral-platform = Lateral Platform Length (mm)  
outside-proj-1-ps = Platform Outside Projection Side 2 (mm)  
outside-proj-2-ps = Platform Outside Projection Side 1 (mm)  
size-es = End Support Size  
size-ps = Primary Support Size  
t/R = Thickness to Tank Radius Ratio  
w-lateral-platform = Lateral Platform Width (mm)

CA-es = 0 mm  
CA-ps = 0 mm  
L-es = 800 mm  
L-es-shell-overlap = 300 mm  
L-ps-unbraced = 1,030 mm

LL-platform = 0.01 kPa  
 Ma-es = A36  
 Ma-ps = A36  
 W-add-platform = 0 kg  
 W-platform = 1,192.06 kg  
 d-ps = 2,035 mm  
 k = 2.1  
 l-lateral-platform = 8,305 mm  
 outside-proj-1-ps = 381 mm  
 outside-proj-2-ps = 381 mm  
 size-es = h300x300  
 size-ps = h300x150  
 w-lateral-platform = 0 mm

L-ps = OD  
 L-ps = 7.016  
 L-ps = 7.02 m

$A\text{-platform} = (L\text{-ps} * d\text{-ps}) + (2 * w\text{-lateral-platform} * l\text{-lateral-platform})$   
 $A\text{-platform} = (7.016 * 2.035) + (2 * 0.0 * 8.305)$   
 $A\text{-platform} = 14.28 \text{ m}^2$

$W\text{-total-platform} = W\text{-platform} + W\text{-add-platform}$   
 $W\text{-total-platform} = 1,192.0605 + 0$   
 $W\text{-total-platform} = 1,192.06 \text{ kg}$

#### **Agitator Bridge Primary Support Design** **Agitator Bridge Primary Support Material Properties**

Material (A36) = A36  
 Minimum Tensile Strength (Sut) = 400.0 MPa  
 Minimum Yield Strength (Sy) = 250.0 MPa  
 Modulus of Elasticity at Design Temperature (E) = 199,000 MPa

#### **Agitator Bridge Primary Support I-Beam Size H300X150 Section Properties**

Description	Variable	New	Corroded	Unit
Weight	W	36.7	36.7	kg/m
Cross Sectional Area	A	4,678	4,678	mm <sup>2</sup>
Radius of Gyration About X Axis	rx	124.0	124.0	mm
Radius of Gyration About Y Axis	ry	32.9	32.9	mm
Moment Of Inertia About X Axis	Ix	72,100,000	72,100,000	mm <sup>4</sup>
Moment Of Inertia About Y Axis	Iy	5,080,000	5,080,000	mm <sup>4</sup>
Section Modulus About X Axis	Sx	481,000	481,000	mm <sup>3</sup>
Section Modulus About Y Axis	Sy	67,700	67,700	mm <sup>3</sup>
Plastic Section Modulus About X Axis	Zx	542,000	542,000	mm <sup>3</sup>
Plastic Section Modulus About Y Axis	Zy	105,000	105,000	mm <sup>3</sup>
Warping Constant	cw	107,000,000,000	107,000,000,000	mm <sup>6</sup>
Torsional Constant	j	124,000	124,000	mm <sup>4</sup>
Centroid to Edge Max x Distance	ex	75.0	75.0	mm
Centroid to Edge Max y Distance	ey	150.0	150.0	mm
I-Beam Flange Width	wf	150.0	150.0	mm
I-Beam Flange Thickness	tf	9.0	9.0	mm
I-Beam Depth	d	300.0	300.0	mm



I-Beam Web Thickness	tw	6.5	6.5	mm
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### Agitator Bridge Primary Support Allowable Flexural Strength per AISC-360

$L_p$  = Limiting laterally unbraced length for the limit state of yielding per AISC-360 F2-5 (mm)

$M_a$  = Allowable Flexural Strength (N.mm)

$M_n$  = Nominal flexural strength per AISC-360 F2 (N.mm)

$M_p$  = Yielding per AISC-360 F2-1 (N.mm)

$M_{pa}$  = Allowable Flexural Strength Assuming the Member is Braced (N.mm)

$Y_{pf}$  = Limiting slenderness parameter for compact flange

$Y_{pw}$  = Limiting slenderness parameter for compact web

$Y_{rf}$  = Limiting slenderness parameter for noncompact flange

$Y_{rw}$  = Limiting slenderness parameter for noncompact web

$b_f$  = Flange width (mm)

$h$  = Web height (mm)

$$b_f = w_f / 2 = 75.0 \text{ mm}$$

$$h = d - (2 * t_f) = 282.0 \text{ mm}$$

$$Y_{pf} = 0.38 * \text{SQRT}((E / F_y))$$

$$Y_{pf} = 0.38 * \text{SQRT}((199,000 / 250.0))$$

$$Y_{pf} = 10.72$$

$$Y_{rf} = 1 * \text{SQRT}((E / F_y))$$

$$Y_{rf} = 1 * \text{SQRT}((199,000 / 250.0))$$

$$Y_{rf} = 28.21$$

$$Y_{pw} = 3.76 * \text{SQRT}((E / F_y))$$

$$Y_{pw} = 3.76 * \text{SQRT}((199,000 / 250.0))$$

$$Y_{pw} = 106.08$$

$$Y_{rw} = 5.7 * \text{SQRT}((E / F_y))$$

$$Y_{rw} = 5.7 * \text{SQRT}((199,000 / 250.0))$$

$$Y_{rw} = 160.82$$

As per AISC-360 table B4.1b Flange width to thickness ratio check :

$$(b_f / t_f) \leq Y_{pf}$$

==> Flange is compact

As per AISC-360 table B4.1b Web height to thickness ratio check :

$$(h / t_w) \leq Y_{pw}$$

==> Web is compact

$$M_p = F_y * Z_x$$

$$M_p = 250.0 * 542,000$$

$$M_p = 135,500,000 \text{ N.mm}$$

Unbraced length (Lb) = 1,030 mm

$L_p = 1.76 * r_y * \text{SQRT}((E / F_y))$   
 $L_p = 1.76 * 32.9 * \text{SQRT}((199,000 / 250.0))$   
 $L_p = 1,633.67 \text{ mm}$

$L_b \leq L_p$

$M_{pa} = M_p / 1.67$   
 $M_{pa} = 135,500,000 / 1.67$   
 $M_{pa} = 81,137,724.55 \text{ N.mm}$

$M_n = (M_p) = 135,500,000 \text{ N.mm}$

$M_a = M_n / 1.67$   
 $M_a = 135,500,000 / 1.67$   
 $M_a = 81,137,724.55 \text{ N.mm}$

#### **Agitator Bridge Primary Support Loads**

$P_s = W_{\text{-agitator}} / 2$   
 $P_s = 0 / 2$   
 $P_s = 0 \text{ N}$

$P_d = (F_a + (M_x / (0.5 * d_{\text{-ps}}))) / 2$   
 $P_d = (0 + (0.0 / (0.5 * 2,035))) / 2$   
 $P_d = 0.0 \text{ N}$

$P_{sd} = P_s + P_d$   
 $P_{sd} = 0 + 0.0$   
 $P_{sd} = 0.0 \text{ N}$

$W_{\text{-support}} = W_{\text{-primary-support}} + ((0.5 * (W_{\text{-total-platform}} + (A_{\text{-platform}} * LL_{\text{-platform}}))) / L_{\text{-ps}})$   
 $W_{\text{-support}} = 359.9041 + ((0.5 * (11,690.1196 + (14.2776 * 11.5))) / 7.016)$   
 $W_{\text{-support}} = 1,204.71 \text{ N/m}$

$M_{\text{-max-support}} = ((W_{\text{-support}} * (L_{\text{-ps}}^2)) / 8) + ((P_{sd} * L_{\text{-ps}}) / 4)$   
 $M_{\text{-max-support}} = ((1,204.7096 * (7.016^2)) / 8) + ((0.0 * 7.016) / 4)$   
 $M_{\text{-max-support}} = 7,412.62 \text{ N.m}$

$V_{\text{-max-support}} = ((W_{\text{-support}} * (L_{\text{-ps}} + \text{outside-proj-1-ps} + \text{outside-proj-2-ps})) + (M_y / (L_{\text{-ps}} / 2)) + P_{sd}) / 2$   
 $V_{\text{-max-support}} = ((1,204.7096 * (7.016 + 0.381 + 0.381)) + (0.0 / (7,016 / 2)) + 0.0) / 2$   
 $V_{\text{-max-support}} = 4,685.12 \text{ N}$

#### **Agitator Bridge Primary Support Required Section Modulus**

$S_{b\text{-allowable-support}} = M_a / S_{x\text{-primary-support}}$   
 $S_{b\text{-allowable-support}} = 81,137,724.5509 / 481,000$   
 $S_{b\text{-allowable-support}} = 168.69 \text{ MPa}$

$Sx_{reqd-support} = M_{max-support} / S_{b-allowable-support}$   
 $Sx_{reqd-support} = 7,412,616.7912 / 168.6855$   
 $Sx_{reqd-support} = 43,943.41 \text{ mm}^3$

$Sx_{primary-support} \geq Sx_{reqd-support} \implies \text{PASS}$

#### **Agitator Bridge Primary Support Maximum Allowable Deflection**

$\Delta_{sd} = ((5 * W_{support} * (L_{ps}^4)) / (384 * E * I_{x-primary-support})) + ((P_{sd} * (L_{ps}^3)) / (48 * E * I_{x-primary-support}))$   
 $\Delta_{sd} = ((5 * 1.2047 * (7,016^4)) / (384 * 199,000 * 72,100,000)) + ((0.0 * (7,016^3)) / (48 * 199,000 * 72,100,000))$   
 $\Delta_{sd} = 2.65 \text{ mm}$

$\Delta_{sd-allowable} = L_{ps} / 240$   
 $\Delta_{sd-allowable} = 7,016 / 240$   
 $\Delta_{sd-allowable} = 29.23 \text{ mm}$

$\Delta_{sd} \leq \Delta_{sd-allowable} \implies \text{PASS}$

$\Delta_d = (P_d * (L_{ps}^3)) / (48 * E * I_{x-primary-support})$   
 $\Delta_d = (0.0 * (7,016^3)) / (48 * 199,000 * 72,100,000)$   
 $\Delta_d = 0.0 \text{ mm}$

$\Delta_{d-allowable} = 0.5 * L_{ps} * \text{TAN}(0.1)$   
 $\Delta_{d-allowable} = 0.5 * 7,016 * \text{TAN}(0.1)$   
 $\Delta_{d-allowable} = 6.12 \text{ mm}$

$\Delta_d \leq \Delta_{d-allowable} \implies \text{PASS}$

#### **Agitator Bridge End Support Design**

##### **Agitator Bridge End Support Material Properties**

Material (A36) = A36  
 Minimum Tensile Strength ( $S_{ut}$ ) = 400.0 MPa  
 Minimum Yield Strength ( $S_y$ ) = 250.0 MPa  
 Modulus of Elasticity at Design Temperature ( $E$ ) = 199,000 MPa

##### **Agitator Bridge End Support I-Beam Size H300X300 Section Properties**

Description	Variable	New	Corroded	Unit
Weight	W	94.0	94.0	kg/m
Cross Sectional Area	A	11,980	11,980	mm <sup>2</sup>
Radius of Gyration About X Axis	rx	131.0	131.0	mm
Radius of Gyration About Y Axis	ry	75.1	75.1	mm
Moment Of Inertia About X Axis	Ix	204,000,000	204,000,000	mm <sup>4</sup>
Moment Of Inertia About Y Axis	Iy	67,500,000	67,500,000	mm <sup>4</sup>
Section Modulus About X Axis	Sx	1,360,000	1,360,000	mm <sup>3</sup>
Section Modulus About Y Axis	Sy	450,000	450,000	mm <sup>3</sup>
Plastic Section Modulus About X Axis	Zx	1,500,000	1,500,000	mm <sup>3</sup>
Plastic Section Modulus About Y Axis	Zy	684,000	684,000	mm <sup>3</sup>

Warping Constant	cw	1,372,000,000,000	1,372,000,000,000	mm^6
Torsional Constant	j	881,000	881,000	mm^4
Centroid to Edge Max x Distance	ex	150.0	150.0	mm
Centroid to Edge Max y Distance	ey	150.0	150.0	mm
I-Beam Flange Width	wf	300.0	300.0	mm
I-Beam Flange Thickness	tf	15.0	15.0	mm
I-Beam Depth	d	300.0	300.0	mm
I-Beam Web Thickness	tw	10.0	10.0	mm

### **Agitator Bridge End Support Allowable Flexural Strength per AISC-360**

$L_p$  = Limiting laterally unbraced length for the limit state of yielding per AISC-360 F2-5 (mm)

$M_a$  = Allowable Flexural Strength (N.mm)

$M_n$  = Nominal flexural strength per AISC-360 F2 (N.mm)

$M_p$  = Yielding per AISC-360 F2-1 (N.mm)

$M_{pa}$  = Allowable Flexural Strength Assuming the Member is Braced (N.mm)

$Y_{pf}$  = Limiting slenderness parameter for compact flange

$Y_{pw}$  = Limiting slenderness parameter for compact web

$Y_{rf}$  = Limiting slenderness parameter for noncompact flange

$Y_{rw}$  = Limiting slenderness parameter for noncompact web

$bf$  = Flange width (mm)

$h$  = Web height (mm)

$$bf = wf / 2 = 150.0 \text{ mm}$$

$$h = d - (2 * tf) = 270.0 \text{ mm}$$

$$Y_{pf} = 0.38 * \text{SQRT}((E / F_y))$$

$$Y_{pf} = 0.38 * \text{SQRT}((199,000 / 250.0))$$

$$Y_{pf} = 10.72$$

$$Y_{rf} = 1 * \text{SQRT}((E / F_y))$$

$$Y_{rf} = 1 * \text{SQRT}((199,000 / 250.0))$$

$$Y_{rf} = 28.21$$

$$Y_{pw} = 3.76 * \text{SQRT}((E / F_y))$$

$$Y_{pw} = 3.76 * \text{SQRT}((199,000 / 250.0))$$

$$Y_{pw} = 106.08$$

$$Y_{rw} = 5.7 * \text{SQRT}((E / F_y))$$

$$Y_{rw} = 5.7 * \text{SQRT}((199,000 / 250.0))$$

$$Y_{rw} = 160.82$$

As per AISC-360 table B4.1b Flange width to thickness ratio check :

$$(bf / tf) \leq Y_{pf}$$

==> Flange is compact

As per AISC-360 table B4.1b Web height to thickness ratio check :  
 $(h / t_w) \leq Y_{pw}$

==> Web is compact

$$\begin{aligned}M_p &= F_y * Z_x \\M_p &= 250.0 * 1,500,000 \\M_p &= 375,000,000 \text{ N.mm}\end{aligned}$$

Unbraced length ( $L_b$ ) = 500 mm

$$\begin{aligned}L_p &= 1.76 * r_y * \text{SQRT}((E / F_y)) \\L_p &= 1.76 * 75.1 * \text{SQRT}((199,000 / 250.0)) \\L_p &= 3,729.14 \text{ mm}\end{aligned}$$

$L_b \leq L_p$

$$\begin{aligned}M_{pa} &= M_p / 1.67 \\M_{pa} &= 375,000,000 / 1.67 \\M_{pa} &= 224,550,898.2 \text{ N.mm}\end{aligned}$$

$$M_n = (M_p) = 375,000,000 \text{ N.mm}$$

$$\begin{aligned}M_a &= M_n / 1.67 \\M_a &= 375,000,000 / 1.67 \\M_a &= 224,550,898.2 \text{ N.mm}\end{aligned}$$

### **Agitator Bridge End Support Allowable Compressive Strength per AISC-360**

$F_e$  = Elastic Buckling Stress per AISC-360 E3-4 (MPa)

$P_a$  = Allowable Compressive Strength (N)

$P_n$  = Nominal compressive strength per AISC-360 E3-1 (N)

$Y_{rf}$  = Limiting slenderness parameter for flanges

$Y_{rw}$  = Limiting slenderness parameter for web

$b_f$  = Flange width (mm)

$h$  = Web height (mm)

Radius of gyration :

$((K * L_y) / r_y) > ((K * L_x) / r_x) \implies$  Radius of gyration about y axis governs

$$\begin{aligned}F_e &= ((\pi^2) * E) / (((K * L_y) / r_y)^2) \\F_e &= ((\pi^2) * 199,000) / (((2.1 * 800) / 75.1)^2) \\F_e &= 3,924.77 \text{ MPa}\end{aligned}$$

$$\begin{aligned}b_f &= w_f / 2 \\b_f &= 300.0 / 2 \\b_f &= 150.0 \text{ mm}\end{aligned}$$

$$\begin{aligned}h &= d - (2 * t_f) \\h &= 300.0 - (2 * 15.0)\end{aligned}$$

$$h = 270.0 \text{ mm}$$

$$Y_{rf} = 0.56 * \text{SQRT}((E / F_y))$$

$$Y_{rf} = 0.56 * \text{SQRT}((199,000 / 250.0))$$

$$Y_{rf} = 15.8$$

$$Y_{rw} = 1.49 * \text{SQRT}((E / F_y))$$

$$Y_{rw} = 1.49 * \text{SQRT}((199,000 / 250.0))$$

$$Y_{rw} = 42.04$$

As per AISC-360 table B4.1a Flange width to thickness ratio check :  
 $(b_f / t_f) \leq Y_{rf} \implies$  Flange is not slender

As per AISC-360 table B4.1a Web height to thickness ratio check :  
 $(h / t_w) \leq Y_{rw} \implies$  Web is not slender

$F_{cr}$  = Critical stress per AISC-360 E3-2 (MPa)

$$(F_y / F_e) \leq 2.25$$

$$F_{cr} = (0.658^{(F_y / F_e)}) * F_y$$

$$F_{cr} = (0.658^{(250.0 / 3,924.7693)}) * 250.0$$

$$F_{cr} = 243.42 \text{ MPa}$$

$$P_n = F_{cr} * A_g$$

$$P_n = 243.4229 * 11,980$$

$$P_n = 2,916,205.85 \text{ N}$$

$$P_a = P_n / 1.67$$

$$P_a = 2,916,205.8549 / 1.67$$

$$P_a = 1,746,231.05 \text{ N}$$

#### **Agitator Bridge End Support Required Length**

$$F_{a\text{-leg}} = V_{\text{max-support}} + (L_{\text{es}} * W_{\text{end-support}})$$

$$F_{a\text{-leg}} = 4,685.1157 + (800 * 0.9218)$$

$$F_{a\text{-leg}} = 5,422.58 \text{ N}$$

$$t/R = (t_s - CA) / (D / 2)$$

$$t/R = (8 - 3) / (7,000 / 2)$$

$$t/R = 0.0014285714285714286$$

$$S_{cs} = 12410.56312772094 * \text{ratio}$$

$$S_{cs} = 12410.56312772094 * 0.0014$$

$$S_{cs} = 17.73 \text{ MPa}$$

$$A_{\text{s-required}} = F_{a\text{-leg}} / S_{cs}$$

$$A_{\text{s-required}} = 5,422.5758 / 17.7294$$

$$A_{\text{s-required}} = 305.85 \text{ mm}^2$$

$$L_{\text{es-min-overlap}} = A_{\text{s-required}} / (2 * (t_s - CA) * \text{TAN}(30))$$

$$L\text{-es-min-overlap} = 305.8526 / (2 * (8 - 3) * \text{TAN}(30))$$

$$L\text{-es-min-overlap} = 52.98 \text{ mm}$$

$$L\text{-es-min} = L\text{-es-min-overlap} + (L\text{-es} - L\text{-es-shell-overlap})$$

$$L\text{-es-min} = 52.9752 + (800 - 300)$$

$$L\text{-es-min} = 552.98 \text{ mm}$$

$$L\text{-es} \geq L\text{-es-min} \implies \text{PASS}$$

#### **Agitator Bridge End Support Shell Reinforcement Requirements**

$(2 * L\text{-es-min-overlap} * \text{TAN}(30)) < d\text{-ps} \implies$  Supports compression zones are not overlapping

$$A\text{s-available} = 2 * L\text{-es-shell-overlap} * \text{TAN}(30) * (t\text{s} - C\text{A})$$

$$A\text{s-available} = 2 * 300 * \text{TAN}(30) * (8 - 3)$$

$$A\text{s-available} = 1,732.05 \text{ mm}^2$$

$$A\text{s-available} \geq A\text{s-required} \implies \text{PASS}$$

#### **Agitator Bridge End Support Required Section Modulus**

$$M\text{x-es} = V\text{-max-support} * e\text{y}$$

$$M\text{x-es} = 4,685.1157 * 150.0$$

$$M\text{x-es} = 702,767.35 \text{ N.mm}$$

$$f\text{bx-es} = M\text{x-es} / S\text{x-end-support}$$

$$f\text{bx-es} = 702,767.352 / 1,360,000$$

$$f\text{bx-es} = 0.52 \text{ MPa}$$

$$M\text{y-es} = (M\text{z} / (0.5 * L\text{-ps})) * (L\text{-es} - L\text{-es-shell-overlap})$$

$$M\text{y-es} = (0.0 / (0.5 * 7,016)) * (800 - 300)$$

$$M\text{y-es} = 0.0 \text{ N.mm}$$

$$f\text{by-es} = M\text{y-es} / S\text{y-end-support}$$

$$f\text{by-es} = 0.0 / 450,000$$

$$f\text{by-es} = 0.0 \text{ MPa}$$

$$f\text{b-es} = f\text{bx-es} + f\text{by-es}$$

$$f\text{b-es} = 0.5167 + 0.0$$

$$f\text{b-es} = 0.52 \text{ MPa}$$

$$f\text{b-es-req} = M\text{a} / S\text{x-end-support}$$

$$f\text{b-es-req} = 224,550,898.2036 / 1,360,000$$

$$f\text{b-es-req} = 165.11 \text{ MPa}$$

$$f\text{b-es} \leq f\text{b-es-req} \implies \text{PASS}$$

#### **Agitator Bridge End Support Required Cross Sectional Area**

$$f\text{c-es} = F\text{a-leg} / A\text{-end-support}$$

$$f\text{c-es} = 5,422.5758 / 11,980$$

$$f\text{c-es} = 0.45 \text{ MPa}$$

$$\begin{aligned}f_c\text{-es-req} &= P_a / A\text{-end-support}\\f_c\text{-es-req} &= 1,746,231.0509 / 11,980\\f_c\text{-es-req} &= 145.76 \text{ MPa}\end{aligned}$$

$$\begin{aligned}A\text{-es-req} &= F_a\text{-leg} / f_c\text{-es-req}\\A\text{-es-req} &= 5,422.5758 / 145.7622\\A\text{-es-req} &= 37.2 \text{ mm}^2\end{aligned}$$

$$A\text{-end-support} \geq A\text{-es-req} \implies \text{PASS}$$

**Agitator Bridge Total Downward Force (Including Dynamic Loads)**

$$\begin{aligned}F_a\text{-total} &= F_a\text{-leg} * 4\\F_a\text{-total} &= 5,422.5758 * 4\\F_a\text{-total} &= 21,690.3 \text{ N}\end{aligned}$$



# Shell Design [Back](#)

Ac = Convective Design Response Spectrum Acceleration Coefficient

Ai = Impulsive Design Response Spectrum Acceleration Coefficient

Av = Vertical ground acceleration coefficient description

CG-shell = Shell center of gravity (m)

D = Tank Nominal Diameter per API-650 5.6.1.1 Note 1 (m)

G = Product Design Specific Gravity

Gt = Hydrotest Specific Gravity

H = Shell height (m)

HL = Max Liquid Level (m)

Pe = Design External Pressure (kPa)

Pi = Design Internal Pressure (kPa)

Rwi = Impulsive Force Reduction Factor

V = Wind velocity (km/hr)

W-ins = Shell Insulation Weight (kg)

W-shell = Shell Nominal Weight (kg)

W-shell-corr = Shell Corroded Weight (kg)

d-ins = Insulation Density (kg/m<sup>3</sup>)

h-min = Minimum Shell Course Height per API-650 5.6.1.2 (mm)

t-ins = Insulation Thickness (mm)

Ac = 0.2

Ai = 0.31

Av = 0.38

D = 7.0 m

G = 1.11

Gt = 1.02

H = 4.5 m

HL = 3.71 m

Pe = 0.0 kPa

Pi = 0.0 kPa

Rwi = 4

V = 112.32 km/hr

d-ins = 130 kg/m<sup>3</sup>

h-min = 800 mm

t-ins = 0 mm

## API-650 Design Method: One Foot (1ft)

Rwi = Impulsive Force Reduction Factor

Rwi = 4

### Course # 1 (bottom course) Design

A1 = Shell Course Cross Sectional Area (mm<sup>2</sup>)

CA = Corrosion allowance per API-650 5.3.2 (mm)

D1 = Shell Course Centerline Diameter (m)

H = Design Liquid Level per API-650 5.6.3.2 (m)

H' = Effective Design Liquid Level per API-650 F.2 (m)

H-max = Maximum Liquid Level for the Installed Thickness (m)

H-max-@-Pi = Maximum Liquid Level for the Installed Thickness @ Design Internal Pressure (m)

Ht' = Effective Hydrostatic Test Liquid Level per API-650 F.2 (m)

JE = Joint efficiency

Ma = Course Material

Pi-max-@-H = Maximum Allowable Internal Pressure for the Installed Thickness @ Design Liquid Level (kPa)

Rwi = Impulsive Force Reduction Factor

W-1 = Shell Course Nominal Weight (kg)  
 W-1-corr = Shell Course Nominal Weight (kg)  
 Ws-tot-top = Top Weight Total (N)  
 h1 = Course Height (m)  
 loc = Course Location (m)  
 t = Installed Thickness (mm)  
 t-min = Minimum Required Thickness (mm)  
 td = Course Design Thickness per A.4.1 (mm)  
 tt = Course Hydrostatic Test Thickness per A.4.1 (mm)

CA = 3 mm  
 H = 3.71 m  
 JE = 0.7  
 Ma = A36  
 Rwi = 4  
 h1 = 1.83 m  
 loc = 0 m  
 t = 8 mm

Shell Course Center of Gravity (CG-1) = 0.91 m

D1 = ID + t  
 D1 = 7.0 + 0.008  
 D1 = 7.01 m

W-1 =  $\pi * D_c * t * h_1 * d$   
 W-1 =  $\pi * 7.008 * 0.008 * 1.828 * 7,840$   
 W-1 = 2,524.21 kg

W-1-corr =  $\pi * D_c * (t - CA) * h_1 * d$   
 W-1-corr =  $\pi * 7.008 * (0.008 - 0.003) * 1.828 * 7,840$   
 W-1-corr = 1,577.63 kg

### Material Properties

Material (A36) = A36  
 Minimum Tensile Strength (Sut) = 400.0 MPa  
 Minimum Yield Strength (Sy) = 250.0 MPa  
 As per API-650 A.4.1, Allowable Design Stress (Sd) = 145 MPa  
 As per API-650 A.4.1, Allowable Hydrostatic Test Stress (St) = 145 MPa  
 Permissible Design Metal Temperature (MDMT-permissible) = -30 °C

### Thickness Required by Erection

As per API-650 5.6.1.1 NOTE 4, Thickness Required by Erection (t-erect) = 6 mm

### Thickness Required by Design

H' = H  
 H' = 3.71  
 H' = 3.71 m

$td = ((4.9 * D * (H' - 0.3) * SG) / (JE * Sd)) + CA$   
 $td = ((4.9 * 7.0 * (3.71 - 0.3) * 1.11) / (0.7 * 145)) + 3$   
 td = 4.28 mm

### Hydrostatic Test Required Thickness

Ht' = H  
 Ht' = 3.71  
 Ht' = 3.71 m

$tt = (4.9 * D * (Ht' - 0.3) * SGt) / (JE * St)$   
 $tt = (4.9 * 7.0 * (3.71 - 0.3) * 1.025) / (0.7 * 145)$

$$tt = 1.18 \text{ mm}$$

### Seismic Design Required Thickness

Nc = Convective Hoop Membrane Unit Force per API 650 Section E.6.1.4 (N/mm)

Nh = Product Hydrostatic Membrane Force per API 650 Section E.6.1.4 and Section 5.6.3.2 (N/mm)

Ni = Impulsive Hoop Membrane Unit Force per API 650 Section E.6.1.4 (N/mm)

Sd-seismic = Maximum Allowable Hoop Tension Membrane Stress per API-650 E.6.2.4 (MPa)

ts = Seismic Minimum Thickness per API 650 Section E.6.2.4 (mm)

As per API 650 Section E.6.1.4, Shell Course Liquid Surface to Analysis Point Distance (Y) = 3.71 m

$$Ni = 8.48 * Ai * G * D * H * ((Y / H) - (0.5 * ((Y / H)^2))) * \tanh((0.866 * (D / H)))$$

$$Ni = 8.48 * 0.306 * 1.11 * 7.0 * 3.71 * ((3.71 / 3.71) - (0.5 * ((3.71 / 3.71)^2))) * \tanh((0.866 * (7.0 / 3.71)))$$

$$Ni = 34.66 \text{ N/mm}$$

$$Nc = (1.85 * Ac * G * (D^2) * \cosh(((3.68 * (H - Y)) / D))) / \cosh(((3.68 * H) / D))$$

$$Nc = (1.85 * 0.2003 * 1.11 * (7.0^2) * \cosh(((3.68 * (3.71 - 3.71)) / 7.0))) / \cosh(((3.68 * 3.71) / 7.0))$$

$$Nc = 5.62 \text{ N/mm}$$

$$Nh = 4.9 * (H - H_{\text{offset}}) * D * G$$

$$Nh = 4.9 * (3.71 - 0) * 7.0 * 1.11$$

$$Nh = 141.25 \text{ N/mm}$$

S\_T+ = Total Combined Hoop Stress per API 650 Sections E.6.1.4, EC.6.1.3 (MPa)

S\_T- = Total Combined Hoop Stress per API 650 Sections E.6.1.4, EC.6.1.3 (MPa)

$$S_{T+} = (Nh + \sqrt{((Ni^2) + (Nc^2) + (((Av * Nh) / 2.5)^2)})) / \max((t - CA), 0.0001)$$

$$S_{T+} = (141.2508 + \sqrt{((34.6566^2) + (5.619^2) + (((0.3808 * 141.2508) / 2.5)^2)})) / \max((8 - 3), 0.0001)$$

$$S_{T+} = 36.49 \text{ MPa}$$

$$S_{T-} = (Nh - \sqrt{((Ni^2) + (Nc^2) + (((Av * Nh) / 2.5)^2)})) / \max((t - CA), 0.0001)$$

$$S_{T-} = (141.2508 - \sqrt{((34.6566^2) + (5.619^2) + (((0.3808 * 141.2508) / 2.5)^2)})) / \max((8 - 3), 0.0001)$$

$$S_{T-} = 20.01 \text{ MPa}$$

$$Sd\text{-seismic} = \min((1.33 * Sd), (0.9 * Fy * E))$$

$$Sd\text{-seismic} = \min((1.33 * 145), (0.9 * 250.0 * 0.7))$$

$$Sd\text{-seismic} = 157.5 \text{ MPa}$$

$$ts = ((\sigma_{At+} * (tn - CA)) / S_{\text{membrane}}) + CA$$

$$ts = ((36.4856 * (8 - 3)) / 157.5) + 3$$

$$ts = 4.16 \text{ mm}$$

### Vertical Axial Load Design (Empty Tank)

Roof Total Weight (Wr-tot) = 0.0 kg

Upper Courses Weight (Ws-pl-top) = 36,183.21 N

$$Ws\text{-tot-top} = Wr\text{-tot} + Wss + Ws\text{-pl-top} + W\text{-1} + Fd_{\text{agitator}}$$

$$Ws\text{-tot-top} = 0.0 + 21,458.4928 + 36,183.2143 + 24,754.0853 + 21,690.303$$

$$Ws\text{-tot-top} = 104,086.1 \text{ N}$$

$$A1 = \pi * ((D1 / 2)^2)$$

$$A1 = \pi * ((7,008 / 2)^2)$$

$$A1 = 38,572,524.87 \text{ mm}^2 [38.57 \text{ m}^2]$$

T1 = Meridional unit force for cylindrical walls T1 per API-620 5.10.2.5.c Equation 10 (lbf/in)

T2 = Latitudinal unit force for cylindrical walls T2 per API-620 5.10.2.5.c Equation 11 (lbf/in)

$$T1 = (Rc / 2) * (P + (W / At))$$

$$T1 = (137.9528 / 2) * (0 + (-23,399.4851 / 59,787.5331))$$

$$T1 = -27.0 \text{ lbf/in} [-4,727.69 \text{ N/m}]$$

$$T2 = P * Rc$$

$$T2 = 0 * 137.9528$$

$$T2 = 0.0 \text{ lbf/in} [0.0 \text{ N/m}]$$

$$(T1 < 0) \text{ AND } (T2 = 0)$$

Thickness calculation based on T1

M = Compression factor  
 Scc = Computed Compressive Stress (psi)  
 Sta = Allowable Tensile Stress per API-620 5.5.3.3 (psi)  
 Stc = Computed Tensile Stress (psi)  
 t\_req = Minimum Required Thickness (in)

As per API-620 Section 5.5.4.5, Figure 5-1 and Figure F-1, Graphical solution (Ratio) = 0.0013591609589041094

$$t\_req = (Ratio * R) + c$$

$$t\_req = (0.0014 * 137.9528) + 0$$

$$t\_req = 0.19 \text{ in} [4.76 \text{ mm}]$$

$$Scc = ABS(T1) / (t\_installed - c)$$

$$Scc = ABS(-26.9958) / (0.315 - 0)$$

$$Scc = 85.71 \text{ psi} [0.59 \text{ MPa}]$$

As per API-620 Section 5.5.4.2, Maximum Allowable Compressive Stress (Scs) = 2,446.49 psi [16.87 MPa]

$$Scc \leq Scs \implies \text{PASS}$$

$$Stc = T2 / (t\_installed - c)$$

$$Stc = 0.0 / (0.315 - 0)$$

$$Stc = 0.0 \text{ psi} [0.0 \text{ kPa}]$$

$$M = Scc / 15000$$

$$M = 85.7116 / 15000$$

$$M = 0.0057141090267562$$

As per API-620 Figure F-1, Tension factor (N) = 1.0

$$Sta = \text{MIN}((Sts * N), (Sts * E))$$

$$Sta = \text{MIN}((21,030.472 * 0.9971), (21,030.472 * 0.7))$$

$$Sta = 14,721.33 \text{ psi} [101,500 \text{ kPa}]$$

$$Stc \leq Sta \implies \text{PASS}$$

#### Minimum Required Thickness

$$t\_min = \text{MAX}(t\_erec, t_d, t_t, t_s, t\_axial-load)$$

$$t\_min = \text{MAX}(6, 4.2791, 1.1812, 4.1583, 4.7625)$$

$$t\_min = 6 \text{ mm}$$

#### Rating of Installed Thickness

$$H\_max = (((t - CA) * Sd * JE) / (4.9 * D * SG)) + 0.3 + loc$$

$$H\_max = (((8 - 3) * 145 * 0.7) / (4.9 * 7.0 * 1.11)) + 0.3 + 0$$

$$H\_max = 13.63 \text{ m}$$

$H_{\text{-max-@-Pi}} = \text{MAX}(H_{\text{-max}}, 0)$   
 $H_{\text{-max-@-Pi}} = \text{MAX}(13.6297, 0)$   
 $H_{\text{-max-@-Pi}} = 13.63 \text{ m}$

$Pi_{\text{-max-@-H}} = \text{MAX}((((H_{\text{-max}} - (H + loc)) * (9.8 * SG)) + P), 0)$   
 $Pi_{\text{-max-@-H}} = \text{MAX}((((13.6297 - (3.71 + 0)) * (9.8 * 1.11)) + 0.0), 0)$   
 $Pi_{\text{-max-@-H}} = 107.91 \text{ kPa}$

### Course # 2 Design

$A_2$  = Shell Course Cross Sectional Area ( $\text{mm}^2$ )  
 $CA$  = Corrosion allowance per API-650 5.3.2 (mm)  
 $D_2$  = Shell Course Centerline Diameter (m)  
 $H$  = Design Liquid Level per API-650 5.6.3.2 (m)  
 $H'$  = Effective Design Liquid Level per API-650 F.2 (m)  
 $H_{\text{-max}}$  = Maximum Liquid Level for the Installed Thickness (m)  
 $H_{\text{-max-@-Pi}}$  = Maximum Liquid Level for the Installed Thickness @ Design Internal Pressure (m)  
 $H_t'$  = Effective Hydrostatic Test Liquid Level per API-650 F.2 (m)  
 $JE$  = Joint efficiency  
 $Ma$  = Course Material  
 $Pi_{\text{-max-@-H}}$  = Maximum Allowable Internal Pressure for the Installed Thickness @ Design Liquid Level (kPa)  
 $R_{wi}$  = Impulsive Force Reduction Factor  
 $W_2$  = Shell Course Nominal Weight (kg)  
 $W_2\text{-corr}$  = Shell Course Nominal Weight (kg)  
 $W_{s\text{-tot-top}}$  = Top Weight Total (N)  
 $h_2$  = Course Height (m)  
 $loc$  = Course Location (m)  
 $t$  = Installed Thickness (mm)  
 $t_{\text{-min}}$  = Minimum Required Thickness (mm)  
 $t_d$  = Course Design Thickness per A.4.1 (mm)  
 $t_t$  = Course Hydrostatic Test Thickness per A.4.1 (mm)

$CA = 3 \text{ mm}$   
 $H = 1.88 \text{ m}$   
 $JE = 0.7$   
 $Ma = A36$   
 $R_{wi} = 4$   
 $h_2 = 1.83 \text{ m}$   
 $loc = 1.83 \text{ m}$   
 $t = 8 \text{ mm}$

Shell Course Center of Gravity (CG-2) = 2.74 m

$D_2 = ID + t$   
 $D_2 = 7.0 + 0.008$   
 $D_2 = 7.01 \text{ m}$

$W_2 = \pi * D_c * t * h_2 * d$   
 $W_2 = \pi * 7.008 * 0.008 * 1.828 * 7,840$   
 $W_2 = 2,524.21 \text{ kg}$

$W_2\text{-corr} = \pi * D_c * (t - CA) * h_2 * d$   
 $W_2\text{-corr} = \pi * 7.008 * (0.008 - 0.003) * 1.828 * 7,840$   
 $W_2\text{-corr} = 1,577.63 \text{ kg}$

### Material Properties

Material (A36) = A36  
Minimum Tensile Strength ( $S_u$ ) = 400.0 MPa  
Minimum Yield Strength ( $S_y$ ) = 250.0 MPa

As per API-650 A.4.1, Allowable Design Stress (Sd) = 145 MPa  
 As per API-650 A.4.1, Allowable Hydrostatic Test Stress (St) = 145 MPa  
 Permissible Design Metal Temperature (MDMT-permissible) = -30 °C

#### Thickness Required by Erection

As per API-650 5.6.1.1, Thickness Required by Erection (t-erec) = 5 mm

#### Thickness Required by Design

H' = H  
 H' = 1.882  
 H' = 1.88 m

$td = ((4.9 * D * (H' - 0.3) * SG) / (JE * Sd)) + CA$   
 $td = ((4.9 * 7.0 * (1.882 - 0.3) * 1.11) / (0.7 * 145)) + 3$   
 td = 3.59 mm

#### Hydrostatic Test Required Thickness

Ht' = H  
 Ht' = 1.882  
 Ht' = 1.88 m

$tt = (4.9 * D * (Ht' - 0.3) * SGt) / (JE * St)$   
 $tt = (4.9 * 7.0 * (1.882 - 0.3) * 1.025) / (0.7 * 145)$   
 tt = 0.55 mm

#### Seismic Design Required Thickness

Nc = Convective Hoop Membrane Unit Force per API 650 Section E.6.1.4 (N/mm)  
 Nh = Product Hydrostatic Membrane Force per API 650 Section E.6.1.4 and Section 5.6.3.2 (N/mm)  
 Ni = Impulsive Hoop Membrane Unit Force per API 650 Section E.6.1.4 (N/mm)  
 Sd-seismic = Maximum Allowable Hoop Tension Membrane Stress per API-650 E.6.2.4 (MPa)  
 ts = Seismic Minimum Thickness per API 650 Section E.6.2.4 (mm)

As per API 650 Section E.6.1.4, Shell Course Liquid Surface to Analysis Point Distance (Y) = 1.88 m

$Ni = 8.48 * Ai * G * D * H * ((Y / H) - (0.5 * ((Y / H)^2))) * \text{TANH}((0.866 * (D / H)))$   
 $Ni = 8.48 * 0.306 * 1.11 * 7.0 * 3.71 * ((1.882 / 3.71) - (0.5 * ((1.882 / 3.71)^2))) * \text{TANH}((0.866 * (7.0 / 3.71)))$   
 Ni = 26.24 N/mm

$Nc = (1.85 * Ac * G * (D^2) * \text{COSH}(((3.68 * (H - Y)) / D))) / \text{COSH}(((3.68 * H) / D))$   
 $Nc = (1.85 * 0.2003 * 1.11 * (7.0^2) * \text{COSH}(((3.68 * (3.71 - 1.882)) / 7.0))) / \text{COSH}(((3.68 * 3.71) / 7.0))$   
 Nc = 8.42 N/mm

$Nh = 4.9 * (H - H_{\text{offset}}) * D * G$   
 $Nh = 4.9 * (1.882 - 0) * 7.0 * 1.11$   
 Nh = 71.65 N/mm

S\_T+ = Total Combined Hoop Stress per API 650 Sections E.6.1.4, EC.6.1.3 (MPa)  
 S\_T- = Total Combined Hoop Stress per API 650 Sections E.6.1.4, EC.6.1.3 (MPa)

$S_{T+} = (Nh + \text{SQRT}(((Ni^2) + (Nc^2) + (((Av * Nh) / 2.5)^2)))) / \text{MAX}((t - CA), 0.0001)$   
 $S_{T+} = (71.6534 + \text{SQRT}(((26.2428^2) + (8.4196^2) + (((0.3808 * 71.6534) / 2.5)^2)))) / \text{MAX}((8 - 3), 0.0001)$   
 S\_T+ = 20.26 MPa

$S_{T-} = (Nh - \text{SQRT}(((Ni^2) + (Nc^2) + (((Av * Nh) / 2.5)^2)))) / \text{MAX}((t - CA), 0.0001)$   
 $S_{T-} = (71.6534 - \text{SQRT}(((26.2428^2) + (8.4196^2) + (((0.3808 * 71.6534) / 2.5)^2)))) / \text{MAX}((8 - 3), 0.0001)$   
 S\_T- = 8.4 MPa

$Sd-seismic = \text{MIN}((1.33 * Sd), (0.9 * F_y * E))$   
 $Sd-seismic = \text{MIN}((1.33 * 145), (0.9 * 250.0 * 0.7))$   
 $Sd-seismic = 157.5 \text{ MPa}$

$t_s = ((\text{SIGMA}_{t+} * (t_n - CA)) / S_{\text{membrane}}) + CA$   
 $t_s = ((20.2592 * (8 - 3)) / 157.5) + 3$   
 $t_s = 3.64 \text{ mm}$

### Vertical Axial Load Design (Empty Tank)

Roof Total Weight ( $W_{r-tot}$ ) = 0.0 kg  
 Upper Courses Weight ( $W_{s-pl-top}$ ) = 11,429.13 N

$W_{s-tot-top} = W_{r-tot} + W_{ss} + W_{s-pl-top} + W-2 + F_{d\_agitator}$   
 $W_{s-tot-top} = 0.0 + 21,458.4928 + 11,429.1291 + 24,754.0853 + 21,690.303$   
 $W_{s-tot-top} = 79,332.01 \text{ N}$

$A_2 = \pi * ((D_2 / 2)^2)$   
 $A_2 = \pi * ((7,008 / 2)^2)$   
 $A_2 = 38,572,524.87 \text{ mm}^2 [38.57 \text{ m}^2]$

$T_1$  = Meridional unit force for cylindrical walls  $T_1$  per API-620 5.10.2.5.c Equation 10 (lbf/in)  
 $T_2$  = Latitudinal unit force for cylindrical walls  $T_2$  per API-620 5.10.2.5.c Equation 11 (lbf/in)

$T_1 = (R_c / 2) * (P + (W / A_t))$   
 $T_1 = (137.9528 / 2) * (0 + (-17,834.5454 / 59,787.5331))$   
 $T_1 = -20.58 \text{ lbf/in} [-3,603.33 \text{ N/m}]$

$T_2 = P * R_c$   
 $T_2 = 0 * 137.9528$   
 $T_2 = 0.0 \text{ lbf/in} [0.0 \text{ N/m}]$

$(T_1 < 0) \text{ AND } (T_2 = 0)$

Thickness calculation based on  $T_1$

$M$  = Compression factor  
 $S_{cc}$  = Computed Compressive Stress (psi)  
 $S_{ta}$  = Allowable Tensile Stress per API-620 5.5.3.3 (psi)  
 $S_{tc}$  = Computed Tensile Stress (psi)  
 $t_{req}$  = Minimum Required Thickness (in)

As per API-620 Section 5.5.4.5, Figure 5-1 and Figure F-1, Graphical solution (Ratio) =  
 0.0013591609589041094

$t_{req} = (\text{Ratio} * R) + c$   
 $t_{req} = (0.0014 * 137.9528) + 0$   
 $t_{req} = 0.19 \text{ in} [4.76 \text{ mm}]$

$S_{cc} = \text{ABS}(T_1) / (t_{\text{installed}} - c)$   
 $S_{cc} = \text{ABS}(-20.5756) / (0.315 - 0)$   
 $S_{cc} = 65.33 \text{ psi} [0.45 \text{ MPa}]$

As per API-620 Section 5.5.4.2, Maximum Allowable Compressive Stress ( $S_{cs}$ ) = 2,446.49 psi [16.87 MPa]

$S_{cc} \leq S_{cs} \implies \text{PASS}$

$S_{tc} = T_2 / (t_{\text{installed}} - c)$

$$Stc = 0.0 / (0.315 - 0)$$

$$Stc = 0.0 \text{ psi } [0.0 \text{ kPa}]$$

$$M = Scc / 15000$$

$$M = 65.3274 / 15000$$

$$M = 0.004355161499603866$$

As per API-620 Figure F-1, Tension factor (N) = 1.0

$$Sta = \text{MIN}((Sts * N), (Sts * E))$$

$$Sta = \text{MIN}((21,030.472 * 0.9978), (21,030.472 * 0.7))$$

$$Sta = 14,721.33 \text{ psi } [101,500 \text{ kPa}]$$

Stc <= Sta ==> PASS

### Minimum Required Thickness

$$t\text{-min} = \text{MAX}(t\text{-erec}, td, tt, ts, t\text{-axial-load})$$

$$t\text{-min} = \text{MAX}(5, 3.5934, 0.548, 3.6432, 4.7625)$$

$$t\text{-min} = 5 \text{ mm}$$

### Rating of Installed Thickness

$$H\text{-max} = (((t - CA) * Sd * JE) / (4.9 * D * SG)) + 0.3 + loc$$

$$H\text{-max} = (((8 - 3) * 145 * 0.7) / (4.9 * 7.0 * 1.11)) + 0.3 + 1.828$$

$$H\text{-max} = 15.46 \text{ m}$$

$$H\text{-max-@-Pi} = \text{MAX}(H\text{-max}, 0)$$

$$H\text{-max-@-Pi} = \text{MAX}(15.4577, 0)$$

$$H\text{-max-@-Pi} = 15.46 \text{ m}$$

$$Pi\text{-max-@-H} = \text{MAX}((((H\text{-max} - (H + loc)) * (9.8 * SG)) + P), 0)$$

$$Pi\text{-max-@-H} = \text{MAX}((((15.4577 - (1.882 + 1.828)) * (9.8 * 1.11)) + 0.0), 0)$$

$$Pi\text{-max-@-H} = 127.79 \text{ kPa}$$

### Course # 3 Design

$$A3 = \text{Shell Course Cross Sectional Area (mm}^2\text{)}$$

$$CA = \text{Corrosion allowance per API-650 5.3.2 (mm)}$$

$$D3 = \text{Shell Course Centerline Diameter (m)}$$

$$H = \text{Design Liquid Level per API-650 5.6.3.2 (m)}$$

$$H' = \text{Effective Design Liquid Level per API-650 F.2 (m)}$$

$$H\text{-max} = \text{Maximum Liquid Level for the Installed Thickness (m)}$$

$$H\text{-max-@-Pi} = \text{Maximum Liquid Level for the Installed Thickness @ Design Internal Pressure (m)}$$

$$Ht' = \text{Effective Hydrostatic Test Liquid Level per API-650 F.2 (m)}$$

$$JE = \text{Joint efficiency}$$

$$Ma = \text{Course Material}$$

$$Pi\text{-max-@-H} = \text{Maximum Allowable Internal Pressure for the Installed Thickness @ Design Liquid Level (kPa)}$$

$$Rwi = \text{Impulsive Force Reduction Factor}$$

$$W\text{-3} = \text{Shell Course Nominal Weight (kg)}$$

$$W\text{-3-corr} = \text{Shell Course Nominal Weight (kg)}$$

$$Ws\text{-tot-top} = \text{Top Weight Total (N)}$$

$$h3 = \text{Course Height (m)}$$

$$loc = \text{Course Location (m)}$$

$$t = \text{Installed Thickness (mm)}$$

$$t\text{-min} = \text{Minimum Required Thickness (mm)}$$

$$td = \text{Course Design Thickness per A.4.1 (mm)}$$

$$tt = \text{Course Hydrostatic Test Thickness per A.4.1 (mm)}$$

$$CA = 3 \text{ mm}$$

$$H = 0.05 \text{ m}$$

$$JE = 0.7$$



Ma = A36  
Rwi = 4  
h3 = 0.84 m  
loc = 3.66 m  
t = 8 mm

Shell Course Center of Gravity (CG-3) = 4.08 m

D3 = ID + t  
D3 = 7.0 + 0.008  
D3 = 7.01 m

W-3 =  $\pi * D_c * t * h_3 * d$   
W-3 =  $\pi * 7.008 * 0.008 * 0.844 * 7,840$   
W-3 = 1,165.45 kg

W-3-corr =  $\pi * D_c * (t - CA) * h_3 * d$   
W-3-corr =  $\pi * 7.008 * (0.008 - 0.003) * 0.844 * 7,840$   
W-3-corr = 728.4 kg

### Material Properties

Material (A36) = A36  
Minimum Tensile Strength (Sut) = 400.0 MPa  
Minimum Yield Strength (Sy) = 250.0 MPa  
As per API-650 A.4.1, Allowable Design Stress (Sd) = 145 MPa  
As per API-650 A.4.1, Allowable Hydrostatic Test Stress (St) = 145 MPa  
Permissible Design Metal Temperature (MDMT-permissible) = -30 °C

### Thickness Required by Erection

As per API-650 5.6.1.1, Thickness Required by Erection (t-erec) = 5 mm

### Thickness Required by Design

H' = H  
H' = 0.054  
H' = 0.05 m

Design liquid level is below the design point under consideration

$td = ((4.9 * D * (H' - 0.3) * SG) / (JE * Sd)) + CA$   
 $td = ((4.9 * 7.0 * (0.054 - 0.3) * 1.11) / (0.7 * 145)) + 3$   
td = 2.91 mm

### Hydrostatic Test Required Thickness

Ht' = H  
Ht' = 0.054  
Ht' = 0.05 m

Hydrotest Design liquid level is below the design point under consideration

$tt = (4.9 * D * (Ht' - 0.3) * SGt) / (JE * St)$   
 $tt = (4.9 * 7.0 * (0.054 - 0.3) * 1.025) / (0.7 * 145)$   
tt = -0.09 (Set to 0 mm since it cannot be less than 0)

### Seismic Design Required Thickness

Nc = Convective Hoop Membrane Unit Force per API 650 Section E.6.1.4 (N/mm)  
Nh = Product Hydrostatic Membrane Force per API 650 Section E.6.1.4 and Section 5.6.3.2 (N/mm)  
Ni = Impulsive Hoop Membrane Unit Force per API 650 Section E.6.1.4 (N/mm)  
Sd-seismic = Maximum Allowable Hoop Tension Membrane Stress per API-650 E.6.2.4 (MPa)  
ts = Seismic Minimum Thickness per API 650 Section E.6.2.4 (mm)

As per API 650 Section E.6.1.4, Shell Course Liquid Surface to Analysis Point Distance (Y) = 0.05 m

$$\begin{aligned} N_i &= 8.48 * A_i * G * D * H * ((Y / H) - (0.5 * ((Y / H)^2))) * \tanh((0.866 * (D / H))) \\ N_i &= 8.48 * 0.306 * 1.11 * 7.0 * 3.71 * ((0.054 / 3.71) - (0.5 * ((0.054 / 3.71)^2))) * \tanh((0.866 * (7.0 / 3.71))) \\ N_i &= 1.0 \text{ N/mm} \end{aligned}$$

$$\begin{aligned} N_c &= (1.85 * A_c * G * (D^2) * \cosh(((3.68 * (H - Y)) / D))) / \cosh(((3.68 * H) / D)) \\ N_c &= (1.85 * 0.2003 * 1.11 * (7.0^2) * \cosh(((3.68 * (3.71 - 0.054)) / 7.0))) / \cosh(((3.68 * 3.71) / 7.0)) \\ N_c &= 19.61 \text{ N/mm} \end{aligned}$$

$$\begin{aligned} N_h &= 4.9 * (H - H_{\text{offset}}) * D * G \\ N_h &= 4.9 * (0.054 - 0) * 7.0 * 1.11 \\ N_h &= 2.06 \text{ N/mm} \end{aligned}$$

S\_T+ = Total Combined Hoop Stress per API 650 Sections E.6.1.4, EC.6.1.3 (MPa)

S\_T- = Total Combined Hoop Stress per API 650 Sections E.6.1.4, EC.6.1.3 (MPa)

$$\begin{aligned} S_{T+} &= (N_h + \sqrt{((N_i^2) + (N_c^2) + (((A_v * N_h) / 2.5)^2))}) / \max((t - CA), 0.0001) \\ S_{T+} &= (2.0559 + \sqrt{((1.0015^2) + (19.6131^2) + (((0.3808 * 2.0559) / 2.5)^2))}) / \max((8 - 3), 0.0001) \\ S_{T+} &= 4.34 \text{ MPa} \end{aligned}$$

$$\begin{aligned} S_{T-} &= (N_h - \sqrt{((N_i^2) + (N_c^2) + (((A_v * N_h) / 2.5)^2))}) / \max((t - CA), 0.0001) \\ S_{T-} &= (2.0559 - \sqrt{((1.0015^2) + (19.6131^2) + (((0.3808 * 2.0559) / 2.5)^2))}) / \max((8 - 3), 0.0001) \\ S_{T-} &= -3.52 \text{ MPa} \end{aligned}$$

$$\begin{aligned} S_d\text{-seismic} &= \min((1.33 * S_d), (0.9 * F_y * E)) \\ S_d\text{-seismic} &= \min((1.33 * 145), (0.9 * 250.0 * 0.7)) \\ S_d\text{-seismic} &= 157.5 \text{ MPa} \end{aligned}$$

$$\begin{aligned} t_s &= ((\sigma_{T+} * (t_n - CA)) / S_{\text{membrane}}) + CA \\ t_s &= ((4.3394 * (8 - 3)) / 157.5) + 3 \\ t_s &= 3.14 \text{ mm} \end{aligned}$$

### Vertical Axial Load Design (Empty Tank)

Roof Total Weight (W<sub>r-tot</sub>) = 0.0 kg

Upper Courses Weight (W<sub>s-pl-top</sub>) = 0 N

$$\begin{aligned} W_{s\text{-tot-top}} &= W_{r\text{-tot}} + W_{ss} + W_{s\text{-pl-top}} + W_3 + F_{d\_agitator} \\ W_{s\text{-tot-top}} &= 0.0 + 21,458.4928 + 0 + 11,429.1291 + 21,690.303 \\ W_{s\text{-tot-top}} &= 54,577.92 \text{ N} \end{aligned}$$

$$\begin{aligned} A_3 &= \pi * ((D_3 / 2)^2) \\ A_3 &= \pi * ((7,008 / 2)^2) \\ A_3 &= 38,572,524.87 \text{ mm}^2 [38.57 \text{ m}^2] \end{aligned}$$

T1 = Meridional unit force for cylindrical walls T1 per API-620 5.10.2.5.c Equation 10 (lbf/in)

T2 = Latitudinal unit force for cylindrical walls T2 per API-620 5.10.2.5.c Equation 11 (lbf/in)

$$\begin{aligned} T_1 &= (R_c / 2) * (P + (W / A_t)) \\ T_1 &= (137.9528 / 2) * (0 + (-12,269.6056 / 59,787.5331)) \\ T_1 &= -14.16 \text{ lbf/in} [-2,478.98 \text{ N/m}] \end{aligned}$$

$$\begin{aligned} T_2 &= P * R_c \\ T_2 &= 0 * 137.9528 \\ T_2 &= 0.0 \text{ lbf/in} [0.0 \text{ N/m}] \end{aligned}$$

(T1 < 0) AND (T2 = 0)

Thickness calculation based on T1

M = Compression factor  
Scc = Computed Compressive Stress (psi)  
Sta = Allowable Tensile Stress per API-620 5.5.3.3 (psi)  
Stc = Computed Tensile Stress (psi)  
t\_req = Minimum Required Thickness (in)

As per API-620 Section 5.5.4.5, Figure 5-1 and Figure F-1, Graphical solution (Ratio) = 0.0013591609589041094

t\_req = (Ratio \* R) + c  
t\_req = (0.0014 \* 137.9528) + 0  
t\_req = 0.19 in [4.76 mm]

Scc = ABS(T1) / (t-installed - c)  
Scc = ABS(-14.1553) / (0.315 - 0)  
Scc = 44.94 psi [0.31 MPa]

As per API-620 Section 5.5.4.2, Maximum Allowable Compressive Stress (Scs) = 2,446.49 psi [16.87 MPa]

Scc <= Scs ==> PASS

Stc = T2 / (t-installed - c)  
Stc = 0.0 / (0.315 - 0)  
Stc = 0.0 psi [0.0 kPa]

M = Scc / 15000  
M = 44.9432 / 15000  
M = 0.002996213972451531

As per API-620 Figure F-1, Tension factor (N) = 1.0

Sta = MIN((Sts \* N) , (Sts \* E))  
Sta = MIN((21,030.472 \* 0.9985) , (21,030.472 \* 0.7))  
Sta = 14,721.33 psi [101,500 kPa]

Stc <= Sta ==> PASS

#### Minimum Required Thickness

t-min = MAX(t-erec , td , tt , ts , t-axial-load)  
t-min = MAX(5 , 2.9077 , 0 , 3.1378 , 4.7625)  
t-min = 5 mm

#### Rating of Installed Thickness

H-max = (((t - CA) \* Sd \* JE) / (4.9 \* D \* SG)) + 0.3 + loc  
H-max = (((8 - 3) \* 145 \* 0.7) / (4.9 \* 7.0 \* 1.11)) + 0.3 + 3.656  
H-max = 17.29 m

H-max-@-Pi = MAX(H-max , 0)  
H-max-@-Pi = MAX(17.2857 , 0)  
H-max-@-Pi = 17.29 m

Pi-max-@-H = MAX((((H-max - (H + loc)) \* (9.8 \* SG)) + P) , 0)  
Pi-max-@-H = MAX((((17.2857 - (0.054 + 3.656)) \* (9.8 \* 1.11)) + 0.0) , 0)  
Pi-max-@-H = 147.68 kPa

### Shell Design Summary Results

$W_{ins} = t_{ins} * d_{ins} * \pi * (OD + t_{ins}) * H$   
 $W_{ins} = 0.0 * 130 * \pi * (7.016 + 0.0) * 4.5$   
 $W_{ins} = 0.0 \text{ kg}$

$W_{shell-corr} = W-1-corr + W-2-corr + W-3-corr$   
 $W_{shell-corr} = 1,577.6339 + 1,577.6339 + 728.4043$   
 $W_{shell-corr} = 3,883.67 \text{ kg}$

$W_{shell} = W-1 + W-2 + W-3$   
 $W_{shell} = 2,524.2142 + 2,524.2142 + 1,165.4468$   
 $W_{shell} = 6,213.88 \text{ kg}$

$CG_{shell} = ((CG-1 * W-1) + (CG-2 * W-2) + (CG-3 * W-3)) / W_{shell}$   
 $CG_{shell} = ((0.914 * 2,524.2142) + (2.742 * 2,524.2142) + (4.078 * 1,165.4468)) / 6,213.8752$   
 $CG_{shell} = 2.25 \text{ m}$

### Shell Design Summary

Course	Height (m)	Material	CA (mm)	JE	Sy (mpa)	Sut (mpa)	Sd (mpa)	St (mpa)	t-erec (mm)
3	0.84	A36	3	0.7	250.0	400.0	145	145	5
2	1.83	A36	3	0.7	250.0	400.0	145	145	5
1	1.83	A36	3	0.7	250.0	400.0	145	145	6

### Shell Design Summary (continued)

Course	t-design (mm)	t-test (mm)	t-seismic (mm)	t-ext (mm)	t-min (mm)	t-installed (mm)	Status	H-max-@-Pi (m)	Pi-max-@-H (kPa)
3	2.91	0	3.14	N/A	5	8	PASS	17.29	147.68
2	3.59	0.55	3.64	N/A	5	8	PASS	15.46	127.79
1	4.28	1.18	4.16	N/A	6	8	PASS	13.63	107.91

## Intermediate Stiffeners Design

### Stiffeners Design For Wind Loading

D = Nominal Tank Diameter (m)  
H1 = Maximum Unstiffened Transformed Shell Height per API-650 5.9.6.1 (m)  
N = Actual Wind Girders Quantity  
Ns = Required Number of Girders per API 650 5.9.6.3 and 5.9.6.4  
Pwd = Design Wind Pressure Including Inward Drag per API-650 5.9.6.1 (kPa)  
Pwv = Wind Pressure where Design Wind Speed V is Used per API-650 5.9.6.1 (kPa)  
V = Wind velocity (km/hr)  
ts\_min = Thickness of the Thinnest Shell Course (mm)

D = 7.0 m  
N = 2  
V = 112.32 km/hr

Shell Courses Heights (W) = [1.83 1.83 0.84 ] m

ts\_min = MIN(ts\_1 , ts\_2 , ts\_3)  
ts\_min = MIN(8 , 8 , 8)  
ts\_min = 8 mm

### Stiffeners Required Quantity

HTS = Height of Transformed Shell per API 650 5.9.6.2 (m)

**Transformed shell courses heights**

Variable	Equation	Value	Unit		
Wtr_1	$W_1 * \text{SQRT}(((t_{\min} / ts_1)^5))$	1.83	m	N/A	N/A
Wtr_2	$W_2 * \text{SQRT}(((t_{\min} / ts_2)^5))$	1.83	m	N/A	N/A
Wtr_3	$W_3 * \text{SQRT}(((t_{\min} / ts_3)^5))$	0.84	m	N/A	N/A

$$\begin{aligned} \text{HTS} &= \text{Wtr}_1 + \text{Wtr}_2 + \text{Wtr}_3 \\ \text{HTS} &= 1.828 + 1.828 + 0.844 \\ \text{HTS} &= 4.5 \text{ m} \end{aligned}$$

$$\begin{aligned} P_{wv} &= 1.48 * ((V / 190)^2) \\ P_{wv} &= 1.48 * ((112.32 / 190)^2) \\ P_{wv} &= 0.52 \text{ kPa} \end{aligned}$$

$$\begin{aligned} P_{wd} &= P_{wv} + 0.24 \\ P_{wd} &= 0.5172 + 0.24 \\ P_{wd} &= 0.76 \text{ kPa} \end{aligned}$$

$$\begin{aligned} H1 &= 9.47 * ts_{\min} * \text{SQRT}(((ts_{\min} / D)^3)) * (1.72 / P_{wd}) \\ H1 &= 9.47 * 8 * \text{SQRT}(((8 / 7.0)^3)) * (1.72 / 0.7572) \\ H1 &= 210.25 \text{ m} \end{aligned}$$

$$\begin{aligned} N_s &= \text{CEILING}(((\text{HTS} / H_{\text{safe}}) - 1)) \\ N_s &= \text{CEILING}(((4.5 / 210.2513) - 1)) \\ N_s &= 0 \end{aligned}$$

$$N \geq N_s \implies \text{PASS}$$

**Stiffeners Design****Real Elevations**

Stiffener #	Size	Elevation (ft)	Spacing below (ft)	Spacing above (ft)	Average Spacing Ls (ft)
1	L75X75X6	1.65	1.65	1.80	1.73
2	L75X75X6	3.45	1.80	1.05	1.42

**Transformed Elevations**

Stiffener #	Size	Transformed Elevation (ft)	Transformed Spacing below (ft)	Transformed Spacing above (ft)	Transformed Average Spacing (ft)
1	L75X75X6	1.65	1.65	1.80	1.73
2	L75X75X6	3.45	1.80	1.05	1.42

**Stiffener #1 Design**

$F_y$  = Intermediate Wind Girder and Shell Yield Strength per API 650 5.9.6.6 (MPa)  
 $Z_{\text{act}}$  = Stiffener Region Section Modulus per API-650 5.9.6.6.2 (cm<sup>3</sup>)  
 $Z_{\text{req}}$  = Intermediate Wind Girder Minimum Required Section Modulus per API-650 5.9.6.6 (cm<sup>3</sup>)  
 $h_1$  = Vertical Distance Between the Intermediate Stiffener and the Top Angle of the Shell or the Top Wind Girder of an Open-top Tank (m)

$Z_{act} = 20.66 \text{ cm}^3$   
 $h1 = 1.8 \text{ m}$

Shell Yield Strength ( $Fy_{shell}$ ) = 250.0 MPa  
Stiffener Yield Strength ( $Fy_{stiffener}$ ) = 250.0 MPa

$Fy = \text{MIN}(Fy_{stiffener}, Fy_{shell})$   
 $Fy = \text{MIN}(250.0, 250.0)$   
 $Fy = 250.0$  (Set to 210 MPa since it cannot be greater than 210)

$Z_{req} = ((6 * h1 * (D^2)) / (0.5 * Fy)) * (P_{wd} / 1.72)$   
 $Z_{req} = ((6 * 1.8 * (7.0^2)) / (0.5 * 210)) * (0.7572 / 1.72)$   
 $Z_{req} = 2.22 \text{ cm}^3$

$Z_{act} \geq Z_{req}$

### Stiffener #2 Design

$Fy$  = Intermediate Wind Girder and Shell Yield Strength per API 650 5.9.6.6 (MPa)  
 $Z_{act}$  = Stiffener Region Section Modulus per API-650 5.9.6.6.2 ( $\text{cm}^3$ )  
 $Z_{req}$  = Intermediate Wind Girder Minimum Required Section Modulus per API-650 5.9.6.6 ( $\text{cm}^3$ )  
 $h1$  = Vertical Distance Between the Intermediate Stiffener and the Top Angle of the Shell or the Top Wind Girder of an Open-top Tank (m)

$Z_{act} = 20.66 \text{ cm}^3$   
 $h1 = 1.05 \text{ m}$

Shell Yield Strength ( $Fy_{shell}$ ) = 250.0 MPa  
Stiffener Yield Strength ( $Fy_{stiffener}$ ) = 250.0 MPa

$Fy = \text{MIN}(Fy_{stiffener}, Fy_{shell})$   
 $Fy = \text{MIN}(250.0, 250.0)$   
 $Fy = 250.0$  (Set to 210 MPa since it cannot be greater than 210)

$Z_{req} = ((6 * h1 * (D^2)) / (0.5 * Fy)) * (P_{wd} / 1.72)$   
 $Z_{req} = ((6 * 1.05 * (7.0^2)) / (0.5 * 210)) * (0.7572 / 1.72)$   
 $Z_{req} = 1.29 \text{ cm}^3$

$Z_{act} \geq Z_{req}$

## Intermediate Stiffeners Summary

Intermediate Stiffeners Summary Table

Stiffener #	Size	Elevation (m)	Z reqd ( $\text{cm}^3$ )	Z act ( $\text{cm}^3$ )	Weight (kg)
1	L75X75X6	1.65	2.22	20.66	154.21
2	L75X75X6	3.45	1.29	20.66	154.21

# Flat Bottom: non Annular Plate Design

[Back](#)

Bottom Type (Flat) = Flat

Bottom Support Type (Continuously Supported on Foundation) = Continuously Supported on Foundation

CA = Corrosion allowance (mm)

CA\_1 = Bottom Shell Course Corrosion Allowance (mm)

E = Joint efficiency

Ma-bottom = Material

Ma\_1 = Bottom Shell Course Material

S = Bottom Shell Course Maximum Stress (MPa)

S1 = Bottom Shell Course Product Stress per API-650 Table 5.1a Note b (MPa)

S2 = Bottom Shell Course Hydrostatic Stress per API-650 Table 5.1a Note b (MPa)

Sd\_1 = Bottom Shell Course Allowable Design Stress (MPa)

St\_1 = Bottom Shell Course Allowable Hydrostatic Test Stress (MPa)

chime = Outside Projection (Chime Distance) (mm)

tb = Installed Thickness (mm)

tb-req = Bottom Required Thickness (mm)

td\_1 = Bottom Shell Course Design Thickness (mm)

ts\_1 = Bottom Shell Course Nominal Thickness (mm)

tt\_1 = Bottom Shell Course Hydrotest Thickness (mm)

CA = 3 mm

CA\_1 = 3 mm

E = 0.7

Ma-bottom = A36

Ma\_1 = A36

Sd\_1 = 145 MPa

St\_1 = 145 MPa

chime = 50 mm

tb = 10 mm

td\_1 = 4.28 mm

ts\_1 = 8 mm

tt\_1 = 1.18 mm

## Bottom Plates Material Properties

Material (A36) = A36

Minimum Tensile Strength (Sut-btm) = 400.0 MPa

Minimum Yield Strength (Sy-btm) = 250.0 MPa

Density (d-btm) = 7,840 kg/m<sup>3</sup>

Permissible Design Metal Temperature (MDMT-permissible-btm) = -30 °C

## Calculation of Hydrostatic Test Stress & Product Stress per API-650 Section 5.5.1

$S1 = ((td_1 - CA_1) / (ts_1 - CA_1)) * Sd_1$

$S1 = ((4.2791 - 3) / (8 - 3)) * 145$

$S1 = 37.09 \text{ MPa}$

As per API-650 5.5.1, first shell course material, A36, is in Group I; therefore, butt welded annular plates are not required

$S2 = (tt_1 / ts_1) * St_1$

$S2 = (1.1812 / 8) * 145$

$S2 = 21.41 \text{ MPa}$

As per API-650 5.5.1, first shell course material, A36, is in Group I; therefore, butt welded annular plates are not required

$S = \text{MAX}(S1, S2)$   
 $S = \text{MAX}(37.094, 21.4084)$   
 $S = 37.09 \text{ MPa}$

### Bottom Weight

A-btm = Bottom Surface Area ( $\text{m}^2$ )  
CA = Corrosion allowance (mm)  
OD-btm = Bottom Outer Diameter (m)  
Wb-pl = Bottom Plates Weight (kg)  
Wb-pl-corr = Bottom Corroded Plates Weight (kg)  
chime = Outside Projection (Chime Distance) (mm)  
tb = Installed Thickness (mm)

CA = 3 mm  
chime = 50 mm  
tb = 10 mm

OD-btm = OD + (chime \* 2)  
OD-btm = 7.016 + (0.05 \* 2)  
OD-btm = 7.12 m

A-btm =  $\pi * ((\text{OD-btm} / 2)^2)$   
A-btm =  $\pi * ((7.116 / 2)^2)$   
A-btm = 39.77  $\text{m}^2$

Wb-pl = A-btm \* tb \* d-btm  
Wb-pl = 39,770,564.9415 \* 10 \* 7.840000000000001E-6  
Wb-pl = 3,118.01 kg

Wb-pl-corr = A-btm \* (tb - CA) \* d-btm  
Wb-pl-corr = 39,770,564.9415 \* (10 - 3) \* 7.840000000000001E-6  
Wb-pl-corr = 2,182.61 kg

### Bottom Design due to External Pressure

P-btm = Downward Pressure (kPa)

Liquid Height to Pressure Conversion Factor (f) = 9.81

P-btm = (d-btm \* 9.80665 \* (tb - CA-btm) \* (1 / 1.0E6)) + (Lmin \* f \* SG)  
P-btm = (7,840 \* 9.80665 \* (10 - 3) \* (1 / 1.0E6)) + (0.0 \* 9.8064 \* 1.11)  
P-btm = 0.54 kPa

P-btm >= Pv ==> There is no uplift due to external pressure

### Bottom Required Thickness

As per API-650 5.4.1, Required Thickness by Erection (tb-erec) = 9 mm

tb-req = tb-erec  
tb-req = 9  
tb-req = 9 mm

tb >= tb-req ==> PASS

### Bottom Outside Projection

As per API-650 5.4.2, Minimum Required Outside Projection (chime) = 50 mm

chime >= chime ==> PASS



# Wind Moment (Per API-650 Section 5.11) [Back](#)

## Wind Pressures per API-650 & ASCE7-10

PWR = Roof Design Wind Pressure per API-650 5.2.1.k (kPa)

PWS = Shell Design Wind Pressure per API-650 5.2.1.k (kPa)

V = Design Wind Velocity (3-sec gust) (kph)

Vs = Adjusted Design Wind Velocity (kph)

$$V = 144.0 \text{ kph}$$

Wind Velocity per API-650 and ASCE7-10

$$V_s = 0.78 * V$$

$$V_s = 0.78 * 144.0$$

$$V_s = 112.32 \text{ kph}$$

### Roof Wind Pressure

$$PWR = 1.48 * ((V_s / 190)^2)$$

$$PWR = 1.48 * ((112.32 / 190)^2)$$

$$PWR = 0.52 \text{ kPa}$$

### Shell Wind Pressure

$$PWS = 0.89 * ((V_s / 190)^2)$$

$$PWS = 0.89 * ((112.32 / 190)^2)$$

$$PWS = 0.31 \text{ kPa}$$

## Wind Overturning and Sliding Stability

As = Shell Total Vertical Projected Area (m<sup>2</sup>)

CA-btm = Corrosion Allowance of Bottom Plates Under the Shell (mm)

CA\_1 = Bottom Shell Course Corrosion Allowance (mm)

COF = Maximum Allowable Sliding Friction Coefficient

D-outer = Tank Max Outer Diameter (m)

DLR = Nominal Weight of Roof Plates and Attached Structural (N)

DLS = Nominal Weight of Shell Plates and Framing (N)

F-friction = Friction Force (N)

F-wind = Sliding Force (N)

Fby = Yield Strength of Bottom Plates Under the Shell (MPa)

MDL = Moment About the Shell-To-Bottom Joint from the Nominal Weight of the Shell (N.m)

MF = Moment About the Shell-To-Bottom Joint From Liquid Weight (N.m)

MWS = Shell Wind Overturning Moment per API-650 5.11.2.2 (N.m)

Mw = Overturning Moment About the Shell-To-Bottom Joint from Wind Pressures per API-650 5.11.2.2 (N.m)

Wb-pl-corr = Bottom Corroded Plates Weight (N)

Ws-framing = Shell New Framing Weight (N)

Ws-framing-corr = Shell Corroded Framing Weight (N)

Ws-pl = Shell New Plates Weight (N)

Ws-pl-corr = Shell Corroded Plates Weight (N)

Xs = Moment Arm of Wind Force on Shell (m)

tb = Thickness of Bottom Plates Under the Shell (mm)

ts-ins = Shell Insulation Thickness (mm)

ts\_1 = Bottom Shell Course Nominal Thickness (mm)

wL = Tank Content Resisting Weight per API-650 5.11.2.3 (N/m)

CA-btm = 3 mm  
 CA\_1 = 3 mm  
 COF = 0.4  
 DLR = 0.0 N  
 DLS = 69,045.8 N  
 Fby = 250.0 MPa  
 Wb-pl-corr = 21,404.08 N  
 Ws-framing = 8,123.97 N  
 Ws-framing-corr = 5,520.52 N  
 Ws-pl = 60,937.3 N  
 Ws-pl-corr = 38,085.81 N  
 tb = 10 mm  
 ts-ins = 0 mm  
 ts\_1 = 8 mm

### **Overturning Moments**

D-outer = OD + (2 \* (ts-ins / 1000))  
 D-outer = 7.016 + (2 \* (0 / 1000))  
 D-outer = 7.02 m

As = D-outer \* H  
 As = 7.016 \* 4.5  
 As = 31.57 m<sup>2</sup>

Xs = H / 2  
 Xs = 4.5 / 2  
 Xs = 2.25 m

MWS = PWS \* As \* Xs  
 MWS = 311.0262 \* 31.572 \* 2.25  
 MWS = 22,094.37 N.m

Mw = MWR + MWS  
 Mw = 0.0 + 22,094.3692  
 Mw = 22,094.37 N.m

### **Resistance to Overturning per API-650 5.11.2**

MDL = (D / 2) \* DLS  
 MDL = (7.0 / 2) \* 69,045.8046  
 MDL = 241,660.32 N.m

As per API-650 5.11.2.3, the corroded thickness of plates under the shell used in the liquid resisting weight calculation shall not be greater than first shell course corroded thickness

$wL = \text{MIN}((70.4 * L_{\text{max}} * D), (59 * (ts_1 - CA_1) * \text{SQRT}((F_{by} * L_{\text{max}}))))$   
 $wL = \text{MIN}((70.4 * 3.71 * 7.0), (59 * (8 - 3) * \text{SQRT}((250.0 * 3.71))))$   
 wL = 1,828.29 N/m

MF = (D / 2) \* wL \* pi \* D  
 MF = (7.0 / 2) \* 1,828.288 \* pi \* 7.0  
 MF = 140,721.54 N.m

### **An unanchored tank must meet the criteria from API-650 5.11.2.1**

#### **Criterion 1**

$((0.6 * M_w) + M_{Pi}) < ((MDL / 1.5) + MDLR)$   
 $((0.6 * 22,094.3692) + 0.0) < ((241,660.3162 / 1.5) + 0.0)$   
 13,256.6215 < 161,106.8775 ==> Tank is stable

#### **Criterion 2**

$(M_w + (F_p * M_{Pi})) < (((MDL + MF) / 2) + MDLR)$

$(22,094.3692 + (0.4 * 0.0)) < (((241,660.3162 + 140,721.5357) / 2) + 0.0)$   
 $22,094.3692 < 191,190.9259 \Rightarrow$  Tank is stable

### **Criterion 3**

$(MWS + (Fp * MPi)) < ((MDL / 1.5) + MDLR)$   
 $(22,094.3692 + (0.4 * 0.0)) < ((241,660.3162 / 1.5) + 0.0)$   
 $22,094.3692 < 161,106.8775 \Rightarrow$  Tank is stable

### **Resistance to Sliding per API-650 5.11.4**

F-wind = PWS \* As

F-wind = 311.0262 \* 31.572

F-wind = 9,819.72 N

F-friction = COF \* (Wr-pl-corr + W-struct-corr + Ws-pl-corr + Ws-framing-corr + Wb-pl-corr)

F-friction = 0.4 \* (0.0 + 0.0 + 38,085.8123 + 5,520.5153 + 21,404.0787)

F-friction = 26,004.16 N

F-friction  $\geq$  F-wind  $\Rightarrow$  Tank is stable

### **Anchorage Requirement**

Tank anchorage due to wind is not required per API-650 5.11

# Seismic Design [Back](#)

## Site Ground Motion Design

Ac = Convective Design Response Spectrum Acceleration Coefficient per API 650 Sections E.4.6.1

Ac-min = Adjusted Convective Design Response Spectrum Acceleration Coefficient

Af = Acceleration Coefficient for Sloshing Wave Height per API 650 Sections E.7.2

Ai = per API 650 Sections E.4.6.1

Ai = Impulsive Design Response Spectrum Acceleration Coefficient per API 650 Sections E.4.6.1

Anchorage\_System = Anchorage System

Av = Vertical Ground Acceleration Coefficient per API 650 Sections E.6.1.3 and E.2.2

D = Nominal Tank Diameter (m)

Fa = Site Acceleration Coefficient

Fv = Site Velocity Coefficient

H = Maximum Design Product Level (m)

I = Importance Factor

K = Spectral Acceleration Adjustment Coefficient

Ks = Sloshing Coefficient per API 650 Section E.4.5.2

Q = MCE to Design Level Scale Factor

Rwc = Convective Force Reduction Factor

Rwi = Impulsive Force Reduction Factor

S1 = Spectral Response Acceleration at a Period of One Second

SD1 = Design Spectral Response Acceleration at a Period of One Second per API 650 Sections E.4.6.1 and E.2.2

SDS = Design Spectral Response Acceleration at Short Period per API 650 Sections E.4.6.1 and E.2.2

Seismic\_Site\_Class = Seismic Site Class

Seismic\_Use\_Group = Seismic Use Group

Ss = Spectral Response Acceleration Short Period

TL = Regional Dependent Transistion Period for Longer Period Ground Motion (sec)

Tc = Convective Natural Period per API 650 Section E.4.5.2 (sec)

Anchorage\_System = MECHANICALLY-ANCHORED

D = 7.0 m

Fa = 1.02

Fv = 1.5

H = 3.71 m

I = 1.5

K = 1.5

Q = 0.67

Rwc = 2

Rwi = 4

S1 = 0.5

Seismic\_Site\_Class = SEISMIC-SITE-CLASS-D

Seismic\_Use\_Group = SEISMIC-USE-GROUP-III

Ss = 1.2

TL = 4 sec

$SDS = Q * Fa * Ss$

$SDS = 0.6667 * 1.02 * 1.2$

$SDS = 0.82$

$SD1 = Q * Fv * S1$

$SD1 = 0.6667 * 1.5 * 0.5$

$SD1 = 0.5$

$Ks = 0.578 / \text{SQRT}(\text{TANH}(((3.68 * H) / D)))$

$Ks = 0.578 / \text{SQRT}(\text{TANH}(((3.68 * 3.71) / 7.0)))$

$Ks = 0.59$

$T_c = 1.8 * K_s * \text{SQRT}(D)$   
 $T_c = 1.8 * 0.5898 * \text{SQRT}(7.0)$   
 $T_c = 2.81 \text{ sec}$

$A_i = \text{SDS} * (I / R_{wi})$   
 $A_i = 0.816 * (1.5 / 4)$   
 $A_i = 0.31$

$A_i = \text{MAX}(A_i, 0.007)$   
 $A_i = \text{MAX}(0.306, 0.007)$   
 $A_i = 0.31$

$T_c \leq T_L$

$A_c = K * \text{SD1} * (1 / T_c) * (I / R_{wc})$   
 $A_c = 1.5 * 0.5 * (1 / 2.8089) * (1.5 / 2)$   
 $A_c = 0.2$

$A_{c-\text{min}} = \text{MIN}(A_c, A_i)$   
 $A_{c-\text{min}} = \text{MIN}(0.2003, 0.306)$   
 $A_{c-\text{min}} = 0.2$

$A_v = (2 / 3) * 0.7 * \text{SDS}$   
 $A_v = (2 / 3) * 0.7 * 0.816$   
 $A_v = 0.38$

Vertical Ground Acceleration Coefficient Specified by user ( $A_v$ ) = 0.38

$A_f = K * \text{SD1} * I * (1 / T_c)$   
 $A_f = 1.5 * 0.5 * 1 * (1 / 2.8089)$   
 $A_f = 0.27$

As per API-650 E.4.6.1, for tanks falling in SUG III, the importance factor shall be set equal to 1 in the determination of sloshing wave height = for tanks falling in SUG III, the importance factor shall be set equal to 1 in the determination of sloshing wave height

## Seismic Design

$A$  = Roof Surface Area ( $\text{m}^2$ )  
 $A_{rs}$  = Roof Area Supported by The Shell ( $\text{m}^2$ )  
 $A_c$  = Convective Design Response Spectrum Acceleration Coefficient  
 $A_f$  = Acceleration Coefficient for Sloshing Wave Height  
 $A_{h\text{-shell}}$  = Roof Horizontal Projected Area Supported by The Shell ( $\text{m}^2$ )  
 $A_i$  = Impulsive Design Response Spectrum Acceleration Coefficient  
Anchorage\_System = Anchorage System  
 $A_v$  = Vertical Ground Acceleration Coefficient  
 $D$  = Nominal Tank Diameter (m)  
 $\Delta$  = Sloshing Wave Height Above Product Design Height per API 650 Section E.7.2 (m)  
Event\_Type = Event Type  
 $F_a$  = Site Acceleration Coefficient  
 $F_c$  = Allowable Longitudinal Shell Compression Stress per API 650 Section E.6.2.2.3 (MPa)  
Freeboard = Actual Freeboard (m)  
Freeboard\_required = Minimum Required Freeboard per API-650 Table E.7 (m)  
 $F_v$  = Site Velocity Coefficient  
 $F_y$  = Yield Strength (MPa)  
 $G$  = Specific Gravity  
 $G_e$  = Effective Specific Gravity per API 650 Section E.2.2  
 $H$  = Maximum Design Product Level (m)  
 $H_{\text{shell}}$  = Shell height (m)  
 $H_{rcg}$  = Top of Shell to Roof and roof appurtenances Center of Gravity (m)

I = Importance Factor  
 J = Anchorage Ratio per API 650 Section E.6.2.1.1.1  
 K = Spectral Acceleration Adjustment Coefficient  
 Ks = Sloshing Coefficient  
 MU = Friction Coefficient  
 Min\_Anchor\_Quantity = Minimum Anchor Quantity  
 Min\_Anchor\_Spacing = Minimum Anchor Spacing (m)  
 Mrw = Ringwall Overturning Moment per API 650 Section E.6.1.5 (N.m)  
 Ms = Slab Overturning Moment per API 650 Section E.6.1.5 (N.m)  
 Overturn\_Stability\_Ratio = Overturning Stability Ratio per API 650 Section E.6.2.3  
 P = Design Pressure (MPa)  
 Q = MCE to Design Level Scale Factor  
 S1 = Spectral Response Acceleration at a Period of One Second  
 SD1 = Design Spectral Response Acceleration at a Period of 1 Second  
 SDS = Design Spectral Response Acceleration at Short Period  
 Sb = Roof Balanced Snow Load (Pa)  
 Sc = Mechanically Anchored Maximum Longitudinal Shell Compression Stress per API 650 Section E.6.2.2.2 (MPa)  
 Seismic\_Site\_Class = Seismic Site Class  
 Seismic\_Use\_Group = Seismic Use Group  
 Ss = Spectral Response Acceleration Short Period  
 TL = Regional Dependent Transistion Period for Longer Period Ground Motion (sec)  
 Tc = Convective Natural Period (sec)  
 V = Total Design Base Shear per API 650 Section E.6.1 (N)  
 Vc = Design Base Shear for Convective Component per API 650 Section E.6.1 (N)  
 Vi = Design Base Shear for Impulsive Component per API 650 Section E.6.1 (N)  
 Vmax = Local Shear Transfer per API 650 Section E.7.7 (N/m)  
 Vs = Self Anchored Sliding Resistance Maximum Allowable Base Shear per API 650 Section E.7.6 (N)  
 W-struct = Roof Structure Weight (kg)  
 W\_T = Total Weight of Tank Shell, Roof, Framing, Knuckles, Product, Bottom, Attachments, Appurtenances, Participating Balanced Snow Load per API-650 Eq E.6.2.3-1 (N)  
 Wb-attachments = Bottom Attachments Weight (kg)  
 Wb-pl = Bottom Plates Weight (kg)  
 Wc = Convective Effective Weight per API 650 Section E.6.1.1 (N)  
 Weff = Total Effective Weight per API 650 Section E.6.1.1 (N)  
 Wf = Tank Bottom Total Weight (N)  
 Wfd = Tank Foundation Weight (N)  
 Wg = Soil Weight (N)  
 Wi = Impulsive Effective Weight per API 650 Section E.6.1.1 (N)  
 Wp = Tank Contents Total Weight (N)  
 Wr = Total Weight of Fixed Tank Roof including Framing, Knuckles, any Permanent Attachments and 10 % of the Roof Balanced Design Snow Load (N)  
 Wr-DL-add = Roof Additional Dead Weight (kg)  
 Wr-attachments = Roof Attachments Weight (kg)  
 Wr-pl = Roof Plates Nominal Weight (kg)  
 Wrs = Roof Load Acting on The Tank Shell Including 10 % of the Roof Balanced Design Snow Load (N)  
 Ws = Total Weight of Tank Shell and Appurtenances (N)  
 Ws-attachments = Shell Attachments Weight (kg)  
 Ws-framing = Shell Framing Weight (kg)  
 Ws-pl = Shell Plates Nominal Weight (kg)  
 Wss = Roof Structure Weight Supported by The Tank Shell (kg)  
 Xc = Height from tank shell bottom to the center of action of convective lateral force for computing ringwall overturning moment per API 650 Section E.6.1.2.1 (m)  
 Xcs = Height from tank shell bottom to the center of action of convective lateral force for computing slab overturning moment per API 650 Section E.6.1.2.2 (m)  
 Xi = Height from tank shell bottom to the center of action of impulsive lateral force for computing ringwall overturning moment per API 650 Section E.6.1.2.1 (m)  
 Xis = Height from tank shell bottom to the center of action of impulsive lateral force for computing slab

overturning moment per API 650 Section E.6.1.2.2 (m)  
 $X_r$  = Height from tank shell bottom to the center of gravity of roof and roof appurtenances per API 650 Section E.6.1.2 (m)  
 $X_s$  = Height from tank shell bottom to shell's center of gravity (m)  
 $ca_1$  = Bottom Shell Course Corrosion Allowance (mm)  
 $ca_{bottom}$  = Bottom Corrosion Allowance (mm)  
 $hs$  = Additional Shell Height Required Above Sloshing Height (mm)  
 $t_{bottom}$  = Bottom Plate Thickness (mm)  
 $ta$  = Thickness, excluding corrosion allowance, of the bottom annulus under the shell required to provide the resisting force for self anchorage per API-650 E.2.2 (mm)  
 $tb_{corr}$  = Bottom Plates Corroded Thickness (mm)  
 $ts_1$  = Bottom Shell Course Thickness (mm)  
 $ts_{1c}$  = Shell Course 1 Corroded Thickness (mm)  
 $wa$  = (N/m)  
 $wa$  = Self Anchored Force Resisting Uplift per API 650 Section E.6.2.1.1 (N/m)  
 $wa_{max}$  = Self Anchored Force Resisting Uplift Max Limit per API 650 Section E.6.2.1.1 (N/m)  
 $wint$  = Calculated Design Uplift Due to Product Pressure (N/m)  
 $wrs$  = Specified Tank Roof Load Acting on Tank Shell (N/m)  
 $wt$  = Tank and Roof Weight Acting at base of Shell per API 650 Section E.6.2.1.1.1 (N/m)

$A = 0 \text{ m}^2$   
 $A_{rs} = 0 \text{ m}^2$   
 $Ac = 0.2$   
 $Af = 0.27$   
 $Ah_{shell} = 0 \text{ m}^2$   
 $Ai = 0.31$   
 Anchorage\_System = MECHANICALLY-ANCHORED  
 $Av = 0.38$   
 $D = 7.0 \text{ m}$   
 Event\_Type = MAXIMUM-CONSIDERED-EARTHQUAKE-MCE  
 $Fa = 1.02$   
 $Fv = 1.5$   
 $Fy = 250.0 \text{ MPa}$   
 $G = 1.11$   
 $H = 3.71 \text{ m}$   
 $H_{shell} = 4.5 \text{ m}$   
 $Hrcg = 0.0 \text{ m}$   
 $I = 1.5$   
 $K = 1.5$   
 $Ks = 0.59$   
 $MU = 0.4$   
 Min\_Anchor\_Quantity = 6  
 Min\_Anchor\_Spacing = 3 m  
 $P = 0.0 \text{ MPa}$   
 $Q = 0.67$   
 $S_1 = 0.5$   
 $SD_1 = 0.5$   
 $SDS = 0.82$   
 $S_b = 0.0 \text{ Pa}$   
 Seismic\_Site\_Class = SEISMIC-SITE-CLASS-D  
 Seismic\_Use\_Group = SEISMIC-USE-GROUP-III  
 $S_s = 1.2$   
 $TL = 4 \text{ sec}$   
 $T_c = 2.81 \text{ sec}$   
 $W_{struct} = 2,188.16 \text{ kg}$   
 $W_{b-attachments} = 0 \text{ kg}$   
 $W_{b-pl} = 3,118.01 \text{ kg}$   
 $W_{fd} = 0 \text{ N}$   
 $W_g = 0 \text{ N}$   
 $W_p = 1,554,187.91 \text{ N}$

$W_{r-DL-add} = 0 \text{ kg}$   
 $W_{r-attachments} = 0 \text{ kg}$   
 $W_{r-pl} = 0 \text{ kg}$   
 $W_{s-attachments} = 2,120.18 \text{ kg}$   
 $W_{s-framing} = 828.41 \text{ kg}$   
 $W_{s-pl} = 6,213.88 \text{ kg}$   
 $W_{ss} = 2,188.16 \text{ kg}$   
 $X_s = 2.25 \text{ m}$   
 $ca_1 = 3 \text{ mm}$   
 $ca\_bottom = 3 \text{ mm}$   
 $hs = 0 \text{ mm}$   
 $t\_bottom = 10 \text{ mm}$   
 $ts_1 = 8 \text{ mm}$

$W_f = W_{b-pl}$   
 $W_f = 30,577.2552$   
 $W_f = 30,577.26 \text{ N}$

$W_r = (W_{r-pl} + W_{r-attachments} + W_{struct} + W_{r-DL-add}) + (0.1 * S_b * A_h)$   
 $W_r = (0.0 + 0.0 + 21,458.4928 + 0.0) + (0.1 * 0.0 * 0)$   
 $W_r = 21,458.49 \text{ N}$

$W_{rs} = ((W_{r-pl} + W_{r-attachments} + W_{r-DL-add}) * 1) + W_{ss} + (0.1 * S_b * A_{h-shell})$   
 $W_{rs} = ((0.0 + 0.0 + 0.0) * 1) + 21,458.4928 + (0.1 * 0.0 * 0)$   
 $W_{rs} = 21,458.49 \text{ N}$

$W_s = W_{s-pl} + W_{s-framing} + W_{s-attachments}$   
 $W_s = 60,937.2996 + 8,123.9667 + 20,791.8972$   
 $W_s = 89,853.16 \text{ N}$

$W\_T = W_s + W_r + W_p + W_f$   
 $W\_T = 89,853.1635 + 21,458.4928 + 1,554,187.9068 + 30,577.2552$   
 $W\_T = 1,696,076.82 \text{ N}$

#### Effective Weight of Product

$W_i = (\text{TANH}((0.866 * (D / H))) / (0.866 * (D / H))) * W_p$   
 $W_i = (\text{TANH}((0.866 * (7.0 / 3.71))) / (0.866 * (7.0 / 3.71))) * 1,554,187.9068$   
 $W_i = 881,383.56 \text{ N}$

$W_c = 0.23 * (D / H) * \text{TANH}(((3.67 * H) / D)) * W_p$   
 $W_c = 0.23 * (7.0 / 3.71) * \text{TANH}(((3.67 * 3.71) / 7.0)) * 1,554,187.9068$   
 $W_c = 647,437.67 \text{ N}$

$W_{eff} = W_i + W_c$   
 $W_{eff} = 881,383.5587 + 647,437.6748$   
 $W_{eff} = 1,528,821.23 \text{ N}$

#### Design Loads

$V_i = A_i * (W_s + W_r + W_f + W_i)$   
 $V_i = 0.306 * (89,853.1635 + 21,458.4928 + 30,577.2552 + 881,383.5587)$   
 $V_i = 313,121.38 \text{ N}$

$V_c = A_c * W_c$   
 $V_c = 0.2003 * 647,437.6748$   
 $V_c = 129,681.77 \text{ N}$

$V = \text{SQRT}((V_i^2) + (V_c^2))$   
 $V = \text{SQRT}((313,121.3759^2) + (129,681.7663^2))$   
 $V = 338,913.49 \text{ N}$



**Center of Action for Effective Lateral Forces**

$$X_r = H_{\text{shell}} + H_{\text{rcg}}$$

$$X_r = 4.5 + 0.0$$

$$X_r = 4.5 \text{ m}$$

$$X_i = 0.375 * H$$

$$X_i = 0.375 * 3.71$$

$$X_i = 1.39 \text{ m}$$

$$X_c = (1.0 - ((\cosh(((3.67 * H) / D)) - 1) / (((3.67 * H) / D) * \sinh(((3.67 * H) / D)))) * H$$

$$X_c = (1.0 - ((\cosh(((3.67 * 3.71) / 7.0)) - 1) / (((3.67 * 3.71) / 7.0) * \sinh(((3.67 * 3.71) / 7.0)))) * 3.71$$

$$X_c = 2.28 \text{ m}$$

$$X_{is} = 0.375 * (1.0 + (1.333 * (((0.866 * (D / H)) / \tanh((0.866 * (D / H)))) - 1.0))) * H$$

$$X_{is} = 0.375 * (1.0 + (1.333 * (((0.866 * (7.0 / 3.71)) / \tanh((0.866 * (7.0 / 3.71)))) - 1.0))) * 3.71$$

$$X_{is} = 2.81 \text{ m}$$

$$X_{cs} = (1.0 - ((\cosh(((3.67 * H) / D)) - 1.937) / (((3.67 * H) / D) * \sinh(((3.67 * H) / D)))) * H$$

$$X_{cs} = (1.0 - ((\cosh(((3.67 * 3.71) / 7.0)) - 1.937) / (((3.67 * 3.71) / 7.0) * \sinh(((3.67 * 3.71) / 7.0)))) * 3.71$$

$$X_{cs} = 2.8 \text{ m}$$

**Overturning Moment**

$$M_{rw} = \text{SQRT}((((A_i * ((W_i * X_i) + (W_s * X_s) + (W_r * X_r)))^2) + ((A_c * (W_c * X_c))^2)))$$

$$M_{rw} = \text{SQRT}((((0.306 * ((881,383.5587 * 1.3912) + (89,853.1635 * 2.25) + (21,458.4928 * 4.5)))^2) + ((0.2003 * (647,437.6748 * 2.2798))^2)))$$

$$M_{rw} = 552,412.65 \text{ N.m}$$

$$M_s = \text{SQRT}((((A_i * ((W_i * X_{is}) + (W_s * X_s) + (W_r * X_r)))^2) + ((A_c * (W_c * X_{cs}))^2)))$$

$$M_s = \text{SQRT}((((0.306 * ((881,383.5587 * 2.8069) + (89,853.1635 * 2.25) + (21,458.4928 * 4.5)))^2) + ((0.2003 * (647,437.6748 * 2.8015))^2)))$$

$$M_s = 922,958.06 \text{ N.m}$$

**Resistance to Design Loads**

$$G_e = G * (1 - (0.4 * A_v))$$

$$G_e = 1.11 * (1 - (0.4 * 0.3808))$$

$$G_e = 0.94$$

$$w_{rs} = W_{rs} / (\pi * D)$$

$$w_{rs} = 21,458.4928 / (\pi * 7.0)$$

$$w_{rs} = 975.78 \text{ N/m}$$

$$w_t = (W_s / (\pi * D)) + w_{rs}$$

$$w_t = (89,853.1635 / (\pi * 7.0)) + 975.7786$$

$$w_t = 5,061.66 \text{ N/m}$$

$$w_{int} = P * 1000000 * ((\pi * (D^2) / 4)) / (\pi * D)$$

$$w_{int} = 0.0 * 1000000 * ((\pi * (7.0^2) / 4)) / (\pi * 7.0)$$

$$w_{int} = 0.0 \text{ N/m}$$

**Bottom Annular Plates Requirements**

$$t_{b\text{-corr}} = t_{\text{bottom}} - c_{a\text{\_bottom}}$$

$$t_{b\text{-corr}} = 10 - 3$$

$$t_{b\text{-corr}} = 7 \text{ mm}$$

$$t_{s1\_c} = t_{s1} - c_{a1}$$

$$t_{s1\_c} = 8 - 3$$

$$t_{s1\_c} = 5 \text{ mm}$$

$t_a = \text{MIN}(t_{b\text{-corr}}, t_{s1\_c})$   
 $t_a = \text{MIN}(7, 5)$   
 $t_a = 5 \text{ mm}$

$w_{a\_max} = 201.1 * H * D * G_e$   
 $w_{a\_max} = 201.1 * 3.71 * 7.0 * 0.9409$   
 $w_{a\_max} = 4,914.04 \text{ N/m}$

$w_a = 99 * t_a * \text{SQRT}((F_y * H * G_e))$   
 $w_a = 99 * 5 * \text{SQRT}((250.0 * 3.71 * 0.9409))$   
 $w_a = 14,623.11 \text{ N/m}$

$w_a > w_{a\_max}$

$w_a = w_{a\_max}$   
 $w_a = 4,914.0428$   
 $w_a = 4,914.04 \text{ N/m}$

### **Tank Stability**

$J = Mr_w / ((D^2) * (((w_t * (1 - (0.4 * A_v))) + w_a) - (F_p * w_{int})))$   
 $J = 552,412.6546 / ((7.0^2) * (((5,061.6572 * (1 - (0.4 * 0.3808))) + 4,914.0428) - (0.4 * 0.0)))$   
 $J = 1.22$

$J \leq 1.54 \implies$  Tank is stable, anchoring is not required

$Sc = ((w_t * (1 + (0.4 * A_v))) + ((1.273 * Mr_w) / (D^2))) * (1 / (1000 * t_s))$   
 $Sc = ((5,061.6572 * (1 + (0.4 * 0.3808))) + ((1.273 * 552,412.6546) / (7.0^2))) * (1 / (1000 * 5))$   
 $Sc = 4.04 \text{ MPa}$

$F_c = (83 * (t_s / (2.5 * D))) + (7.5 * \text{SQRT}((G * H)))$   
 $F_c = (83 * (5 / (2.5 * 7.0))) + (7.5 * \text{SQRT}((1.11 * 3.71)))$   
 $F_c = 38.93 \text{ MPa}$

$Sc < F_c$

$\text{Overturn\_Stability\_Ratio} = (0.5 * D * (W_T + W_{fd} + W_g)) / M_s$   
 $\text{Overturn\_Stability\_Ratio} = (0.5 * 7.0 * (1,696,076.8184 + 0 + 0)) / 922,958.0622$   
 $\text{Overturn\_Stability\_Ratio} = 6.43$

$\text{Overturn\_Stability\_Ratio} \geq 2.0 \implies$  PASS

### **Freeboard**

$\Delta L_{TAs} = 0.42 * D * A_f$   
 $\Delta L_{TAs} = 0.42 * 7.0 * 0.267$   
 $\Delta L_{TAs} = 0.78 \text{ m}$

$\text{Freeboard} = H_{\text{shell}} - L_{\text{max-operating}}$   
 $\text{Freeboard} = 4.5 - 3.71$   
 $\text{Freeboard} = 0.79 \text{ m [790.0 mm]}$

$\text{Freeboard\_required} = \Delta L_{TAs}$   
 $\text{Freeboard\_required} = 0.785$   
 $\text{Freeboard\_required} = 0.78 \text{ m [784.98 mm]}$

$\text{Freeboard} \geq \text{Freeboard\_required} \implies$  PASS

### **Sliding Resistance**

$V_s = MU * (W_s + W_r + W_f + W_p) * (1.0 - (0.4 * A_v))$   
 $V_s = 0.4 * (89,853.1635 + 21,458.4928 + 30,577.2552 + 1,554,187.9068) * (1.0 - (0.4 * 0.3808))$   
 $V_s = 575,092.16 \text{ N}$

$$V \leq V_s$$

**Local Shear Transfer**

$$V_{\max} = (2 * V) / (\pi * D)$$

$$V_{\max} = (2 * 338,913.4942) / (\pi * 7.0)$$

$$V_{\max} = 30,822.72 \text{ N/m}$$

# Anchor Bolt Design [Back](#)

A-s = Installed Bolt Nominal Root Area ( $\text{mm}^2$ )  
A-s-r = Anchor Required Root Area ( $\text{mm}^2$ )  
Av = Seismic Vertical Earthquake Acceleration Coefficient (g)  
Ca-anchor = Anchor Corrosion Allowance (mm)  
D = Tank nominal diameter (m)  
Dac = Bolt Circle Diameter (m)  
Fp = Design Pressure Operating Ratio  
Fty = Minimum Yield Strength of the Bottom Shell Course (MPa)  
Fy = Anchor Yield Strength per API-650 Table 5.21a (MPa)  
Fy-ambient = Anchor Yield Strength at Ambient Temperature per API-650 Table 5.21a (MPa)  
H = Tank Height (m)  
MWS = Shell Wind Overturning Moment (N.m)  
Ma-anchor = Anchor Material  
Mrw = Seismic Overturning Moment (N.m)  
N = Anchors Quantity  
N-min = Minimum Required Number of Anchors per API-650 5.12.3  
OD = Tank Outer diameter (m)  
P = Internal Pressure (kPa)  
P-attachment = Anchor Attachment Design Load per API-650 5.12.13 and Steel Plate Engineering Data-Volume 2 Part V (N)  
PWR = Roof Wind Pressure (kPa)  
Pt = Test Pressure (kPa)  
Sd = Allowable Anchor Stress per API-650 Table 5.21a (MPa)  
Sd-shell = Allowable Shell Stress at Anchor Attachment per API-650 Table 5.21a (MPa)  
Tb = Load per Anchor per API-650 5.12.2 (N)  
U = Net Uplift Load per API-650 5.12.2 (N)  
W1 = Corroded Weight of the Roof Plates Plus the Corroded Weight of the Shell and any Other Corroded Permanent Attachments Acting on the Shell (N)  
W2 = Corroded Weight of the Shell and any Corroded Permanent Attachments Acting on the Shell Including the Portion of the Roof Plates and Framing Acting on The Shell (N)  
W3 = Nominal Weight of the Roof Plates Plus the Nominal Weight of the Shell and any Other Permanent Attachments Acting on the Shell (N)  
Wr-pl = Roof Plates Nominal Weight (N)  
Wr-pl-corr = Roof Corroded Plates Weight (N)  
Wrs-pl-corr = Roof Plates Corroded Weight Acting on The Shell (N)  
Ws-framing = Shell New Framing Weight (stiffeners) (N)  
Ws-framing-corr = Shell Corroded Framing Weight (stiffeners) (N)  
Ws-pl = Shell Plates Nominal Weight (N)  
Ws-pl-corr = Shell Corroded Plates Weight (N)  
Wss = Roof Structure Nominal Weight Acting on The Shell (N)  
Wss-corr = Roof Structure Corroded Weight Acting on The Shell (N)  
Y-bolt = Anchor Yield Load (N)  
d = Anchor Bolt Diameter (mm)  
d-req = Bolt Required Diameter per ANSI B1.1 (mm)  
p = Bolt Thread Pitch (mm)  
position\_angles = Anchors Position Angles (deg)

Av = 0.38 g  
Ca-anchor = 1.5 mm  
D = 7.0 m  
Dac = 7.18 m  
Fp = 0.4  
Fty = 250.0 MPa  
H = 4.5 m  
MWS = 22,094.37 N.m  
Ma-anchor = A307-B  
Mrw = 552,412.65 N.m

N = 12  
 OD = 7.02 m  
 P = 0.0 kPa  
 PWR = 0 kPa  
 Pt = 0.0 kPa  
 Wr-pl = 0.0 N  
 Wr-pl-corr = 0.0 N  
 Wrs-pl-corr = 0 N  
 Ws-framing = 8,123.97 N  
 Ws-framing-corr = 5,520.52 N  
 Ws-pl = 60,937.3 N  
 Ws-pl-corr = 38,085.81 N  
 Wss = 21,458.49 N  
 Wss-corr = 0.0 N  
 d = 27 mm  
 p = 3.0 mm  
 position\_angles = [0 30 60 90 120 150 180 210 240 270 300 330 ] deg

### **Anchors Spacing Requirements**

#### **Max Allowable Spacing Between Anchors at Shell Outer Diameter per API-650 5.12.3**

Max Allowable Spacing (max\_allowable\_spacing) = 3 m

Actual Spacing (actual\_spacing) = 1.84 m

actual\_spacing <= max\_allowable\_spacing ==> PASS

N-min = CEILING(((pi \* OD) / 3))

N-min = CEILING(((pi \* 7.016) / 3))

N-min = 8

N >= N-min ==> PASS

Anchors meet spacing requirements.

#### **Anchors Average Spacing (half the span on each side of the anchor) at Bolt Circle**

Anchors are equally spaced.

Average Spacing (average\_spacing) = 1.88 m

Bolt loads will be based on equally spaced anchors.

### **Anchor Material Properties**

Material (A307-B) = A307-B

Minimum Tensile Strength (Sut-anchor) = 415 MPa

Minimum Yield Strength (Sy-anchor) = 250 MPa

Minimum Yield Strength at Ambient Temperature (Sy-ambient-anchor) = 250 MPa

Fy = MIN(Sy-ambient-anchor , 380)

Fy = MIN(250 , 380)

Fy = 250 MPa

Fy-ambient = MIN(Sy-ambient-anchor , 380)

Fy-ambient = MIN(250 , 380)

Fy-ambient = 250 MPa

### **Uplift Load Cases per API-650 Table 5.21a**

W1 = Ws-pl-corr + Ws-framing-corr + Wr-pl-corr

W1 = 38,085.8123 + 5,520.5153 + 0.0

W1 = 43,606.33 N

W2 = Ws-pl-corr + Ws-framing-corr + Wrs-pl-corr + Wss-corr

W2 = 38,085.8123 + 5,520.5153 + 0 + 0.0

W2 = 43,606.33 N

$$W3 = Ws-pl + Ws-framing + Wr-pl + Wss$$

$$W3 = 60,937.2996 + 8,123.9667 + 0.0 + 21,458.4928$$

$$W3 = 90,519.76 \text{ N}$$

#### **Uplift Case 1: Design Pressure Only**

$$U = (P * (D^2) * 785) - W1$$

$$U = (0.0 * (7.0^2) * 785) - 43,606.3275$$

$$U = -43,606.33 \text{ (Set to 0 N since it cannot be less than 0)}$$

$$Tb = U / N$$

$$Tb = 0 / 12$$

$$Tb = 0 \text{ N}$$

$$Sd = (5 / 12) * Fy$$

$$Sd = (5 / 12) * 250$$

$$Sd = 104.17 \text{ MPa}$$

$$A-s-r = Tb / Sd$$

$$A-s-r = 0 / 104.1667$$

$$A-s-r = 0.0 \text{ mm}^2$$

$$P\text{-attachment} = 1.5 * Tb$$

$$P\text{-attachment} = 1.5 * 0$$

$$P\text{-attachment} = 0.0 \text{ N}$$

$$Sd\text{-shell} = (2 / 3) * Fty$$

$$Sd\text{-shell} = (2 / 3) * 250.0$$

$$Sd\text{-shell} = 166.67 \text{ MPa}$$

#### **Uplift Case 2: Test Pressure Only**

$$U = (Pt * (D^2) * 785) - W3$$

$$U = (0.0 * (7.0^2) * 785) - 90,519.7592$$

$$U = -90,519.76 \text{ (Set to 0 N since it cannot be less than 0)}$$

$$Tb = U / N$$

$$Tb = 0 / 12$$

$$Tb = 0 \text{ N}$$

$$Sd = (5 / 9) * Fy\text{-ambient}$$

$$Sd = (5 / 9) * 250$$

$$Sd = 138.89 \text{ MPa}$$

$$A-s-r = Tb / Sd$$

$$A-s-r = 0 / 138.8889$$

$$A-s-r = 0.0 \text{ mm}^2$$

$$P\text{-attachment} = 1.5 * Tb$$

$$P\text{-attachment} = 1.5 * 0$$

$$P\text{-attachment} = 0.0 \text{ N}$$

$$Sd\text{-shell} = (5 / 6) * Fty$$

$$Sd\text{-shell} = (5 / 6) * 250.0$$

$$Sd\text{-shell} = 208.33 \text{ MPa}$$

#### **Uplift Case 3: Wind Load Only**

$$U = ((PWR * (D^2) * 785) + ((4 * MWS) / D)) - W2$$

$$U = ((0 * (7.0^2) * 785) + ((4 * 22,094.3692) / 7.0)) - 43,606.3275$$

$$U = -30,980.97 \text{ (Set to 0 N since it cannot be less than 0)}$$

$$T_b = U / N$$

$$T_b = 0 / 12$$

$$T_b = 0 \text{ N}$$

$$S_d = 0.8 * F_y$$

$$S_d = 0.8 * 250$$

$$S_d = 200.0 \text{ MPa}$$

$$A-s-r = T_b / S_d$$

$$A-s-r = 0 / 200.0$$

$$A-s-r = 0.0 \text{ mm}^2$$

$$P\text{-attachment} = 1.5 * T_b$$

$$P\text{-attachment} = 1.5 * 0$$

$$P\text{-attachment} = 0.0 \text{ N}$$

$$S_d\text{-shell} = (5 / 6) * F_{ty}$$

$$S_d\text{-shell} = (5 / 6) * 250.0$$

$$S_d\text{-shell} = 208.33 \text{ MPa}$$

#### **Uplift Case 4: Seismic Load Only**

$$U = ((4 * Mr_w) / D) - (W_2 * (1 - (0.4 * A_v)))$$

$$U = ((4 * 552,412.6546) / 7.0) - (43,606.3275 * (1 - (0.4 * 0.3808)))$$

$$U = 278,700.16 \text{ N}$$

$$T_b = U / N$$

$$T_b = 278,700.1623 / 12$$

$$T_b = 23,225.01 \text{ N}$$

$$S_d = 0.8 * F_y$$

$$S_d = 0.8 * 250$$

$$S_d = 200.0 \text{ MPa}$$

$$A-s-r = T_b / S_d$$

$$A-s-r = 23,225.0135 / 200.0$$

$$A-s-r = 116.13 \text{ mm}^2$$

$$P\text{-attachment} = 3 * T_b$$

$$P\text{-attachment} = 3 * 23,225.0135$$

$$P\text{-attachment} = 69,675.04 \text{ N}$$

$$S_d\text{-shell} = (5 / 6) * F_{ty}$$

$$S_d\text{-shell} = (5 / 6) * 250.0$$

$$S_d\text{-shell} = 208.33 \text{ MPa}$$

#### **Uplift Case 5: Design Pressure + Wind Load**

$$U = (((F_p * P) + PWR) * (D^2) * 785) + ((4 * MWS) / D) - W_1$$

$$U = (((0.4 * 0.0) + 0) * (7.0^2) * 785) + ((4 * 22,094.3692) / 7.0) - 43,606.3275$$

$$U = -30,980.97 \text{ (Set to 0 N since it cannot be less than 0)}$$

$$T_b = U / N$$

$$T_b = 0 / 12$$

$$T_b = 0 \text{ N}$$

$$S_d = (5 / 9) * F_y$$

$$S_d = (5 / 9) * 250$$

$$S_d = 138.89 \text{ MPa}$$

$$A-s-r = T_b / S_d$$

$$A-s-r = 0 / 138.8889$$

$$A-s-r = 0.0 \text{ mm}^2$$

$$P\text{-attachment} = 1.5 * Tb$$

$$P\text{-attachment} = 1.5 * 0$$

$$P\text{-attachment} = 0.0 \text{ N}$$

$$Sd\text{-shell} = (5 / 6) * Fty$$

$$Sd\text{-shell} = (5 / 6) * 250.0$$

$$Sd\text{-shell} = 208.33 \text{ MPa}$$

#### Uplift Case 6: Design Pressure + Seismic Load

$$U = ((Fp * P * (D^2) * 785) + ((4 * Mrw) / D)) - (W1 * (1 - (0.4 * Av)))$$

$$U = ((0.4 * 0.0 * (7.0^2) * 785) + ((4 * 552,412.6546) / 7.0)) - (43,606.3275 * (1 - (0.4 * 0.3808)))$$

$$U = 278,700.16 \text{ N}$$

$$Tb = U / N$$

$$Tb = 278,700.1623 / 12$$

$$Tb = 23,225.01 \text{ N}$$

$$Sd = 0.8 * Fy$$

$$Sd = 0.8 * 250$$

$$Sd = 200.0 \text{ MPa}$$

$$A-s-r = Tb / Sd$$

$$A-s-r = 23,225.0135 / 200.0$$

$$A-s-r = 116.13 \text{ mm}^2$$

$$P\text{-attachment} = 3 * Tb$$

$$P\text{-attachment} = 3 * 23,225.0135$$

$$P\text{-attachment} = 69,675.04 \text{ N}$$

$$Sd\text{-shell} = (5 / 6) * Fty$$

$$Sd\text{-shell} = (5 / 6) * 250.0$$

$$Sd\text{-shell} = 208.33 \text{ MPa}$$

#### Uplift Case 7: Frangibility Pressure

Not applicable. It is applied if the roof to shell joint is frangible.

#### Summary of Uplift Cases

Uplift Cases	Total Uplift Load (N)	Load per Anchor (N)	Anchor Allowable Stress (MPa)	Anchor Required Area (mm <sup>2</sup> )	Anchor Bolt Required Diameter (mm)	Attachment Design Load (N)	Allowable Shell Stress at Anchor Attachment (MPa)
Design Pressure	0	0	104.17	0.0	6.68	0.0	166.67
Test Pressure	0	0	138.89	0.0	6.68	0.0	208.33
Wind Load	0	0	200.0	0.0	6.68	0.0	208.33
Seismic Load	278,700.16	23,225.01	200.0	116.13	18.84	69,675.04	208.33
Design Pressure + Wind	0	0	138.89	0.0	6.68	0.0	208.33



Design Pressure + Seismic	278,700.16	23,225.01	200.0	116.13	18.84	69,675.04	208.33
<ul style="list-style-type: none"> <li>Anchor Bolt Required Diameter = <math>\text{SQRT}((A-s-r * (4 / \pi))) + (1.22687 * p) + (Ca\text{-anchor} * 2)</math></li> <li>Governing Uplift Case = Seismic Load</li> <li>Anchor Bolt Minimum Required Area = 116.13 mm<sup>2</sup></li> </ul>							

### Bolt Required Diameter per ANSI B1.1

$$d\text{-req} = \text{SQRT}((A * (4 / \pi))) + (1.22687 * n) + (Ca * 2)$$

$$d\text{-req} = \text{SQRT}((116.1251 * (4 / \pi))) + (1.22687 * 3.0) + (1.5 * 2)$$

$$d\text{-req} = 18.84 \text{ mm}$$

$$d \geq d\text{-req} \Rightarrow \text{PASS}$$

$$A-s = (\pi / 4) * ((d - (1.22687 * n))^2)$$

$$A-s = (\pi / 4) * ((27 - (1.22687 * 3.0))^2)$$

$$A-s = 427.09 \text{ mm}^2$$

$$Y\text{-bolt} = A-s * Sy\text{-ambient-anchor}$$

$$Y\text{-bolt} = 427.0948 * 250$$

$$Y\text{-bolt} = 106,773.69 \text{ N}$$

### Anchorage Summary

Required Number of Anchors = 8

Actual Number of Anchors = 12

Bolt Hole Circle Radius = 3.59 m

Required Bolt Diameter = 18.84 mm

Actual Bolt Diameter = 27 mm

Bolt Thread Pitch = 3.0 mm

# Anchor Chair Design [Back](#)

## Anchor Chair Design per AISI Steel Plate Engineering Data-Volume 2 Part V

CA = Chair Corrosion Allowance (mm)

D = Tank Nominal Diameter (m)

Earthquakes-Considered = Earthquakes Considered

Et = Bottom Plates Thermal Expansion Coefficient per API-650 Table P.1a (mm/m.cdeg)

Ma-chair = Chair Material

R = Nominal Shell Radius (mm)

Ssw-chair = Chair Allowable Stress for Seismic or Wind Design per API-650 5.12.9 (MPa)

T = Difference Between the Ambient Temperature and the Maximum Design Temperature (°C)

V = Wind Velocity (kph)

Y-bolt = Anchor Bolt Yield Load (N)

a = Top Plate Width Along Shell (mm)

b = Top Plate Length (mm)

bmin = Top Plate Minimum Length (mm)

c = Top Plate Thickness (mm)

c-corr = Top Plate Corroded Thickness (mm)

d = Anchor Bolt Diameter (mm)

e = Anchor Bolt Eccentricity (mm)

emin = Minimum Calculated Eccentricity (mm)

emin-btm = Minimum Eccentricity Based on Bolt Clearance From Bottom Plates per API-650 5.12.4 (mm)

emin-req = Minimum Required Eccentricity (mm)

f = Top Plate Outside To Hole Edge Distance (mm)

fmin = Top Plate Outside to Hole Edge Minimum Distance (mm)

g = Vertical Plates Distance (mm)

gmin = Vertical Plates Minimum Distance (mm)

h = Chair Height (mm)

h-eff = Effective Chair Height (mm)

hmax = Chair Maximum Height (mm)

j = Vertical Plate Thickness (mm)

j-corr = Vertical Plate Corroded Thickness (mm)

jmin = Vertical Plate Minimum Thickness (mm)

k = Vertical Plates Average Width (mm)

m = Base or Bottom Plate Thickness (mm)

outside-projection = Bottom Outside Projection (mm)

t = Shell Thickness (mm)

CA = 1.5 mm

D = 7.0 m

Earthquakes-Considered = ASCE7-MAPPED-SS-AND-S1

Et = 1.2E-5 mm/m.cdeg

Ma-chair = A36

R = 3,500 mm

T = 12 °C

V = 112.32 kph

Y-bolt = 106,773.69 N

a = 180 mm

b = 180 mm

c = 16 mm

d = 27 mm

e = 83 mm

f = 76.5 mm

g = 90 mm

h = 305 mm

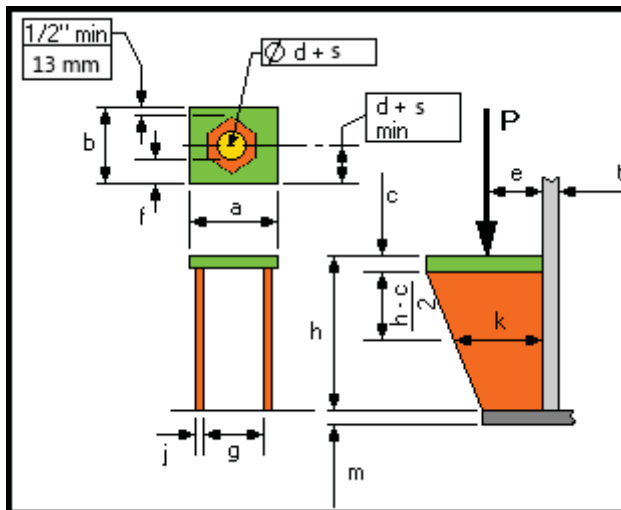
j = 14 mm

k = 115.58 mm

m = 10 mm

outside-projection = 50 mm

t = 8 mm



### Anchor Chair Material Properties

Material (A36) = A36

Minimum Tensile Strength ( $S_{ut-chair}$ ) = 400.0 MPa

Minimum Yield Strength ( $S_{y-chair}$ ) = 250.0 MPa

As per API-650 A.4.1, Allowable Design Stress ( $S_{d-chair}$ ) = 145 MPa

As per API-650 A.4.1, Allowable Hydrostatic Test Stress ( $S_{t-chair}$ ) = 145 MPa

$S_{sw-chair} = 1.33 * S_{d-chair}$

$S_{sw-chair} = 1.33 * 145$

$S_{sw-chair} = 192.85 \text{ MPa}$

### Size Requirements

$c_{-corr} = c - (2 * CA)$

$c_{-corr} = 16 - (2 * 1.5)$

$c_{-corr} = 13.0 \text{ mm}$

$j_{-corr} = j - (2 * CA)$

$j_{-corr} = 14 - (2 * 1.5)$

$j_{-corr} = 11.0 \text{ mm}$

Chair Minimum Height ( $h_{min}$ ) = 305 mm

$h \geq h_{min} \implies \text{PASS}$

$h_{max} = 3 * a$   
 $h_{max} = 3 * 180$   
 $h_{max} = 540 \text{ mm}$

$h_{\text{-eff}} = \text{MIN}(h_{max}, h)$   
 $h_{\text{-eff}} = \text{MIN}(540, 305)$   
 $h_{\text{-eff}} = 305 \text{ mm}$

$e_{min} = (0.886 * d) + 15$   
 $e_{min} = (0.886 * 27) + 15$   
 $e_{min} = 38.92 \text{ mm}$

$e_{min\text{-btm}} = (d / 2) + \text{outside-projection} + 3 + (500 * E_t * D * T)$   
 $e_{min\text{-btm}} = (27 / 2) + 50 + 3 + (500 * 1.2E-5 * 7.0 * 12)$   
 $e_{min\text{-btm}} = 67.0 \text{ mm}$

$e_{min\text{-req}} = \text{MAX}(e_{min}, e_{min\text{-btm}})$   
 $e_{min\text{-req}} = \text{MAX}(38.922, 67.004)$   
 $e_{min\text{-req}} = 67.0 \text{ mm}$

$e \geq e_{min\text{-req}} \implies \text{PASS}$

$g_{min} = d + 26$   
 $g_{min} = 27 + 26$   
 $g_{min} = 53 \text{ mm}$

$g \geq g_{min} \implies \text{PASS}$

$f_{min} = (d / 2) + 4$   
 $f_{min} = (27 / 2) + 4$   
 $f_{min} = 17.5 \text{ mm}$

$f \geq f_{min} \implies \text{PASS}$

$j_{min} = \text{MAX}(13, (0.04 * (h_{\text{-eff}} - c_{\text{-corr}})), ((P_{\text{-design}} / (172.3689 * k)) + (2 * CA)))$   
 $j_{min} = \text{MAX}(13, (0.04 * (305 - 13.0)), ((69,675.0406 / (172.3689 * 115.5773)) + (2 * 1.5)))$   
 $j_{min} = 13 \text{ mm}$

$j \geq j_{min} \implies \text{PASS}$

$b_{min} = e_{min} + d + 7$   
 $b_{min} = 38.922 + 27 + 7$   
 $b_{min} = 72.92 \text{ mm}$

$b \geq b_{min} \implies \text{PASS}$

#### Top Plate Minimum Required Thickness

Uplift Cases	P-chair (N)	P-design (N)	Sd-chair (MPa)	c-min (mm)	Status
Design Pressure	0.0	0.0	145	3.0	PASS
Test Pressure	0.0	0.0	145	3.0	PASS
Wind Load	0.0	0.0	192.85	3.0	PASS
Seismic Load	69,675.04	69,675.04	192.85	14.46	PASS
Design Pressure + Wind	0.0	0.0	192.85	3.0	PASS

Design Pressure + Seismic	69,675.04	69,675.04	192.85	14.46	PASS
<ul style="list-style-type: none"> <li>P-chair = Anchor Chair Uplift Load</li> <li>P-design = Anchor Chair Design Load = min(P-chair, Y-bolt)</li> <li>Sd-chair = Anchor Chair Allowable Stress</li> <li>c-min = Top Plate Minimum Required Thickness</li> <li>c-min = <math>\text{SQRT}(((P\text{-design} / (Sd\text{-chair} * f)) * ((0.375 * g) - (0.22 * d)))) + (2 * CA)</math></li> <li>Governing Uplift Case = Seismic Load</li> <li>Governing Thickness (c-min) = 14.46 mm</li> </ul>					

### Top Plate Stress

Uplift Cases	P-chair (N)	P-design (N)	S-top-plate (MPa)	Sd-chair (MPa)	Stress Ratio	Status
Design Pressure	0.0	0.0	0.0	145	0.0%	PASS
Test Pressure	0.0	0.0	0.0	145	0.0%	PASS
Wind Load	0.0	0.0	0.0	192.85	0.0%	PASS
Seismic Load	69,675.04	69,675.04	149.88	192.85	77.72%	PASS
Design Pressure + Wind	0.0	0.0	0.0	192.85	0.0%	PASS
Design Pressure + Seismic	69,675.04	69,675.04	149.88	192.85	77.72%	PASS
<ul style="list-style-type: none"> <li>P-chair = Anchor Chair Uplift Load</li> <li>P-design = Anchor Chair Design Load = min(P-chair, Y-bolt)</li> <li>S-top-plate = Top Plate Stress</li> <li>S-top-plate = <math>(P\text{-design} / (f * (c\text{-corr}^2))) * ((0.375 * g) - (0.22 * d))</math></li> <li>Sd-chair = Anchor Chair Allowable Stress</li> <li>Governing Uplift Case = Seismic Load</li> <li>Governing Stress (S-top-plate) = 149.88 MPa</li> </ul>						

Z = Chair Reduction Factor

### Shell Stress at Anchor Attachment

$$Z = 26 / (((0.177 * a * m) / \text{SQRT}((R * t))) * ((m / t)^2)) + 26)$$

$$Z = 26 / (((0.177 * 180 * 10) / \text{SQRT}((3,500 * 8))) * ((10 / 8)^2)) + 26)$$

$$Z = 0.9$$

Uplift Cases	P-chair (N)	P-design (N)	S-Shell (MPa)	Sd-shell (MPa)	Stress Ratio	Status
Design Pressure	0.0	0.0	0.0	166.67	0.0%	PASS
Test Pressure	0.0	0.0	0.0	208.33	0.0%	PASS
Wind Load	0.0	0.0	0.0	208.33	0.0%	PASS
Seismic Load	69,675.04	69,675.04	101.76	208.33	48.85%	PASS
Design Pressure + Wind	0.0	0.0	0.0	208.33	0.0%	PASS
Design Pressure + Seismic	69,675.04	69,675.04	101.76	208.33	48.85%	PASS
<ul style="list-style-type: none"> <li>P-chair = Anchor Chair Uplift Load</li> </ul>						

- P-design = Anchor Chair Design Load = min(P-chair, Y-bolt)
- S-Shell = Stress at Attachment
- $$S\text{-Shell} = ((P\text{-design} * e) / (t^2)) * (((1.32 * Z) / (((1.43 * a * (h^2)) / (R * t)) + (4 * a * (h^2))^{0.333}))) + (0.031 / \text{SQRT}((R * t))))$$
- Sd-shell = Allowable Stress at Anchor Attachment
- Governing Uplift Case = Seismic Load
- Governing Stress (S-Shell) = 101.76 MPa
- Governing Allowable Stress (Sd-Shell) = 208.33 MPa

# Appurtenances Design [Back](#)

## Plan View

LABEL	MARK	CUST. MARK	DESCRIPTION	OUTSIDE PROJ (mm)	INSIDE PROJ (mm)	ORIENT	RADIUS (mm)	REMARKS	REF DWG
Agitator-Bridge	AB01		AGITATOR BRIDGE	--	--	90'	--		

## Elevation View

LABEL	MARK	CUST. MARK	DESCRIPTION	OUTSIDE PROJ (mm)	INSIDE PROJ (mm)	ORIENT	ELEVATION (mm)	REMARKS	REF DWG
Agitator-Bridge	AB01		AGITATOR BRIDGE	--	--	90'	1136		
Anchor-Chair-Bolts	AC01A		ANCHOR CHAIRS	--	--	SEE TABLE	--		
Angle-Baffle-0001	BF01A		BAFFLE WALL	--	--	90'	0		
Circular-Clean-Out-N3 (8")	CCD01A		CIRCULAR CLEAN OUT	200	0	145'	110		
MH1 (24")	SM01A		24" SHELL MANWAY	250	0	60'	750	W/ DAVIT	
MH2 (24")	SM01A		24" SHELL MANWAY	250	0	200'	750	W/ DAVIT	
N1 (10")	SN01A		10" SHELL NOZZLE	225	0	115'	350		
N2A (6")	SN02A		6" SHELL NOZZLE	200	0	145'	75		
N2B (6")	SN02A		6" SHELL NOZZLE	200	0	350'	75		
N3 (6")	SN03A		6" SHELL NOZZLE	200	0	330'	4075		
Name-Plate	NP01A		STD API	--	--	0'	1010		

# Shell Nozzle: N1 (10")

## Repad Design

NOZZLE Description : 10 in SCH 80 TYPE RFSO  
Material: A106-B

t\_rpr = (Repad Required Thickness)  
t\_n = (Thickness of Neck)  
Sd\_n = (Stress of Neck Material)  
Sd\_s = (Stress of Shell Course Material)  
CA = (Corrosion Allowance of Neck)

MOUNTED ON SHELL 1 : Elevation = 0.35 m

### COURSE PARAMETERS:

t-calc = 4.28 mm  
t\_cr = 1.28 mm (Course t-calc less C.A.)  
t\_c = 5 mm (Course t less C.A.)  
t\_Basis = 1.28 mm  
Repad Type: Circular  
Repad Size (Do): 585 mm

(SHELL NOZZLE REF. API-650 TABLE 5.6A, AND FOOTNOTE A OF TABLE 5-7)

Required Area = t\_Basis \* D  
Required Area = 1.28 \* 273.05  
Required Area = 349.26 mm<sup>2</sup>

Available Shell Area = (t\_c - t\_Basis) \* D  
Available Shell Area = (5 - 1.28) \* 273.05  
Available Shell Area = 1015.99 mm<sup>2</sup>

Available Nozzle Neck Area = 2 \* [(4 \* (t\_n - CA)) + t\_c] \* (t\_n - CA) \* MIN((Sd\_n/Sd\_s) 1)  
Available Nozzle Neck Area = 2 \* [(4 \* (15.1 - 3)) + 5] \* (15.1 - 3) \* MIN((124.1/145) 1)  
Available Nozzle Neck Area = 1106.06 mm<sup>2</sup>

A-rpr = (Required Area - Available Shell Area - Available Nozzle Neck Area)  
A-rpr = 349.26 - 1015.99 - 1106.06  
A-rpr = -1772.8 mm<sup>2</sup>

Since A-rpr <= 0, t\_rpr = 0

No Reinforcement Pad required.

# Shell Nozzle: N2A (6")

## Repad Design

NOZZLE Description : 6 in SCH 80 TYPE RFSO



Material: A106-B

$t_{rpr}$  = (Repad Required Thickness)  
 $t_n$  = (Thickness of Neck)  
 $Sd_n$  = (Stress of Neck Material)  
 $Sd_s$  = (Stress of Shell Course Material)  
CA = (Corrosion Allowance of Neck)

MOUNTED ON SHELL 1 : Elevation = 0.075 m

#### COURSE PARAMETERS:

$t_{calc}$  = 4.28 mm  
 $t_{cr}$  = 1.28 mm (Course  $t_{calc}$  less C.A.)  
 $t_c$  = 5 mm (Course  $t$  less C.A.)  
 $t_{Basis}$  = 1.28 mm  
Repad Type: Dog House  
Repad Size (Do): 400 mm

(SHELL NOZZLE REF. API-650 TABLE 5.6A, AND FOOTNOTE A OF TABLE 5-7)

Required Area =  $t_{Basis} * D$   
Required Area =  $1.28 * 168.27$   
Required Area = 215.24 mm<sup>2</sup>

Available Shell Area =  $(t_c - t_{Basis}) * D$   
Available Shell Area =  $(5 - 1.28) * 168.27$   
Available Shell Area = 626.13 mm<sup>2</sup>

Available Nozzle Neck Area =  $2 * [(4 * (t_n - CA)) + t_c] * (t_n - CA) * \text{MIN}((Sd_n/Sd_s) 1)$   
Available Nozzle Neck Area =  $2 * [(4 * (10.97 - 3)) + 5] * (10.97 - 3) * \text{MIN}((124.1/145) 1)$   
Available Nozzle Neck Area = 503.16 mm<sup>2</sup>

A-rpr = (Required Area - Available Shell Area - Available Nozzle Neck Area)  
A-rpr = 215.24 - 626.13 - 503.16  
A-rpr = -914.05 mm<sup>2</sup>

Since A-rpr <= 0,  $t_{rpr}$  = 0

No Reinforcement Pad required.

## Shell Nozzle: N2B (6")

### Repad Design

NOZZLE Description : 6 in SCH 80 TYPE RF50  
Material: A106-B

$t_{rpr}$  = (Repad Required Thickness)  
 $t_n$  = (Thickness of Neck)  
 $Sd_n$  = (Stress of Neck Material)  
 $Sd_s$  = (Stress of Shell Course Material)  
CA = (Corrosion Allowance of Neck)

MOUNTED ON SHELL 1 : Elevation = 0.075 m

COURSE PARAMETERS:

t-calc = 4.28 mm

t\_cr = 1.28 mm (Course t-calc less C.A)

t\_c = 5 mm (Course t less C.A.)

t\_Basis = 1.28 mm

Repad Type: Dog House

Repad Size (Do): 400 mm

(SHELL NOZZLE REF. API-650 TABLE 5.6A, AND FOOTNOTE A OF TABLE 5-7)

Required Area = t\_Basis \* D

Required Area = 1.28 \* 168.27

Required Area = 215.24 mm<sup>2</sup>

Available Shell Area = (t\_c - t\_Basis) \* D

Available Shell Area = (5 - 1.28) \* 168.27

Available Shell Area = 626.13 mm<sup>2</sup>

Available Nozzle Neck Area = 2 \* [(4 \* (t\_n - CA)) + t\_c] \* (t\_n - CA) \* MIN((Sd\_n/Sd\_s) 1)

Available Nozzle Neck Area = 2 \* [(4 \* (10.97 - 3)) + 5] \* (10.97 - 3) \* MIN((124.1/145) 1)

Available Nozzle Neck Area = 503.16 mm<sup>2</sup>

A-rpr = (Required Area - Available Shell Area - Available Nozzle Neck Area)

A-rpr = 215.24 - 626.13 - 503.16

A-rpr = -914.05 mm<sup>2</sup>

Since A-rpr <= 0, t\_rpr = 0

No Reinforcement Pad required.

## Shell Nozzle: N3 (6")

### Repad Design

NOZZLE Description : 6 in SCH 80 TYPE RF50

Material: A106-B

t\_rpr = (Repad Required Thickness)

t\_n = (Thickness of Neck)

Sd\_n = (Stress of Neck Material)

Sd\_s = (Stress of Shell Course Material)

CA = (Corrosion Allowance of Neck)

MOUNTED ON SHELL 3 : Elevation = 4.075 m

COURSE PARAMETERS:

t-calc = 3.14 mm

t\_cr = 0.14 mm (Course t-calc less C.A)

t\_c = 5 mm (Course t less C.A.)

t\_Basis = 0.14 mm

Repad Type: Circular

Repad Size (Do): 400 mm

(SHELL NOZZLE REF. API-650 TABLE 5.6A, AND FOOTNOTE A OF TABLE 5-7)

Required Area =  $t_{\text{Basis}} * D$   
Required Area =  $0.14 * 168.27$   
Required Area = 23.18 mm<sup>2</sup>

Available Shell Area =  $(t_c - t_{\text{Basis}}) * D$   
Available Shell Area =  $(5 - 0.14) * 168.27$   
Available Shell Area = 818.19 mm<sup>2</sup>

Available Nozzle Neck Area =  $2 * [(4 * (t_n - CA)) + t_c] * (t_n - CA) * \text{MIN}((Sd_n/Sd_s) 1)$   
Available Nozzle Neck Area =  $2 * [(4 * (10.97 - 3)) + 5] * (10.97 - 3) * \text{MIN}((124.1/145) 1)$   
Available Nozzle Neck Area = 503.16 mm<sup>2</sup>

A-rpr = (Required Area - Available Shell Area - Available Nozzle Neck Area)  
A-rpr = 23.18 - 818.19 - 503.16  
A-rpr = -1298.17 mm<sup>2</sup>

Since A-rpr <= 0,  $t_{\text{rpr}} = 0$

No Reinforcement Pad required.

## Shell Manway: MH1 (24")

### Repad Design

MANWAY Description : 24 in Neck Thickness 8  
Material: A36

$t_{\text{rpr}}$  = (Repad Required Thickness)  
 $t_n$  = (Thickness of Neck)  
 $Sd_n$  = (Stress of Neck Material)  
 $Sd_s$  = (Stress of Shell Course Material)  
CA = (Corrosion Allowance of Neck)

MOUNTED ON SHELL 1 : Elevation = 0.75 m

COURSE PARAMETERS:  
 $t_{\text{calc}} = 4.28$  mm  
 $t_{\text{cr}} = 1.28$  mm (Course  $t_{\text{calc}}$  less C.A.)  
 $t_c = 5$  mm (Course  $t$  less C.A.)  
 $t_{\text{Basis}} = 1.28$  mm  
Repad Type: Circular  
Repad Size (Do): 1257.3 mm

(SHELL MANWAY REF. API-650 TABLE 5-6, AND FOOTNOTE A OF TABLE 5-7)

Required Area =  $t_{\text{Basis}} * D$   
Required Area =  $1.28 * 625.6$   
Required Area = 800.21 mm<sup>2</sup>

Available Shell Area =  $(t_c - t_{Basis}) * D$   
Available Shell Area =  $(5 - 1.28) * 625.6$   
Available Shell Area = 2327.79 mm<sup>2</sup>

Available Manway Neck Area =  $2 * [(4 * (t_n - CA)) + t_c] * (t_n - CA) * \text{MIN}((Sd_n/Sd_s) 1)$   
Available Manway Neck Area =  $2 * [(4 * (8 - 3)) + 5] * (8 - 3) * \text{MIN}((160/145) 1)$   
Available Manway Neck Area = 250 mm<sup>2</sup>

A-rpr = (Required Area - Available Shell Area - Available Manway Neck Area)  
A-rpr = 800.21 - 2327.79 - 250  
A-rpr = -1777.59 mm<sup>2</sup>

Since A\_rpr <= 0, t\_rpr = 0

No Reinforcement Pad required.

#### **Manway Neck Material Properties**

Material (A36) = A36  
As per API-650 A.4.1, Allowable Design Stress (Sd-neck) = 145 MPa  
Permissible Design Metal Temperature (MDMT-permissible-neck) = -30 °C

#### **Manway Repad Material Properties**

Material (A36) = A36  
Permissible Design Metal Temperature (MDMT-permissible-repad) = -30 °C

#### **Cover Plate and Bolting Flange Design**

CA-cover = Cover Plate and Bolting Flange Corrosion Allowance (mm)  
Db = Bolt Circle Diameter (mm)  
H = Design Liquid Level (m)  
Ma-cover = Cover Plate Material  
Ma-flange = Bolting Flange Material  
SG = Product Specific Gravity  
Sd = Allowable Stress per API-650 5.7.5.6 (MPa)  
tc = Cover Plate Thickness (mm)  
tc-design = Cover Plate Required Thickness per API-650 5.7.5.6 (mm)  
tc-req = Cover Plate Minimum Required Thickness (mm)  
tf = Bolting Flange Thickness (mm)  
tf-design = Cover Plate Required Thickness per API-650 5.7.5.6 (mm)  
tf-req = Bolting Flange Minimum Required Thickness (mm)

CA-cover = 3 mm  
Db = 768.35 mm  
H = 3.71 m  
Ma-cover = A36  
Ma-flange = A36  
SG = 1.11  
tc = 19 mm  
tf = 16 mm

Water Density (Y) = 0.00981 MPa/m  
As per API-650 5.7.5.6, Coefficient For Circular Plate (C) = 0.3

#### **Cover Plate Material Properties and Required Thickness**

Material (A36) = A36  
Minimum Yield Strength (Sy-cover) = 250.0 MPa  
Plate is impact tested

Sd =  $\text{MIN}(Sy\text{-cover}, 205) / 2 = 102.5 \text{ MPa}$

As per API-650 5.7.5.6, Cover Plate Erection Thickness (tc-erec) = 8 mm

$$tc\text{-}design = (Db * \sqrt{((C * Y * H * \max(SG, 1)) / Sd)}) + CA\text{-}cover$$
$$tc\text{-}design = (768.35 * \sqrt{((0.3 * 0.0098 * 3.71 * \max(1.11, 1)) / 102.5)}) + 3$$
$$tc\text{-}design = 11.35 \text{ mm}$$

$$tc\text{-}req = \max(tc\text{-}erec, tc\text{-}design)$$
$$tc\text{-}req = \max(8, 11.3549)$$
$$tc\text{-}req = 11.35 \text{ mm}$$

t-cover >= tc-req ==> PASS

### **Bolting Flange Material Properties and Required Thickness**

Material (A36) = A36

Minimum Yield Strength (Sy-flange) = 250.0 MPa

Permissible Design Metal Temperature (MDMT-permissible-flange) = -2.75 °C

As per API-650 5.7.5.6, Bolting Flange Erection Thickness (tf-erec) = 6 mm

$$tf\text{-}design = tc\text{-}design - 3$$
$$tf\text{-}design = 11.3549 - 3$$
$$tf\text{-}design = 8.35 \text{ mm}$$

$$tf\text{-}req = \max(tf\text{-}erec, tf\text{-}design)$$
$$tf\text{-}req = \max(6, 8.3549)$$
$$tf\text{-}req = 8.35 \text{ mm}$$

t-flange >= tf-req ==> PASS

## **Shell Manway: MH2 (24")**

### **Repad Design**

MANWAY Description : 24 in Neck Thickness 8

Material: A36

t\_rpr = (Repad Required Thickness)  
t\_n = (Thickness of Neck)  
Sd\_n = (Stress of Neck Material)  
Sd\_s = (Stress of Shell Course Material)  
CA = (Corrosion Allowance of Neck)

MOUNTED ON SHELL 1 : Elevation = 0.75 m

#### **COURSE PARAMETERS:**

t-calc = 4.28 mm  
t\_cr = 1.28 mm (Course t-calc less C.A.)  
t\_c = 5 mm (Course t less C.A.)  
t\_Basis = 1.28 mm  
Repad Type: Circular  
Repad Size (Do): 1257.3 mm

(SHELL MANWAY REF. API-650 TABLE 5-6, AND FOOTNOTE A OF TABLE 5-7)

Required Area = t\_Basis \* D

Required Area =  $1.28 * 625.6$   
Required Area =  $800.21 \text{ mm}^2$

Available Shell Area =  $(t_c - t_{\text{Basis}}) * D$   
Available Shell Area =  $(5 - 1.28) * 625.6$   
Available Shell Area =  $2327.79 \text{ mm}^2$

Available Manway Neck Area =  $2 * [(4 * (t_n - CA)) + t_c] * (t_n - CA) * \text{MIN}((Sd_n/Sd_s) 1)$   
Available Manway Neck Area =  $2 * [(4 * (8 - 3)) + 5] * (8 - 3) * \text{MIN}((160/145) 1)$   
Available Manway Neck Area =  $250 \text{ mm}^2$

A-rpr = (Required Area - Available Shell Area - Available Manway Neck Area)  
A-rpr =  $800.21 - 2327.79 - 250$   
A-rpr =  $-1777.59 \text{ mm}^2$

Since  $A_{\text{rpr}} \leq 0$ ,  $t_{\text{rpr}} = 0$

No Reinforcement Pad required.

#### **Manway Neck Material Properties**

Material (A36) = A36  
As per API-650 A.4.1, Allowable Design Stress (Sd-neck) = 145 MPa  
Permissible Design Metal Temperature (MDMT-permissible-neck) =  $-30 \text{ }^\circ\text{C}$

#### **Manway Repad Material Properties**

Material (A36) = A36  
Permissible Design Metal Temperature (MDMT-permissible-repad) =  $-30 \text{ }^\circ\text{C}$

#### **Cover Plate and Bolting Flange Design**

CA-cover = Cover Plate and Bolting Flange Corrosion Allowance (mm)  
Db = Bolt Circle Diameter (mm)  
H = Design Liquid Level (m)  
Ma-cover = Cover Plate Material  
Ma-flange = Bolting Flange Material  
SG = Product Specific Gravity  
Sd = Allowable Stress per API-650 5.7.5.6 (MPa)  
tc = Cover Plate Thickness (mm)  
tc-design = Cover Plate Required Thickness per API-650 5.7.5.6 (mm)  
tc-req = Cover Plate Minimum Required Thickness (mm)  
tf = Bolting Flange Thickness (mm)  
tf-design = Cover Plate Required Thickness per API-650 5.7.5.6 (mm)  
tf-req = Bolting Flange Minimum Required Thickness (mm)

CA-cover = 3 mm  
Db = 768.35 mm  
H = 3.71 m  
Ma-cover = A36  
Ma-flange = A36  
SG = 1.11  
tc = 19 mm  
tf = 16 mm

Water Density (Y) = 0.00981 MPa/m  
As per API-650 5.7.5.6, Coefficient For Circular Plate (C) = 0.3

#### **Cover Plate Material Properties and Required Thickness**

Material (A36) = A36  
Minimum Yield Strength (Sy-cover) = 250.0 MPa  
Plate is impact tested

$$S_d = \text{MIN}(S_y\text{-cover} , 205) / 2 = 102.5 \text{ MPa}$$

As per API-650 5.7.5.6, Cover Plate Erection Thickness (tc-erec) = 8 mm

$$t_{c\text{-design}} = (D_b * \text{SQRT}(((C * Y * H * \text{MAX}(S_G , 1)) / S_d))) + C_{A\text{-cover}}$$

$$t_{c\text{-design}} = (768.35 * \text{SQRT}(((0.3 * 0.0098 * 3.71 * \text{MAX}(1.11 , 1)) / 102.5))) + 3$$

$$t_{c\text{-design}} = 11.35 \text{ mm}$$

$$t_{c\text{-req}} = \text{MAX}(t_{c\text{-erec}} , t_{c\text{-design}})$$

$$t_{c\text{-req}} = \text{MAX}(8 , 11.3549)$$

$$t_{c\text{-req}} = 11.35 \text{ mm}$$

$$t_{\text{cover}} \geq t_{c\text{-req}} \Rightarrow \text{PASS}$$

### **Bolting Flange Material Properties and Required Thickness**

Material (A36) = A36

Minimum Yield Strength ( $S_y\text{-flange}$ ) = 250.0 MPa

Permissible Design Metal Temperature (MDMT-permissible-flange) = -2.75 °C

As per API-650 5.7.5.6, Bolting Flange Erection Thickness (tf-erec) = 6 mm

$$t_{f\text{-design}} = t_{c\text{-design}} - 3$$

$$t_{f\text{-design}} = 11.3549 - 3$$

$$t_{f\text{-design}} = 8.35 \text{ mm}$$

$$t_{f\text{-req}} = \text{MAX}(t_{f\text{-erec}} , t_{f\text{-design}})$$

$$t_{f\text{-req}} = \text{MAX}(6 , 8.3549)$$

$$t_{f\text{-req}} = 8.35 \text{ mm}$$

$$t_{\text{flange}} \geq t_{f\text{-req}} \Rightarrow \text{PASS}$$

# Capacities and Weights [Back](#)

Capacity to Top of Shell (to Tank Height) : 173 M<sup>3</sup>

Capacity to Design Liquid Level : 142 M<sup>3</sup>

Capacity to Maximum Liquid Level : 142 M<sup>3</sup>

Working Capacity (to Normal Working Level) : 0 M<sup>3</sup>

Net working Capacity (Working Capacity - Min Capacity) : 0 M<sup>3</sup>

Minimum Capacity (to Min Liq Level) : 0 M<sup>3</sup>

Component	New Condition (N)	New Condition (Kg)	Corroded (N)	Corroded (Kg)
SHELL	60,938	6,214	38,086	3,884
ROOF	0	0	0	0
RAFTERS	0	0	0	0
GIRDERS	0	0	0	0
FRAMING	0	0	0	0
COLUMNS	0	0	0	0
TRUSS	0	0	0	0
STRUCTURE COMPONENTS	0	0	0	0
BOTTOM	30,578	3,119	21,405	2,183
STAIRWAYS	0	0	0	0
STIFFENERS	8,105	827	3,962	404
WIND GIRDERS	0	0	0	0
AGITATOR BRIDGE	21,459	2,189	21,459	2,189
ANCHOR CHAIRS	1,864	190	1,864	190
APPURTENANCES	18,929	1,931	18,929	1,931
INSULATION	0	0	0	0
FLOATING ROOF	0	0	0	0
TOTAL	141,873	14,470	105,705	10,781

Weight of Tank, Empty : 14,470 Kg

Weight of Tank, Full of Product (Design SG = 1.11) : 172,953 Kg

Weight of Tank, Full of Water : 160,816.97 Kg

Net Working Weight, Full of Product (Design SG = 1.11) : 172,953.06 Kg

Net Working Weight Full of Water : 160,816.97 Kg

Foundation Area Req'd : 39.77 m<sup>2</sup>

Foundation Loading, Empty : 3,567.28 N/m<sup>2</sup>

Foundation Loading, Full of Product Design : 42,646.12 N/m<sup>2</sup>

Foundation Loading, Full of Water : 39,653.61 N/m<sup>2</sup>

## SURFACE AREAS

Roof : 40 m<sup>2</sup>

Shell : 98.96 m<sup>2</sup>

Bottom : 39.77 m<sup>2</sup>

Internal Pressure Moment : 0 N-m

Wind Moment : 22,094.36 N-m

Seismic Moment (Ringwall) : 552,412.65 N-m

Seismic Moment (Slab) : 922,958.06 N-m



MISCELLANEOUS ATTACHED ROOF ITEMS  
MISCELLANEOUS ATTACHED SHELL ITEMS

# Reactions on Foundation [Back](#)

Arss = Area of Tank Roof Supported by the Tank Shell (m<sup>2</sup>)  
 Wrss = Weight of Tank Roof Supported by the Tank Shell (kg)  
 Ws = Weight of the Tank Shell and Shell Appurtenances (kg)  
 Wss = Roof Structure Weight Supported by The Tank Shell (kg)  
 yb = Bottom Plate Density (kg/m<sup>3</sup>)  
 yw = Water Density (kg/m<sup>3</sup>)

Arss = 0 m<sup>2</sup>  
 Wss = 2,188.16 kg  
 yb = 7,840 kg/m<sup>3</sup>  
 yw = 1,000 kg/m<sup>3</sup>

Wrss = (Wr-pl + Wr-ins + Wr-attachments) + Wss  
 Wrss = (0 + 0 + 0) + 2,188.1573  
 Wrss = 2,188.16 kg

Ws = Ws-pl + Ws-ins + Ws-attachments + Ws-framing  
 Ws = 6,213.8752 + 0.0 + 2,120.1835 + 828.4141  
 Ws = 9,162.47 kg

## Unfactored (Working Stress) Downward Reactions on Foundations

Load Case	Location	Equation	Value	Unit
Dead Load	Shell	$(Ws + Wrss) / (\pi * D)$	5,061.66	N/m
Dead Load	Bottom	$t_b * y_b * 9.806649999999999E-6$	0.77	kPa
Internal Pressure	Bottom	P	0.0	kPa
Vacuum	Shell	$(P_v * Arss) / (\pi * D)$	0.0	N/m
Hydrostatic Test	Bottom	$L_{max} * y_w * (9.80665 / 1000)$	36.38	kPa
Minimum Roof Live Load	Shell	$(L_r * Arss) / (\pi * D)$	0.0	N/m
Seismic	Shell	$((4 * (Mrw / D)) + (0.4 * (Ws + Wrss) * A_v)) / (\pi * D)$	15,125.15	N/m
Seismic	Bottom	$(32 * M_s) / (\pi * (D^3) * 1000)$	27.41	kPa
Snow	Shell	$(S * Arss) / (\pi * D)$	0.0	N/m
Stored Liquid	Bottom	$SG * L_{max} * y_w * (9.80665 / 1000)$	40.38	kPa
Pressure Test	Bottom	Pt	0.0	kPa
Wind	Shell	$(2 * (H_s^2) * PWS) / (\pi * D)$	572.8	N/m
<ul style="list-style-type: none"> <li>Seismic bottom reaction varies linearly from <math>32 * M_s / (\pi * D^3)</math> at the tank shell to zero at the center of the tank</li> </ul>				